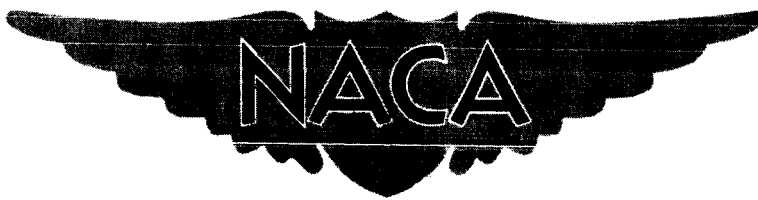


N63-12519

code-1



RESEARCH MEMORANDUM

FORCE AND PRESSURE MEASUREMENTS ON SEVERAL
CANOPY-FUSELAGE CONFIGURATIONS AT

MACH NUMBERS 1.41 AND 2.01

By A. Warner Robins

Langley Aeronautical Laboratory
Langley Field, Va.

REPRODUCTION GRANTED TO
UNCLASSIFIED - AUTHORITY
EXPIRATION DATE
SEP 20 1998

CLASSIFIED DOCUMENT

This material contains information affecting the National Defense of the United States within the meaning of the espionage laws, Title 18, U.S.C., Secs. 793 and 794, the transmission or revelation of which in any manner to an unauthorized person is prohibited by law.

NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

WASHINGTON

December 15, 1955

OTS PRICE

9.10 pd
3.44 mf

XEROX

MICROFILM

NACA RM L55H23

08p

UNCLASSIFIED
CONFIDENTIAL

NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

RESEARCH MEMORANDUM

FORCE AND PRESSURE MEASUREMENTS ON SEVERAL

CANOPY-FUSELAGE CONFIGURATIONS AT

MACH NUMBERS 1.41 AND 2.01

By A. Warner Robins

SUMMARY

An investigation has been conducted in the Langley 4- by 4-foot supersonic pressure tunnel on canopy pressures and canopy-fuselage forces and moments under conditions of combined pitch and sideslip. The canopy configurations tested varied in windshield shape (flat, vee-, and round), location on the fuselage, and fineness ratio. All configurations were tested at Mach numbers of 1.41 and 2.01 at Reynolds numbers of 1.74×10^6 and 1.44×10^6 , respectively, based on fuselage major diameter.

Drags of the canopy-fuselage combinations varied from lowest for the flat-windshield configuration to highest for the vee-windshield configuration. For comparable canopies, the configurations with the forward canopy location produced less drag than those with the rearward-located canopies, regardless of windshield shape. The effects on drag of windshield shape and canopy location were diminished with increase in Mach number from 1.41 to 2.01.

INTRODUCTION

Because of the high air loads and temperatures associated with supersonic flight, the best compromise of aerodynamic, structural, and visibility requirements in the design of canopies for military aircraft is critically dependent on the accuracy with which loads and aerodynamic characteristics can be predicted. Since practical methods for the calculation of pressure distributions and forces on such arbitrary shapes are limited, experimental data are required. A few papers showing experimental results are at present available, among them references 1 and 2 which deal with pressure distributions of two rather specialized canopy configurations at supersonic speeds. Reference 3 is concerned with

CONFIDENTIAL

transonic and supersonic drag comparisons of forward and rearward locations of a canopy on a finned test vehicle. A free-flight drag investigation of windshield-shape effects at transonic and low supersonic speeds is reported in reference 4. Reference 5 deals with the location of a canopy in order to improve the longitudinal development of cross-sectional area for a wing-fuselage combination at transonic speeds.

The present investigation is part of a program of the National Advisory Committee for Aeronautics to determine some of the effects at transonic and supersonic speeds of windshield shape, canopy location, fineness ratio, pitch, sideslip, and Mach number on the aerodynamic characteristics of several canopy-fuselage configurations and on the pressure distributions on the canopies. Reference 6 reports the force and moment characteristics at transonic speeds of some of the configurations of the present investigation. The present tests were made of models with flat, vee-, and round windshield canopies in forward and rearward locations on the fuselage. The fineness ratios of the various canopies were approximately 7.0, 10.0, and 12.0 (based on the ratio of the diameter of an equivalent body of revolution to the length of the canopy in the plane of symmetry). All configurations were tested at Mach numbers of 1.41 and 2.01 at Reynolds numbers of 1.74×10^6 and 1.44×10^6 , respectively, based on fuselage major diameter. Two canopy-fuselage configurations and the fuselage alone were tested for angles of attack from -6° to 12° , and all configurations were tested at 0° , -4° , and -8° sideslip at both 0.4° and 6.5° angle of attack. In all tests, boundary-layer transition was fixed $1/2$ inch behind the fuselage nose point by means of a roughness strip.

SYMBOLS

M	free-stream Mach number
q	free-stream dynamic pressure
p_o	free-stream static pressure
p	local pressure
P	pressure coefficient, $\frac{p - p_o}{q}$
α	angle of attack, deg
β	angle of sideslip, deg
x	distance from foremost point of canopy in plane of symmetry in an axial direction

CONFIDENTIAL

x_b	distance from fuselage nose point in an axial direction
l	canopy-profile length in an axial direction
l_b	fuselage length
ϕ	lateral angle measured from plane of symmetry (see tables X, XI, XVII, and XVIII)
A_b	area of base of model
A_{max}	maximum cross-sectional area of canopy or of a body of revolution
C_N	normal-force coefficient, $\frac{Z}{qA_b}$
C_C	axial-force coefficient, $\frac{X}{qA_b}$
C_Y	lateral-force coefficient, $\frac{Y}{qA_b}$
C_m	pitching-moment coefficient, $\frac{M'}{qA_b l_b}$
C_n	yawing-moment coefficient, $\frac{N}{qA_b l_b}$
C_l	rolling-moment coefficient, $\frac{L}{qA_b l_b}$
C_{Df}	drag coefficient, $\frac{D_f}{qA_b}$
C_D	drag coefficient, $\frac{D}{qA_b}$
ΔC_D	incremental drag coefficient, $\frac{D - D_f}{qA_b}$
C_{DA}	drag coefficient, $\frac{D}{qA_{max}}$
ΔC_{DA}	incremental drag coefficient, $\frac{D - D_f}{qA_{max}}$

X	force along body axis, positive when rearward
Y	force along lateral axis, positive when starboard
Z	force normal to XY-plane, positive when upward
D_f	force on fuselage alone in streamwise direction, positive when rearward
D	force in streamwise direction, positive when rearward
M'	moment about Y-axis, positive when tending to lift nose
N	moment about Z-axis, positive when tending to produce a right turn
L	moment about X-axis, positive when tending to produce a right bank
K	longitudinal location of maximum cross-sectional area, percent of length
P.L.	designation of canopy-fuselage parting line

MODELS AND INSTRUMENTATION

Basic Model and Canopies

The canopy shapes were tested on a drooped-nose-fuselage forebody having an elliptic cross section. Drawings and dimensions of this body, and the base plug which was used to minimize base-pressure corrections, are shown in figures 1 and 2. The various canopy configurations are described in figures 1 to 5. A family of six canopies of approximately the same size, fineness ratio (7.0), and profile was tested. Canopies with flat, vee-, and round windshields were tested at two longitudinal locations on the fuselage. Two smaller flat-windshield canopies of lower windshield slope having fineness ratios of about 10.0 and 12.0 were tested in forward and rearward locations, respectively, on the fuselage. These configurations, which are described in figures 4 and 5 approximate existing supersonic designs. Photographs of all the models are presented in figure 6.

Instrumentation

The forces and moments on the models were measured by means of a six-component strain-gage balance mounted within the fuselage. Moments were measured about a point on the model axis 14.81 inches from the nose.

Pressure instrumentation was provided in each model. The pressure orifices, which were encircled with ink prior to being photographed, may be seen in figure 6. This instrumentation was provided on only one side of the plane of symmetry so that both positive and negative sideslip angles were tested in order to determine the pressures on both the upstream and the downstream sides of the model for a given sideslip angle. The locations of the orifices for each model may be determined from tables X to XVIII.

Small prisms were mounted on the surface of the fuselage so that either angle of attack or angle of sideslip might be measured by a spectrometer head.

TESTS

Test Conditions

Mach numbers	1.41 and 2.01
Reynolds number per foot at $M = 1.41$	4.18×10^6
Reynolds number per foot at $M = 2.01$	3.46×10^6
Stagnation pressure, atm	0.95
Stagnation temperature, $^{\circ}\text{F}$	100

Corrections and Accuracy

Although force and moment data were taken at both positive and negative sideslip angles, the subsequent tabulations and plots show only one value for forces and moments and, essentially, only negative sideslip angles. Both sets of values, however, have been used; the data for all positive sideslip angles greater than 0.3° have been folded and averaged with data for negative angles.

Where angles of attack or sideslip could not be measured optically, the calibrated deflections of the balance under load were applied to the no-wind calibration of the angle mechanism so that the estimated angle accuracy was within $\pm 0.15^{\circ}$.

Base-pressure measurements were made and axial-force data were corrected to correspond to a base pressure equal to free-stream static pressure.

The force and moment coefficients are believed to be correct within the following limits:

C_N	±0.0080
C_c	±0.0040
C_m	±0.0020
C_l	±0.0015
C_n	±0.0040
C_Y	±0.0095
C_D	±0.0040

RESULTS AND DISCUSSION

Force and Moment Data

The six force and moment coefficients based on the body-axis system plus the drag coefficient based on the wind axis are tabulated and presented in tables I to IX for all model configurations. Because of the large amount of data and because drag considerations appear of greatest general interest, incremental drag coefficients (difference between the drag coefficients for the body alone and those for a canopy-fuselage combination) are the only force data discussed.

Figure 7 shows incremental drag coefficients plotted against sideslip angle for all canopy-fuselage configurations at various Mach numbers and angles of attack. Drags of the configurations with the three windshield shapes varied from the lowest for the flat-windshield configurations to the highest for the vee-windshield configurations except for the configurations with the forward-located canopies at $M = 2.01$ where the differences were about the same as the estimated possible inaccuracies of the data. For example, at $M = 1.41$ and $\alpha = 0.4^\circ$ for the forward-located canopy, the incremental drag coefficient for the flat-windshield canopy was about 75 percent of that for the vee-windshield canopy. For the large canopies, the configurations with the forward-located canopies produced less drag than those with the rearward-located canopies, regardless of windshield shape. The effects of both windshield shape and canopy location were less at $M = 2.01$ than at $M = 1.41$.

For the small canopies, the effects of location are not readily apparent in figure 7 because of differences in fineness ratio and size. In order to obtain an indication of the effects of position and fineness ratio for the flat-windshield canopies, incremental drag coefficients for zero angle of attack and sideslip were based on the maximum cross-sectional areas of the canopies themselves and are given in the following table:

CONFIDENTIAL

Flat-windshield canopy				ΔC_{DA} at -	
Size	Location	Fineness ratio	A_{max} , sq in.	M = 1.41	M = 2.01
Large	Forward	6.91	2.59	0.360	0.436
Small	Forward	10.04	1.49	.237	.312
Large	Rearward	7.06	2.46	.535	.543
Small	Rearward	12.06	1.03	.351	.381

It is apparent from this table that the forward location was also the more favorable for the small canopies. Reference 3 which presents transonic and supersonic drag comparisons of forward and rearward locations of a canopy on a finned test vehicle indicates that in the low supersonic range a rearward canopy location produces less drag. This is in contrast to the indications of the present investigation.

The M = 1.41 values from the preceding table have been plotted for all the flat-windshield configurations in figure 8 which also shows from reference 7 some M = 1.40 drag values for bodies of revolution having various locations of maximum cross-sectional area and various fineness ratios. It should be noted that the data from reference 7 are concerned with drags of bodies alone; whereas, the present data relating to canopies include mutual interference effects. Figure 8 seems to indicate that interference effects for the forward-located canopies were small compared to interference effects for the rearward location. Figure 8 also appears to show that the drag differential between the large and small canopy configurations is principally a fineness-ratio effect. The location of maximum cross-sectional area (K in fig. 8), which would in most cases be closely related to windshield slope, would be governed largely by visibility requirements. It would appear that an efficient canopy shape on a canopy-fuselage combination would require a low windshield slope and a fineness ratio of 10 or more.

Pressure Data

All pressure coefficient data for each configuration are presented in tables X to XVIII from which plots of pressure coefficient may be readily made along longitudinal meridians or radially about a particular station. Plots of these coefficients along longitudinal meridians (see tables X, XI, XVII, and XVIII for description) are presented against axial location for various angles of attack and sideslip and for Mach numbers of 1.41 and 2.01 in figures 9 to 17.

Figure 9, 10, 11, and 12 show the pressure-coefficient distributions for the large canopies at Mach numbers of 1.41 and 2.01 and indicate that pressure distributions over the aft portions of the canopies were generally not significantly influenced by windshield shape. Local peak suction were generally highest for the vee-windshield configurations although the large flat-windshield configurations began to show appreciable peaks as sideslip angle increased.

Figures 13 and 14 show pressure-coefficient distribution for the small canopies and for the fuselage alone. These, in addition to figures 9 to 12, show that suction peaks in pressure-coefficient distributions at $M = 2.01$ are generally smaller than those at $M = 1.41$, although the character of the remainder of these distributions at low sideslip angles, especially for positive coefficients, remained much the same. Figures 15, 16, and 17 show the effects of angle of attack on pressure-coefficient distributions for a forward-located round-windshield canopy, a rearward-located round-windshield canopy, and the fuselage alone, respectively. The variation of pressure coefficients over this range of angle of attack (-6.0° to 12.0°) appears to be systematic for these configurations.

Force and Pressure Correlation

A comparison of force and pressure-measurement results was made where there existed identical conditions of pitch and sideslip near zero angle of attack for both force and pressure data. Measured fuselage-alone axial-force data were diminished by the axial forces integrated from the limited pressure data on the fuselage within the area which would be covered by the canopies. The axial forces from pressures on the canopies were added to these corrected fuselage axial forces so that integrated configuration drags for the canopy-fuselage combinations resulted. These integrated values are compared with drag coefficients from force measurements in the following table:

Canopy configuration	Drag coefficient, C_D			
	$M = 1.41$		$M = 2.01$	
	Measured	Integrated	Measured	Integrated
Large forward flat	0.1695	0.1719	0.1900	0.1813
Large forward vee-	.1879	.1883	.1971	.1900
Large forward round	.1781	----	.1954	.1800
Large rearward flat	.1927	.1772	.2034	.1831
Large rearward vee-	.2178	.1893	.2160	.1933
Large rearward round	.1982	.1802	.2087	----
Small forward flat	.1328	.1417	.1475	.1424
Small rearward flat	.1342	.1289	.1420	.1365

UNCLASSIFIED
CONFIDENTIAL 9

The appreciable difference between measured and calculated forces for most of the rearward-located canopies gives credence to the supposition of larger fuselage interference effects for these rearward locations in the previous discussion of force data. In the tabulation both force and pressure-measurement results indicate that the flat-windshield canopy configurations produced less drag than the vee-configurations. The lower chord force for the flat-windshield canopy is associated with the expansions around the edges of the windshield resulting in lower pressures over the remaining two-thirds (approximately) of the canopy frontal projection. This effect is seen in figures 18, 19, and 20 which show pressure contours on half the frontal projections of the forward-located large canopies, the rearward-located large canopies, and on the small canopies, respectively. In contrast to those for the flat canopies, it is indicated by the vee-canopy contours that the expansion around the edges of the vee-windshield has little effect on forces in an axial direction. In reference 4 the drag increments for the flat-windshield canopies of comparable windshield-profile slopes were higher than for the vee-windshield canopies, in contrast to present results; however, the frontal-areas of the flat windshields of reference 4 contributed nearly all of the total canopy frontal-area so that expansions around the windshield edges could not produce reductions in canopy drags.

CONCLUSIONS

Force and pressure measurements have been made on several canopy-fuselage configurations which varied in windshield shape (flat, vee-, and round), canopy location on the fuselage, and fineness ratio. All configurations were tested in pitch and sideslip at Mach numbers of 1.41 and 2.01 for values of Reynolds number based on fuselage major diameter of 1.74×10^6 and 1.44×10^6 , respectively. The results of the tests on these configurations indicate the following conclusions:

1. For canopies which varied only in windshield shape, drags were lowest for the flat-windshield configuration and highest for the vee-configuration.
2. For comparable canopies, the configurations with the forward canopy locations produced less drag than those with the rearward-located canopies, regardless of windshield shape.

CONFIDENTIAL

3. Both the effect of windshield shape and of canopy location were diminished with the increasing of Mach number from 1.41 to 2.01.

Langley Aeronautical Laboratory,
National Advisory Committee for Aeronautics,
Langley Field, Va., August 11, 1955.

CONFIDENTIAL

UNCLASSIFIED
CONFIDENTIAL 11

REFERENCES

1. Cooper, Morton, Smith, Norman F., and Kainer, Julian H.: A Pressure-Distribution Investigation of a Supersonic Aircraft Fuselage and Calibration of the Mach Number 1.59 Nozzle of the Langley 4- by 4-Foot Supersonic Tunnel. NACA RM L9E27a, 1949.
2. Stroud, J. F., and Rich, B. R.: Performance Evaluation of XF-104 Engine Air-Induction System. Rep. No. 9557, Lockheed Aircraft Corp., Feb. 1, 1954.
3. Welsh, Clement J., and Morrow, John D.: Flight Investigation at Mach Numbers From 0.8 to 1.5 of the Drag of a Canopy Located at Two Positions on a Parabolic Body of Revolution. NACA RM L51A29, 1951.
4. Kell, C.: Measurements of the Effect of Windscreen Shapes on the Drag of Cockpit Canopies at Transonic and Low Supersonic Speeds Using the Free Flight Model Technique. Rep. No. Aero. 2529, British R.A.E., Nov. 1954.
5. Robinson, Harold L.: The Effect of Canopy Location on the Aerodynamic Characteristics of a Sweptback Wing-Body Configuration at Transonic Speeds. NACA RM L54E11, 1954.
6. Cornette, Elden S., and Robinson, Harold L.: Transonic Wind-Tunnel Investigation of Effects of Windshield Shape and Canopy Location on the Aerodynamic Characteristics of Canopy-Body Combinations. NACA RM L55G08, 1955.
7. Hart, Roger G., and Katz, Ellis R.: Flight Investigations at High-Subsonic, Transonic, and Supersonic Speeds To Determine Zero-Lift Drag of Fin-Stabilized Bodies of Revolution Having Fineness Ratios of 12.5, 8.91, and 6.04 and Varying Positions of Maximum Diameter. NACA RM L9I30, 1949.

CONFIDENTIAL

TABLE I.- FORCE AND MOMENT COEFFICIENTS FOR BODY ALONE

M	α , deg	β , deg	C_N	C_C	C_m	C_l	C_n	C_Y	C_D
1.41	0.4	0	.0052	.1101	-.0274	-.0002	.0013	.0104	.1101
1.41	0.4	-4	.0051	.1087	-.0278	-.0020	.0437	.2082	.1230
1.41	0.4	-8	.0026	.1044	-.0293	-.0041	.0870	.4486	.1659
1.41	0.4	0	.0000	.1094	-.0274	-.0002	.0013	.0124	.1094
1.41	6.5	0	.2234	.1075	.0179	.0001	-.0049	-.0052	.1321
1.41	6.5	-4	.2195	.1033	.0180	-.0061	.0427	.2166	.1423
1.41	6.5	-8	.2310	.0968	.0166	-.0121	.0831	.4692	.1865
1.41	6.5	0	.2234	.1072	.0180	.0002	-.0049	-.0052	.1318
1.41	-6.0	0.3	-.2312	.1104	-.0741	-.0001	-.0007	-.0188	.1341
1.41	-3.0	0.3	-.1233	.1130	-.0509	-.0001	-.0021	-.0188	.1194
1.41	0	0.3	-.0154	.1100	-.0276	.0000	-.0035	-.0177	.1101
1.41	3.0	0.3	.0899	.1098	-.0052	.0002	-.0052	-.0167	.1144
1.41	6.0	0.3	.1979	.1067	.0177	.0004	-.0065	-.0177	.1269
1.41	9.0	0.3	.3210	.0986	.0377	.0007	-.0076	-.0188	.1477
1.41	12.0	0.3	.4675	.0922	.0558	.0013	-.0095	-.0271	.1875
2.01	0.4	0	-.0250	.1189	-.0330	-.0001	-.0009	-.0050	.1187
2.01	0.4	-4	-.0203	.1220	-.0335	-.0020	.0391	.2348	.1380
2.01	0.4	-8	-.0343	.1179	-.0351	-.0041	.0745	.5284	.1901
2.01	6.5	0	.2399	.1162	.0083	.0001	-.0032	-.0088	.1426
2.01	6.5	-4	.2463	.1111	.0073	-.0060	.0382	.2581	.1559
2.01	6.5	-8	.2635	.1069	.0048	-.0116	.0663	.5686	.2139
2.01	-6.0	0.3	-.2908	.1296	-.0763	.0001	-.0005	-.0063	.1593
2.01	-3.0	0.3	-.1532	.1219	-.0547	.0000	-.0021	-.0127	.1298
2.01	0	0.3	-.0312	.1178	-.0338	.0000	-.0040	-.0164	.1179
2.01	3.0	0.3		.1205	-.0123	.0002	-.0059	-.0203	
2.01	6.0	0.3	.2188	.1169	.0073	.0005	-.0074	-.0266	.1393
2.01	9.0	0.3	.3749	.1102	.0239	.0008	-.0084	-.0355	.1677
2.01	12.0	0.3	.5439	.0908	.0354	.0012	-.0084	-.0469	.2021
2.01	0	0.3	-.0312	.1178	-.0338	.0000	-.0037	-.0152	.1179

TABLE II.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
FORWARD-LOCATED FLAT-WINDSHIELD CANOPY

M	α , deg	β , deg	C_N	C_C	C_m	C_l	C_n	C_Y	C_D
1.41	0.4	0	-.0077	.1696	-.0072	.0001	.0011	.0073	.1695
1.41	0.4	-4	-.0116	.1780	-.0070	.0002	.0739	.2243	.1931
1.41	0.4	-8	-.0013	.1747	-.0078	-.0011	.1482	.5085	.2438
1.41	6.5	0	.2198	.1617	.0413	.0001	-.0040	.0062	.1855
1.41	6.5	-4	.2160	.1645	.0463	-.0025	.0661	.2774	.2068
1.41	6.5	-8	.2198	.1722	.0469	-.0044	.1263	.6454	.2839
1.41	6.5	0	.2173	.1625	.0414	.0001	-.0005	.0032	.1861
2.01	0.4	0	-.0374	.1903	-.0258	.0000	-.0010	-.0063	.1900
2.01	0.4	-4	-.0406	.1945	-.0271	.0009	.0689	.2544	.2115
2.01	0.4	-8	-.0686	.2019	-.0292	.0024	.1330	.5816	.2804
2.01	0.4	0	-.0437	.1931	-.0260	.0000	-.0015	-.0088	.1928
2.01	6.5	0	.2534	.1739	.0119	.0000	-.0037	-.0127	.2015
2.01	6.5	-4	.2635	.1733	.0099	-.0041	.0606	.3316	.2247
2.01	6.5	-8	.2315	.1826	.0105	-.0057	.1123	.7567	.3110
2.01	6.5	0	.2596	.1764	.0119	.0001	-.0038	-.0127	.2047

TABLE III.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
FORWARD-LOCATED VEE-WINDSHIELD CANOPY

M	α , deg	β , deg	C_N	C_c	C_m	C_l	C_n	C_Y	C_D
1.41	0.4	0	-.0179	.1880	-.0096	.0001	.0022	.0073	.1879
1.41	0.4	-4	-.0154	.1965	-.0092	.0003	.0703	.2293	.2119
1.41	0.4	-8	-.0499	.1954	-.0051	.0011	.1434	.5045	.2634
1.41	0.4	0	-.0179	.1901	-.0093	.0001	.0005	.0051	.1900
1.41	6.5	0	.2148	.1776	.0397	.0001	-.0031	.0083	.2008
1.41	6.5	-4	.2148	.1823	.0414	-.0049	.0646	.2873	.2250
1.41	6.5	-8	.1752	.1825	.0483	-.0076	.1241	.6685	.2923
1.41	6.5	0	.2173	.1792	.0397	.0001	-.0032	.0073	.2027
2.01	0.4	0	-.0532	.1975	-.0268	.0000	-.0010	-.0063	.1971
2.01	0.4	-4	-.0516	.1964	-.0277	.0007	.0670	.2625	.2139
2.01	0.4	-8	-.0734	.2016	-.0293	.0020	.1280	.5878	.2810
2.01	0.4	0	-.0500	.1994	-.0268	-.0001	-.0012	-.0102	.1991
2.01	6.5	-4	.2593	.1751	.0083	-.0041	.0577	.3231	.2254
2.01	6.5	-8	.2358	.1834	.0100	-.0062	.1110	.7505	.3114
2.01	6.5	0	.2561	.1797	.0099	.0001	-.0050	-.0228	.2075

TABLE IV.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
FORWARD-LOCATED ROUND-WINDSHIELD CANOPY

M	α , deg	β , deg	C_N	C_c	C_m	C_l	C_n	C_Y	C_D
1.41	0.4	0	-.0128	.1782	-.0054	.0001	-.0024	.0083	.1781
1.41	0.4	-4	-.0167	.1860	-.0054	.0003	.0737	.2316	.2016
1.41	0.4	-8	-.0256	.1881	-.0046	-.0012	.1441	.5106	.2572
1.41	6.5	0	.2170	.1686	.0425	.0001	-.0075	.0031	.1921
1.41	6.5	-4	.2158	.1704	.0442	-.0025	.0666	.2860	.2132
1.41	6.5	-8	.2043	.1743	.0502	-.0044	.1262	.6301	.2821
1.41	6.5	0	.2195	.1696	.0425	.0002	-.0075	.0010	.1934
1.41	0	0.3	.0410	.1812	-.0052	-.0002	-.0113	-.0073	.1812
1.41	3.0	0.3	.0693	.1768	.0187	.0001	-.0131	-.0094	.1802
1.41	6.0	0.3	.1926	.1707	.0417	.0006	-.0152	-.0125	.1900
1.41	9.0	0.3	.3286	.1655	.0627	.0010	-.0153	-.0406	.2151
1.41	12.0	0.3	.5032	.1494	.0779	.0015	-.0136	-.0750	.2511
1.41	-3.0	0.3	-.1489	.1850	-.0291	-.0003	-.0089	-.0135	.1926
1.41	-6.0	0.3	-.2696	.1857	-.0533	-.0002	-.0063	-.0125	.2129
1.41	0	0.3	-.0385	.1805	-.0051	-.0002	-.0114	-.0156	.1806
2.01	0.4	0	-.0499	.1957	-.0249	-.0001	-.0044	-.0038	.1954
2.01	0.4	-4	-.0468	.1958	-.0262	.0008	.0701	.2671	.2136
2.01	0.4	-8	-.0687	.2059	-.0284	.0028	.1322	.6021	.2872
2.01	0.4	0	-.0437	.1957	-.0249	-.0001	-.0044	-.0038	.1954
2.01	6.5	0	.2531	.1766	.0124	-.0001	-.0063	-.0152	.2041
2.01	6.5	-4	.2640	.1791	.0105	-.0041	.0603	.3385	.2310
2.01	6.5	-8	.2406	.1850	.0112	-.0060	.1107	.7459	.3128
2.01	6.5	0	.2593	.1775	.0125	-.0001	-.0064	-.0164	.2057
2.01	0	0.3	-.0593	.1952	-.0257	-.0002	-.0100	-.0216	.1953
2.01	3.0	0.3	.0781	.1856	-.0045	.0000	-.0126	-.0317	.1896
2.01	6.0	0.3	.2281	.1770	.0118	.0003	-.0143	-.0443	.2001
2.01	9.0	0.3	.4125	.1706	.0216	.0006	-.0141	-.0672	.2334
2.01	12.0	0.3	.6406	.1625	.0219	.0006	-.0110	-.1015	.2926
2.01	-3.0	0.3	-.1937	.2055	-.0475	-.0001	-.0068	-.0140	.2154
2.01	-6.0	0.3	-.3343	.2187	-.0705	.0001	-.0037	-.0076	.2525
2.01	0	0.3	-.0593	.1942	-.0257	-.0002	-.0100	-.0229	.1943

TABLE V.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
REARWARD-LOCATED FLAT-WINDSHIELD CANOPY

M	α , deg	β , deg	C_N	C_C	C_m	C_L	C_n	C_Y	C_D
1.41	0.4	0	-.0077	.1928	-.0137	.0000	.0010	.0115	.1927
1.41	0.4	-4	-.0064	.2023	-.0144	.0021	.0760	.2448	.2189
1.41	0.4	-8	-.0307	.2027	-.0142	.0064	.1450	.5459	.2765
1.41	0.4	0	-.0077	.1953	-.0136	.0000	.0010	.0104	.1952
1.41	6.5	0	.2425	.1842	.0243	.0001	-.0075	-.0010	.2105
1.41	6.5	-4	.2349	.1851	.0250	-.0036	.0627	.3232	.2326
1.41	6.5	-8	.2234	.1846	.0272	-.0042	.1160	.7126	.3058
2.01	0.4	0	-.0374	.2037	-.0366	.0001	.0001	-.0063	.2034
2.01	0.4	-4	-.0343	.2065	-.0391	.0024	.0646	.2900	.2260
2.01	0.4	-8	-.0763	.2115	-.0412	.0074	.1170	.6691	.3021
2.01	0.4	0	-.0374	.2027	-.0368	-.0002	-.0003	-.0088	.2024
2.01	6.5	0	.2716	.1857	-.0103	-.0004	-.0025	-.0127	.2153
2.01	6.5	-4	.2700	.1870	-.0099	-.0018	.0456	.3755	.2420
2.01	6.5	-8	.2248	.1891	-.0049	-.0007	.0817	.8279	.3265
2.01	6.5	0	.2778	.1858	-.0103	-.0004	-.0023	-.0127	.2161

TABLE VI.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
REARWARD-LOCATED VEE-WINDSHIELD CANOPY

M	α , deg	β , deg	C_N	C_O	C_m	C_l	C_n	C_Y	C_D
1.41	0.4	0	-.0179	.2179	-.0124	.0002	.0010	.0083	.2178
1.41	0.4	-4	-.0167	.2210	-.0134	.0020	.0734	.2503	.2378
1.41	0.4	-8	-.0461	.2141	-.0114	.0068	.1400	.5736	.2916
1.41	0.4	0	-.0154	.2179	-.0121	.0002	.0011	.0083	.2178
1.41	6.5	0	.2429	.1985	.0236	.0002	-.0028	.0094	.2247
1.41	6.5	-4	.2442	.2016	.0238	.0033	.0572	.3362	.2509
1.41	6.5	-8	.2097	.2013	.0284	.0028	.1053	.7659	.3282
1.41	6.5	0	.2455	.1960	.0236	.0004	-.0049	.0000	.2225
2.01	0.4	0	-.0469	.2163	-.0350	.0001	.0001	-.0050	.2160
2.01	0.4	-4	-.0438	.2145	-.0364	.0012	.0629	.2929	.2341
2.01	0.4	-8	-.0782	.2188	-.0373	.0034	.1164	.6725	.3098
2.01	0.4	0	-.0406	.2182	-.0350	.0001	-.0001	-.0026	.2179
2.01	6.5	0	.2724	.1948	-.0093	-.0001	-.0011	.0000	.2244
2.01	6.5	-4	.2693	.1964	-.0098	-.0009	.0442	.3785	.2515
2.01	6.5	-8	.2223	.1941	-.0028	-.0001	.8451	.8451	.3336
2.01	6.5	0	.2755	.1919	-.0093	-.0001	-.0017	-.0039	.2219

TABLE VII.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
REARWARD-LOCATED ROUND-WINDSHIELD CANOPY

M	α , deg	β , deg	C_N	C_C	C_m	C_l	C_n	C_Y	C_D
1.41	0.4	0	-.0179	.1983	-.0109	-.0001	-.0014	.0062	.1982
1.41	0.4	-4	-.0154	.2048	-.0116	.0017	.0732	.2433	.2212
1.41	0.4	-8	-.0359	.2043	-.0114	.0055	.1397	.5503	.2787
1.41	0.4	0	-.0153	.1973	-.0110	-.0001	-.0015	.0062	.1972
1.41	6.5	0	.2380	.1870	.0271	-.0001	-.0050	.0083	.2127
1.41	6.5	-4	.2418	.1896	.0271	-.0041	.0603	.3217	.2377
1.41	6.5	-8	.2277	.1901	.0288	-.0057	.1111	.6946	.3093
1.41	0	0.3	-.0332	.1975	-.0102	-.0003	-.0105	-.0177	.1976
1.41	3.0	0.3	.0870	.1917	.0104	.0000	-.0015	-.0218	.1961
1.41	6.0	0.3	.2149	.1863	.0268	.0003	-.0116	-.0280	.2079
1.41	9.0	0.3	.3633	.1810	.0385	.0006	-.0111	-.0384	.2358
1.41	12.0	0.3	.5577	.1700	.0418	.0009	-.0099		
1.41	-3.0	0.3	-.1509	.1993	-.0319	-.0006	-.0092	-.0156	.2070
1.41	-6.0	0.3	-.2635	.2007	-.0537	-.0006	-.0071	-.0145	.2272
1.41	0	0.3	-.0332	.1975	-.0101	-.0003	-.0105	-.0177	.1976
2.01	0.4	0	-.0469	.2090	-.0334	-.0004	-.0040	-.0039	.2087
2.01	0.4	-4	-.0469	.2103	-.0357	.0021	.0635	.2898	.2297
2.01	0.4	-8	-.0766	.2148	-.0374	.0064	.1142	.6669	.3051
2.01	0.4	0	-.0406	.2129	-.0334	-.0005	-.0041	-.0063	.2126
2.01	6.5	0	.2724	.1898	-.0069	-.0002	-.0132	-.0203	.2194
2.01	6.5	-4	.2662	.1896	-.0066	-.0024	.0455	.3683	.2437
2.01	6.5	-8	.2301	.1928	-.0021	-.0019	.0802	.8077	.3279
2.01	0	0.3	-.0596	.2076	-.0341	-.0005	-.0068	-.0204	.2077
2.01	3.0	0.3	.0847	.1996	-.0183	-.0003	-.0079	-.0293	.2039
2.01	6.0	0.3	.2415	.1884	-.0064	-.0002	-.0083	-.0420	.2128
2.01	9.0	0.3	.4078	.1845	.0015	-.0001	-.0074	-.0585	.2463
2.01	12.0	0.3	.6055	.1830	.0072	-.0001	-.0046	-.0815	.3053
2.01	-3.0	0.3	-.1977	.2144	-.0523	-.0004	-.0051	-.0141	.2245
2.01	-6.0	0.3	-.3483	.2258	-.0719	-.0003	-.0031	-.0089	.2610
2.01	0	0.3	-.0565	.2047	-.0340	-.0004	-.0068	-.0204	.2048

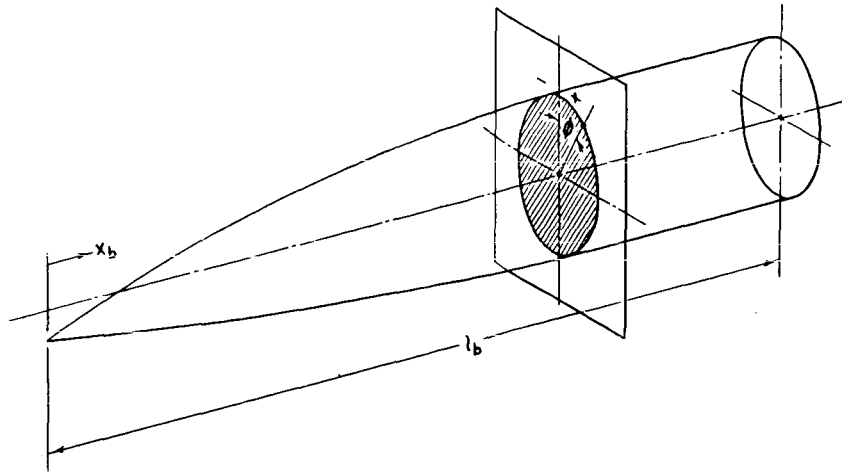
TABLE VIII.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
FORWARD-LOCATED SMALL FLAT-WINDSHIELD CANOPY

M	α , deg	β , deg	C_N	C_C	C_m	C_l	C_n	C_Y	C_D
1.41	0.4	0	-.0077	.1329	-.0168	-.0001	.0005	.0115	.1328
1.41	0.4	-4	-.0051	.1338	-.0173	-.0009	.0588	.2217	.1489
1.41	0.4	-8	-.0321	.1305	-.0169	-.0011	.1151	.4772	.1954
1.41	0.4	0	-.0077	.1321	-.0170	-.0001	.0004	.0104	.1320
1.41	6.5	0	.2202	.1347	.0299	.0002	-.0051	.0021	.1588
1.41	6.5	-4	.2151	.1307	.0312	-.0058	.0557	.2426	.1708
1.41	6.5	-8	.2074	.1226	.0354	-.0107	.1065	.5460	.2199
1.41	6.5	0	.2228	.1332	.0299	.0002	-.0049	.0011	.1576
2.01	0.4	0	-.0312	.1477	-.0285	-.0001	-.0024	-.0089	.1475
2.01	0.4	-4	-.0344	.1462	-.0295	-.0007	.0549	.2506	.1631
2.01	0.4	-8	-.0579	.1475	-.0315	-.0012	.1087	.5653	.2244
2.01	0.4	0	-.0312	.1487	-.0285	-.0001	-.0023	-.0102	.1485
2.01	6.5	0	.2441	.1394	.0109	.0000	-.0047	-.0127	.1661
2.01	6.5	-4	.2488	.1446	.0105	-.0052	.0498	.2870	.1914
2.01	6.5	-8	.2410	.1453	.0094	-.0093	.0934	.6619	.2621
2.01	6.5	0	.2472	.1420	.0109	.0001	-.0047	-.0140	.1691

TABLE IX.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
REARWARD-LOCATED SMALL FLAT-WINDSHIELD CANOPY

M	α , deg	β , deg	C_N	C_c	C_m	C_l	C_n	C_Y	C_D
1.41	0.4	0	-.0025	.1342	-.0227	-.0001	.0014	.0041	.1342
1.41	0.4	-4	-.0039	.1361	-.0229	-.0003	.0577	.2220	.1512
1.41	0.4	-8	-.0231	.1394	-.0213	-.0001	.1119	.4914	.2063
1.41	0.4	0	-.0025	.1326	-.0226	.0000	.0013	.0041	.1326
1.41	6.5	0	.2227	.1259	.0204	-.0002	-.0023	.0073	.1503
1.41	6.5	-4	.2253	.1237	.0215	-.0055	.0522	.2482	.1654
1.41	6.5	-8	.2099	.1207	.0254	-.0096	.0998	.5622	.2206
1.41	6.5	0	.2252	.1251	.0204	-.0001	-.0023	.0073	.1498
2.01	0.4	0	-.0282	.1422	-.0342	-.0001	-.0004	-.0076	.1420
2.01	0.4	-4	-.0282	.1427	-.0351	-.0005	.0525	.2604	.1603
2.01	0.4	-8	-.0517	.1464	-.0371	.0001	.0964	.5804	.2254
2.01	0.4	0	-.0219	.1461	-.0342	-.0001	-.0005	-.0102	.1459
2.01	6.5	0	.2531	.1291	.0014	-.0001	-.0022	-.0076	.1569
2.01	6.5	-4	.2594	.1296	.0006	.0048	.0420	.2966	.1784
2.01	6.5	-8	.2359	.1304	.0021	-.0075	.0729	.6914	.2510
2.01	6.5	0	.2562	.1291	.0012	.0000	-.0023	-.0101	.1573

TABLE X.- PRESSURE COEFFICIENTS FOR FUSELAGE ALONE



(a) $M=1.41$

$\frac{x_b}{l_b}$.160	.200	.240	.320	.360	.400	.440	.480	.520	.560	.600	.640	.680	.720	.760	.800	.840	.880	.920	.960
$\alpha=0.4^\circ; \beta=0^\circ$																				
0	.223	.171	.144	.089	.058	.028	.014	.003	-.021	-.036	-.064	-.081	-.080	-.070	-.057	-.048	-.039	-.030	-.025	
30	.212	.164	.131	.076	.058	.025	.008	-.003	-.030	-.052	-.059	-.070	-.084	-.071	-.059	-.052	-.049	-.047	-.034	-.024
$\alpha=0.4^\circ; \beta=-4^\circ$																				
0	.209	.155	.128	.076	.042	.011	.000	-.013	-.036	-.053	-.073	-.100	-.077	-.102	-.092	-.079	-.069	-.060	-.049	-.041
30	.223	.187	.162	.107	.086	.051	.030	.015	-.021	-.039	-.047	-.063	-.083	-.071	-.065	-.063	-.061	-.061	-.048	-.036
-30	.167	.119	.082	.024	.008	-.024	-.034	-.044	-.070	-.088	-.091	-.097	-.106	-.091	-.076	-.065	-.060	-.053	-.045	-.035
$\alpha=0.4^\circ; \beta=-8^\circ$																				
0	.177	.117	.099	.042	.004	-.030	-.043	-.056	-.081	-.097	-.122	-.148	-.156	-.160	-.149	-.138	-.126	-.115	-.105	-.101
30	.215	.195	.172	.123	.103	.068	.040	.022	-.006	-.036	-.050	-.070	-.092	-.089	-.088	-.093	-.097	-.099	-.088	-.075
-30	.107	.058	.014	-.036	-.052	-.081	-.094	-.098	-.122	-.137	-.139	-.139	-.144	-.122	-.100	-.087	-.077	-.069	-.058	-.053
$\alpha=6.5^\circ; \beta=0^\circ$																				
0	.129	.089	.071	.026	.000	-.019	-.028	-.035	-.054	-.064	-.081	-.093	-.085	-.080	-.060	-.046	-.037	-.030	-.019	-.014
30	.129	.088	.076	.026	.011	-.011	-.025	-.038	-.058	-.072	-.077	-.090	-.096	-.081	-.069	-.059	-.051	-.044	-.030	-.022
$\alpha=6.5^\circ; \beta=-4^\circ$																				
0	.116	.072	.056	.012	-.019	-.036	-.046	-.052	-.071	-.081	-.099	-.111	-.104	-.098	-.076	-.063	-.052	-.043	-.034	-.029
30	.128	.092	.079	.034	.018	-.011	-.029	-.045	-.066	-.085	-.096	-.113	-.127	-.120	-.112	-.108	-.101	-.091	-.089	-.063
-30	.101	.062	.046	-.001	-.011	-.031	-.042	-.049	-.068	-.076	-.080	-.086	-.088	-.071	-.057	-.045	-.038	-.032	-.020	-.013
$\alpha=6.5^\circ; \beta=-8^\circ$																				
0	.072	.026	.017	-.031	-.064	-.087	-.098	-.103	-.123	-.133	-.151	-.167	-.160	-.152	-.125	-.105	-.087	-.072	-.061	-.054
30	.112	.088	.077	.031	.016	-.018	-.038	-.061	-.082	-.109	-.127	-.144	-.168	-.170	-.175	-.175	-.164	-.144	-.128	-.128
-30	.042	.013	-.006	-.045	-.051	-.070	-.076	-.078	-.093	-.096	-.099	-.099	-.099	-.082	-.069	-.062	-.059	-.060	-.054	-.054
$\alpha=-6.0^\circ; \beta=0.3^\circ$																				
0	.320	.269	.235	.170	.137	.097	.079	.066	.036	.014	-.018	-.045	-.051	-.059	-.054	-.050	-.041	-.034	-.022	-.015
30	.301	.244	.195	.125	.103	.061	.037	.025	.013	-.039	-.049	-.063	-.082	-.077	-.069	-.061	-.056	-.048	-.048	
$\alpha=-3.0^\circ; \beta=0.3^\circ$																				
0	.266	.214	.186	.127	.096	.057	.041	.032	.005	-.014	-.045	-.068	-.071	-.076	-.068	-.058	-.047	-.038	-.026	-.020
30	.251	.198	.158	.096	.076	.039	.020	.007	-.025	-.047	-.056	-.070	-.086	-.079	-.068	-.055	-.050	-.048	-.039	-.038
$\alpha=3.0^\circ; \beta=0.3^\circ$																				
0	.178	.130	.108	.061	.031	.004	-.007	-.014	-.033	-.044	-.068	-.084	-.081	-.081	-.064	-.048	-.039	-.030	-.020	-.015
30	.172	.128	.101	.052	.036	.007	-.008	-.018	-.040	-.058	-.064	-.078	-.090	-.076	-.063	-.056	-.045	-.041	-.031	-.022

CONFIDENTIAL

CONFIDENTIAL

NACA RM L55H23

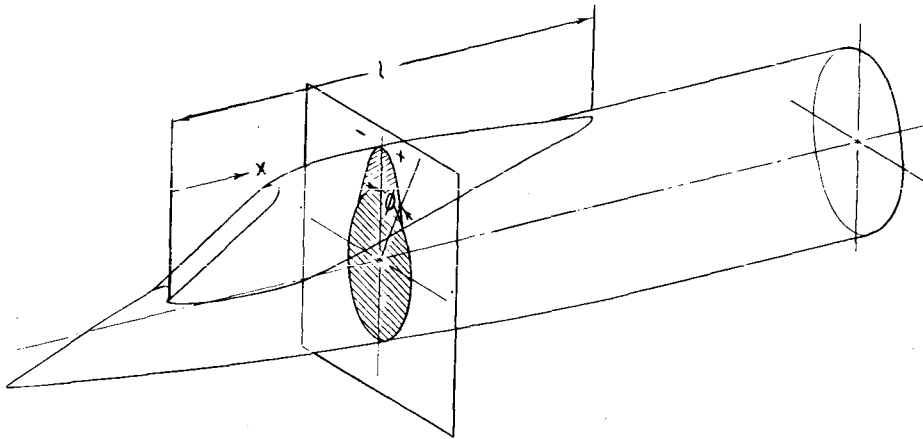
TABLE X.- PRESSURE COEFFICIENTS FOR FUSELAGE ALONE - Concluded

(a) $M=1.41$

$\frac{x}{b}$ θ, deg	.160	.200	.240	.320	.360	.400	.440	.480	.520	.560	.600	.640	.680	.720	.760	.800	.840	.880	.920	.960
$\alpha=9.0^\circ; \beta=0.3^\circ$																				
0	.096	.059	.041	.000	-.023	-.042	-.045	-.046	-.042	-.068	-.085	-.095	-.087	-.080	-.058	-.043	-.035	-.028	-.019	-.017
30	.096	.062	.043	.002	-.008	-.035	-.045	-.054	-.072	-.085	-.090	-.103	-.110	-.092	-.079	-.066	-.055	-.049	-.038	-.032
$\alpha=12.0^\circ; \beta=0.3^\circ$																				
0	.068	.035	.021	-.016	-.036	-.052	-.053	-.051	-.065	-.071	-.085	-.094	-.084	-.076	-.055	-.042	-.034	-.026	-.017	-.014
30	.068	.037	.020	-.017	-.027	-.051	-.061	-.069	-.084	-.097	-.104	-.113	-.116	-.097	-.082	-.069	-.060	-.056	-.048	-.045
(b) $M=2.01$																				
$\alpha=0.4^\circ; \beta=0^\circ$																				
0	.206	.170	.159	.111	.085	.063	.046	.035	.014	.003	-.016	-.034	-.042	-.046	-.039	-.033	-.030	-.025	-.018	-.015
30	.202	.170	.145	.099	.081	.054	.039	.026	.007	-.010	-.017	-.025	-.039	-.039	-.035	-.030	-.027	-.025	-.021	-.019
$\alpha=0.4^\circ; \beta=-4^\circ$																				
0	.199	.158	.148	.098	.073	.046	.033	.020	.000	-.013	-.031	-.050	-.060	-.066	-.061	-.059	-.053	-.051	-.046	-.041
30	.213	.192	.167	.129	.112	.079	.061	.047	.026	.004	-.006	-.018	-.035	-.036	-.035	-.036	-.037	-.040	-.036	-.032
-30	.162	.125	.096	.051	.037	.011	.000	-.012	-.028	-.045	-.051	-.057	-.065	-.061	-.055	-.047	-.040	-.036	-.032	-.029
$\alpha=0.4^\circ; \beta=-8^\circ$																				
0	.188	.143	.130	.073	.051	.029	.003	-.008	-.028	-.043	-.065	-.083	-.098	-.105	-.110	-.113	-.116	-.121	-.118	-.113
30	.222	.208	.196	.153	.141	.103	.085	.066	.045	.022	.006	-.007	-.028	-.034	-.037	-.040	-.045	-.049	-.050	-.054
-30	.116	.076	.039	-.004	-.020	-.038	-.054	-.061	-.075	-.090	-.099	-.103	-.106	-.099	-.088	-.076	-.068	-.063	-.055	-.052
$\alpha=6.5^\circ; \beta=-4^\circ$																				
0	.110	.075	.069	.029	.009	-.012	-.022	-.022	-.042	-.053	-.066	-.076	-.079	-.076	-.065	-.058	-.050	-.043	-.036	-.033
30	.117	.100	.089	.054	.041	.013	.001	-.013	-.025	-.045	-.054	-.065	-.083	-.084	-.083	-.084	-.083	-.083	-.076	-.069
-30	.092	.066	.049	.017	.003	-.012	-.020	-.026	-.037	-.047	-.051	-.054	-.060	-.053	-.043	-.035	-.029	-.027	-.021	-.017
$\alpha=6.5^\circ; \beta=-8^\circ$																				
0	.082	.049	.036	-.008	-.025	-.052	-.064	-.070	-.082	-.096	-.112	-.124	-.133	-.132	-.115	-.098	-.082	-.073	-.069	-.063
30	.115	.103	.099	.058	.049	.015	.006	-.012	-.026	-.046	-.050	-.064	-.078	-.077	-.104	-.111	-.120	-.127	-.134	-.143
-30	.050	.025	.006	-.020	-.029	-.044	-.048	-.052	-.059	-.065	-.067	-.069	-.071	-.066	-.058	-.056	-.056	-.060	-.062	-.065
$\alpha=-6.0^\circ; \beta=0.3^\circ$																				
0	.327	.273	.259	.198	.169	.137	.114	.100	.071	.052	.035	.008	-.010	-.019	-.014	-.015	-.013	-.010	-.008	-.005
30	.299	.255	.215	.154	.134	.095	.077	.059	.034	.009	-.006	-.018	-.033	-.032	-.034	-.033	-.031	-.030	-.024	-.024
$\alpha=-3.0^\circ; \beta=0.3^\circ$																				
0	.270	.220	.210	.154	.124	.099	.082	.068	.045	.026	.004	-.014	-.025	-.033	-.027	-.026	-.023	-.022	-.015	-.010
30	.253	.213	.180	.125	.104	.076	.059	.045	.023	-.002	-.012	-.022	-.034	-.035	-.034	-.033	-.031	-.031	-.025	-.023
$\alpha=3.0^\circ; \beta=0.3^\circ$																				
0	.162	.126	.116	.072	.051	.032	.020	.009	-.009	-.020	-.035	-.050	-.055	-.055	-.048	-.045	-.035	-.030	-.022	-.017
30	.154	.126	.104	.066	.051	.029	.017	.004	-.011	-.028	-.034	-.043	-.055	-.053	-.048	-.039	-.035	-.033	-.026	-.021
$\alpha=6.0^\circ; \beta=0.3^\circ$																				
0	.120	.087	.079	.043	.025	.006	-.003	-.010	-.024	-.033	-.046	-.057	-.057	-.056	-.046	-.040	-.033	-.026	-.019	-.016
30	.114	.091	.074	.043	.030	.010	-.001	-.013	-.023	-.037	-.042	-.052	-.062	-.058	-.051	-.045	-.040	-.036	-.030	-.023
$\alpha=9.0^\circ; \beta=0.3^\circ$																				
0	.089	.057	.051	.021	.004	-.021	-.019	-.023	-.034	-.040	-.051	-.059	-.058	-.056	-.046	-.039	-.031	-.026	-.020	-.017
30	.083	.062	.049	.021	.010	-.009	-.017	-.026	-.036	-.049	-.054	-.063	-.070	-.066	-.057	-.050	-.046	-.041	-.035	-.028
$\alpha=12.0^\circ; \beta=0.3^\circ$																				
0	.062	.036	.031	.006	-.008	-.022	-.026	-.027	-.038	-.042	-.052	-.060	-.057	-.056	-.045	-.038	-.032	-.026	-.019	-.016
30	.059	.042	.032	.006	-.004	-.022	-.026	-.036	-.047	-.057	-.063	-.070	-.079	-.072	-.064	-.059	-.053	-.051	-.046	-.045

CONFIDENTIAL

TABLE XI.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED FLAT-WINDSHIELD CANOPY

(a) $M=1.41$

$\frac{x}{l}$ ϕ, deg	.000	.001	.004	.006	.017	.046	.092	.164	.216	.238	.240	.260	.311	.356	.440	.600	.828	.988
$\alpha=0.4^\circ; \beta=0^\circ$																		
P.L.	.497						.254	.149	.156			.104	-.039	-.092		-.025	.046	
57½	.497				.454		.216	.173					-.039					
45	-.001												-.059	-.092				
30		.801		.565	.432	.313						.100	-.159	-.122				
15						.547	.287	.176				.027	-.121	-.172	-.183		-.025	
10							.505											
7								.343										
3							.556		.244	-.041	-.160	-.188	-.157	-.102	-.002			
0			.820			.635				.220	-.180		-.094		.046			
$\alpha=0.4^\circ; \beta=-4^\circ$																		
P.L.	.632						.361	.240	.246			.163	.023	-.039		-.104	.022	
57½	.632				.570		.311	.272					.023					
45	-.017												.017	-.039				
30		.814		.643	.563	.422						.191	-.089	-.072				
15						.616	.419	.322				.138	-.059	-.126	-.155		-.104	
10							.568											
7								.440										
3							.571		.276	.020	-.148	-.205	-.204	-.168	-.032			
0			.818			.626				.231	-.200			-.156	.022			
-3							.531		.212	-.087	-.152	-.236	-.192	-.148	-.023			
-7								.250										
-10							.431											
-15						.454	.151	.023		-.113	-.207	-.226	-.215		-.002			
-30		.782		.476	.257	.186				-.005	-.220	-.161						
-45	.008										-.132	-.135						
-57½	.305		.321		.107	.073					-.095							
-P.L.	.305				.155	.063	.074			.030	-.095	-.135		-.002	.017			

CONFIDENTIAL

CONFIDENTIAL

NACA RM L55H23

TABLE XI.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED FLAT-WINDSHIELD CANOPY - Continued

(a) $M=1.41$

α/β	.000	.001	.004	.006	.017	.040	.090	.140	.215	.233	.240	.260	.311	.350	.440	.600	.928	.988
$\alpha=0.4^\circ; \beta=-8^\circ$																		
$P_{t,Le}$.738					.666	.348	.345				.270		.093	.024		-.251	-.080
$57\frac{1}{2}$.738				.671	.666	.345	.385						.093				
45	-.028												.099		.024			
30		.815		.796	.662	.521						.281		-.011	-.006			
15						.656	.532	.440				.236	-.003	-.074	-.112		-.251	
10							.609											
7								.503										
3							.575		.296			.071	-.143	-.242	-.310	-.343	-.088	
0		.827				.607				.234		-.225			-.358		-.020	
-3							.512		.203		-.095	-.157	-.380	-.315	-.296	-.104		
-7								.191										
-10							.365											
-15						.352	-.023	-.093			-.245	-.313	-.311	-.257			-.097	
-30		.780		.418	-.143	.056					-.115		-.269	-.194				
-45		.023											-.194		-.167			
$-57\frac{1}{2}$.057		.130		.002	-.018							.147				
$-P_{t,Le}$.057				.059	-.013	.003				-.035	-.147	-.167		-.037	-.021	
$\alpha=6.5^\circ; \beta=0^\circ$																		
$P_{t,Le}$.666					.262	.152	.160				.108	.061	-.034	-.122		-.053	.007
$57\frac{1}{2}$.666				.417	.199	.168							-.034				
45	.692							.170						-.040	-.127			
30		.683		.666	.368	.261						.084		-.158	-.119			
15						.432	.212	.142				.003	-.161	-.213	-.206		-.053	
10							.270											
7								.240										
3							.405		.134		-.127	-.227	-.236	-.188	-.092			
0		.694				.493				.107		-.243			-.088	-.052	.007	
$\alpha=6.5^\circ; \beta=-4^\circ$																		
$P_{t,Le}$.570					.340	.233	.236				.177	.124	.022	-.074		-.120	.003
$57\frac{1}{2}$.570				.512	.285	.259							.022				
45	.667							.251						.027	-.074			
30		.687		.534	.474	.358						.158		-.099	-.104			
15						.491	.326	.260				.092	-.117	-.179	-.201		-.120	
10							.427											
7								.321										
3							.420		.162		-.076	-.223	-.277	-.236	-.138			
0		.690				.488			.115		-.273			-.116	-.038	.003		
-3							.383		.103		-.171	-.293	-.269	-.212	-.119			
-7								.142										
-10							.297											
-15						.343	.085	-.005			-.113	-.216	-.242	-.211		-.026		
-30		.668		.391	.225	.152					.006		-.214	-.172				
-45	.619							.090					-.102		-.159			
$-57\frac{1}{2}$.313		.313		.112	.088							-.063				
$-P_{t,Le}$.313				.176	.077	.088				.042	.000	-.083	-.159		-.026	-.011
$\alpha=6.5^\circ; \beta=-8^\circ$																		
$P_{t,Le}$.636					.420	.324	.321				.258	.176	.085	-.017		-.209	-.046
$57\frac{1}{2}$.636				.592	.179	.354							.085				
45	.656							.336						.098	-.017			
30		.680		.579	.555	.441						.237		-.033	-.052			
15						.518	.420	.355				.159	-.080	-.140	-.169		-.209	
10							.460											
7								.369										
3							.427		.179		-.027	-.212	-.320	-.408	-.257			
0		.696				.468				.120		-.103			-.232	-.035	-.046	
-3							.367		.091		-.180	-.269	-.439	-.314	-.212			
-7								.083										
-10							.233											
-15						.279	-.066	-.141			-.242	-.285	-.292	-.237		-.056		
-30		.665		.320	-.089	.038					-.077		-.258	-.172				
-45	.589							.011					-.153		-.191			
$-57\frac{1}{2}$.125		.195		.025	.015							-.127				
$-P_{t,Le}$.125				.088	.015	.026			-.094	-.049	-.127	-.191		-.056	-.071	

CONFIDENTIAL

TABLE XI. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED FLAT-WINDSHIELD CANOPY - Continued

(b) $M=2.01$

x/l	.000	.001	.004	.006	.017	.046	.092	.164	.216	.236	.240	.260	.311	.356	.440	.600	.828	.968
ϕ, deg																		
$\alpha=0.4^\circ; \beta=0^\circ$																		
P_o/L_o		.380					.221		.142			.115	.093	.039	.002		-.012	.045
$57\frac{1}{2}$.380					.211	.157						.039				
45		.771			.429				.166				.022		.002			
30			.878	.624	.425	.331						.125		-.029	-.047			
15						.565	.340	.233				.116	-.007	-.051	-.075		-.012	
10							.544											
7								.435										
3							.603		.365			.118	-.056	-.081	-.065	-.065	.015	
0			.908			.657				.349		-.056			-.061	.015	.045	
$\alpha=0.4^\circ; \beta=-4^\circ$																		
P_o/L_o		.490					.319		.216			.165	.156	.097	.051		-.096	.021
$57\frac{1}{2}$.490			.512		.306	.248						.097				
45		.725							.253				.087		.051			
30			.852	.680	.549	.440						.218		.022	.002			
15						.638	.433	.336				.200	.049	-.003	-.048		-.096	
10							.594											
7								.494										
3							.609		.382			.144	-.026	-.084	-.084	-.101	-.027	
0			.891			.652				.349		-.075				-.099	-.014	.021
-3							.591		.347			.095	-.094	-.114	-.090	-.096	-.010	
-7								.379										
-10								.490										
-15						.485	.255	.135				.023	-.073	-.109	-.115		.019	
-30			.874	.574	.173	.228						.038		-.086	-.096			
-45		.778						.081					-.031		-.041			
-57 $\frac{1}{2}$.237		.311	.118	.075								-.010				
- P_o/L_o		.237			.129	.074				.051	.033	-.010	-.041			.019	.012	
$\alpha=0.4^\circ; \beta=-8^\circ$																		
P_o/L_o		.579					.420		.299			.263	.226	.158	.101		-.122	-.076
$57\frac{1}{2}$.579			.580		.399	.341						.158				
45		.652							.347				.153		.101			
30			.784	.716	.647	.537						.306		.075	.051			
15						.696	.519	.430				.285	.098	.040	-.017		-.122	
10							.629											
7								.539										
3							.612		.393			.167	.001	-.080	-.139	-.194	-.096	
0			.864			.658				.342		-.069				-.195	-.094	-.076
-3							.586		.341		.087	-.107	-.173	-.160	-.179	-.115		
-7								.350										
-10							.455											
-15						.448	-.035	.050				-.018	-.121	-.158	-.158		-.040	
-30		.869	.562	-.037	.152							-.051		-.139	-.137			
-45		.770						-.002					-.085		-.085			
-57 $\frac{1}{2}$.078	.189	.028	.000									-.053				
- P_o/L_o		.078			.047	.014				-.003	-.016	-.053	-.085			-.040	-.055	
$\alpha=6.5^\circ; \beta=0^\circ$																		
P_o/L_o		.337					.214		.131			.108	.081	.022	-.027		-.049	-.022
$57\frac{1}{2}$.337			.342		.166	.145						.022				
45		.614						.152					.014		-.027			
30			.687	.490	.341	.262						.102		-.050	-.064			
15						.424	.240	.157				.059	-.050	-.087	-.114		-.049	
10							.398											
7								.305										
3							.435		.232			.025	-.121	-.138	-.105	-.076	-.024	
0			.697			.487				.215		-.126				-.071	-.023	-.022

TABLE XI.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED FLAT-WINDSHIELD CANOPY - Concluded

(b) $M=2.01$

x/l θ, deg	.000	.001	.004	.006	.017	.046	.092	.164	.216	.238	.240	.260	.311	.356	.440	.600	.828	.988
$\alpha=6.5^\circ; \beta=-4^\circ$																		
PoL.	.420				.297		.136					.168	.138	.068	.011		-.164	-.060
57½	.420			.407	.262	.217								.068				
45	.562						.224						.069		.011			
30		.655		.528	.440	.350						.170		-.008	-.029			
15					.485	.322	.247					.137	-.011	-.054	-.096		-.164	
10						.439												
7							.357											
3						.438		.244				.046	-.096	-.145	-.153	-.131	-.073	
0		.681			.479				.211				-.137			-.108	-.105	-.060
-3						.424		.215				.004	-.158	-.167	-.130	-.101	-.083	
-7							.241											
-10							.339											
-15					.346	.155	.058					-.027	-.111	-.137	-.133		-.020	
-30		.686		.440	.151	.164						.022		-.099	-.106			
-45	.618						.073						-.039		-.068			
-57½	.221			.250	.102	.067								-.024				
-PoL.	.221				.134	.071						.048	.025	-.024	-.068		-.020	-.091
$\alpha=6.5^\circ; \beta=-8^\circ$																		
PoL.	.500				.386		.278					.248	.210	.135	.066		-.163	-.076
57½	.500			.464	.351	.305								.135				
45	.499						.314						.139		.066			
30		.583		.550	.522	.441						.254		.054	.026			
15					.536	.405	.338					.216	.034	-.018	-.059		-.163	
10						.464												
7							.394											
3						.450		.263				.077	-.062	-.133	-.190	-.202	-.113	
0		.662			.494				.214				-.125			-.195	-.109	-.076
-3					.419		.209			-.001	-.168	-.221	-.207	-.198	-.168			
-7						.210												
-10						.297												
-15					.300	-.097	-.052					-.082	-.163	-.186	-.160		-.060	
-30		.666		.422	-.064	.083						-.054		-.138	-.134			
-45	.570						.004						-.083		-.100			
-57½	.071			.147	.027	.006								-.063				
-PoL.	.071				.057	.017						-.004	-.022	-.063	-.100		-.060	-.083

TABLE XII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED VEE-WINDSHIELD CANOPY

(a) $M=1.41$

$\frac{x}{\lambda}$.008	.044	.076	.124	.140	.164	.212	.244	.260	.295	.311	.324	.345	.356	.440	.600	.828	.988
ϕ, deg																		
$\alpha=0.4^\circ; \beta=0^\circ$																		
P.L.	.463	.404	.380	.390	.376	.367	.217	.181	.061					.038	-.073	-.012	-.043	.069
57½						.367								.038				
51						.408	.228											
48							.290											
45				.390				.173	-.071						-.073			
41								.258										
34											-.231							
30			.380				.368			.124	-.282		-.266	-.227	-.012			
15		.404		.444		.425		.426	.318		-.195	-.346	-.295	-.064	-.043			
9									.299									
3	.463	.456	.450			.493		.471	.186	.114				-.064	-.186	-.165	-.022	.069
1							.513											
0																-.155	-.020	
$\alpha=0.4^\circ; \beta=-4^\circ$																		
P.L.	.548	.490	.487	.490	.469	.453	.303	.258	.133					.105	-.014	.003	-.149	.043
57½						.453								.105				
51						.501	.376											
48							.349											
45				.490			.289	.282	.006						-.014			
41								.356										
34									-.130									
30			.487			.446			.167	-.226			-.150	-.161	.003			
15		.490		.527	.519		.491	.374		-.174	-.329	-.206	-.050	-.149				
9								.346										
3	.548	.526	.527		.575		.488	.085	.028					-.093	-.200	-.210	-.060	.043
1						.559												
0																-.240	-.050	
-1						.353												
-3	.345	.351	.344		.383		.383	.205	.121					-.076	-.199	-.245	-.043	.033
-9									.243									
-15		.307		.338	.324		.345	.256		-.226	-.366	-.335	-.084	-.001				
-30			.271			.282		.070	-.316		-.367	-.288	-.034					
-34									-.313									
-41							.158											
-45			.264			.084	.066	-.150						-.130				
-48						.143												
-51						.276	.123											
-57½						.254								-.026				
-P.L.	.345	.307	.271	.264	.252	.254	.126	.095	-.009					-.026	-.130	-.034	-.001	.033

TABLE XII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED VEE-WINDSHIELD CANOPY - Continued

(a) $M=1.41$

x/l deg	.008	.044	.076	.124	.140	.164	.212	.244	.260	.295	.311	.324	.345	.356	.440	.600	.828	.988
$\alpha=0.4^\circ; \beta=-8^\circ$																		
P.L.	.606	.582	.588	.582	.566	.538	.391		.343	.209				.178	.050	.012	-.205	-.612
57 1/2						.538								.178				
51						.583	.425											
48							.446											
45				.582				.407	.390	.088					.050			
41									.463									
34										-.019								
30			.588			.528				.210	-.123			-.050	-.090	.012		
15		.582		.617		.596			.560	.416		-.152	-.310	-.195	-.040	-.205		
9										.377								
3	.606	.579	.567			.612			.451	-.083	-.124			-.172	-.231	-.271	-.113	-.012
1						.554												
0							.015									-.316	-.106	
-1																		
-3	.177	.107	.006			-.001			.129	.093	.043			-.135	-.236	-.348	-.134	
-9										.224								
-15		.197		.230		.210			.257	.164		-.252	-.380	-.363	-.129	-.043		
-30			.158				.195			.006	-.356		-.432	-.348	-.052			
-45										.359								
-48									.065									
-51				.146			-.020	-.036		-.227				-.187				
-57 1/2						.036												
-P.L.	.177	.197	.158	.146	.135	.130	.033		.013	-.075				-.088	-.187	-.052	-.043	
$\alpha=6.5^\circ; \beta=0^\circ$																		
P.L.	.341	.316	.316	.333	.328	.329	.212		.176	.068				.040	-.069	-.081	-.070	
57 1/2						.329								.040				
51						.364	.229											
48							.243											
45				.333				.202	.191	-.044					-.069			
41									.239									
34										-.153								
30			.316			.316				.062	-.255		-.195	-.196	-.081			
15		.316		.362					.345	.273		-.217	-.368	-.274	-.116	-.070		
9										.213								
3	.341	.348	.339	.362		.375			.347	.074	.007			-.130	-.247	-.113	-.044	
1						.385												
0																-.111	-.042	

TABLE XII. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED VEE-WINDSHIELD CANOPY - Continued

		(a) $M=1.41$																	
α/β		.008	.044	.076	.124	.140	.164	.212	.244	.260	.295	.311	.324	.345	.356	.440	.600	.828	.988
		$\alpha=6.5^\circ; \beta=-4^\circ$																	
P.L.		.416	.399	.406	.412	.410	.405	.298	.244	.128					.099	-.019	-.113	-.167	
57½							.405								.099				
51							.441	.315											
45							.330												
41					.412			.295	.274	.031						-.019			
34									.330										
30										-.061									
15		.399		.406			.388			.105	-.162			-.107	-.142	-.113			
9			.399		.443				.401	.313			-.202	-.354	-.266	-.122	-.167		
3		.416	.421	.405	.428		.450		.354	-.046	-.087								
1							.421												
0																	-.178	-.094	
-1							.166												
-3		.223	.246	.234	.259		.277		.227	.077	.019				-.147	-.262	-.141	-.095	
-9										.174									
-15			.221		.269				.270	.210				-.250	-.389	-.355	-.089	-.043	
-30				.211				.240		.017	-.322				-.294	-.252	-.084		
-34										-.253									
-41									.150										
-45					.238			.111	.101	-.113						-.119			
-48							.158												
-51							.283	.143											
-57½							.249									-.016			
-P.L.		.223	.221	.211	.238	.240	.249	.146	.111	.006					-.016	-.119	-.084	-.043	
		$\alpha=6.5^\circ; \beta=-8^\circ$																	
P.L.		.463	.487	.495	.495	.495	.491	.284	.321	.194					.162	.034	-.119	-.265	
57½							.491								.162				
51							.519	.404											
45						.495		.390	.362	.113						.034			
41								.416											
34										.032									
30					.455		.456		.154	-.065				-.014	-.083	-.119			
15			.467		.595			.467	.351				-.186	-.241	-.267	-.125	-.285		
9									.291										
3		.463	.463	.436	.463		.479		.325	-.194	-.218				-.283	-.263	-.266	-.137	
1							.410												
0																	-.341	-.157	
-1							-.036												
-3		.063	-.049	-.116	-.065		-.092		.047	-.002	-.037				-.200	-.294	-.254	-.247	
-9										.158									
-15			.118		.180				.205	.158				-.265	-.466	-.392	-.092	-.090	
-30				.105			.166			-.023	-.249			-.366	-.298	-.091			
-34										-.338									
-41									.067										
-45					.132			.026	.014	-.177					-.166				
-48							.071												
-51							.182	.060											
-57½							.164								-.073				
-P.L.		.063	.118	.105	.132	.139	.164	.073	.050	-.052				-.073	-.166	-.091	-.090		

TABLE XII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED VEE-WINDSHIELD CANOPY - Continued

(b) $M=2.01$

α/γ β, deg	.008	.044	.076	.124	.140	.164	.212	.244	.260	.295	.311	.324	.345	.356	.440	.600	.828	.988
$\alpha=0.4^\circ; \beta=0^\circ$																		
P.L.	.398	.379	.344		.312	.297	.191		.163		.077			.065	.018	-.028	-.049	.063
57 $\frac{1}{2}$.297								.065				
51						.335	.201											
48						.227												
45								.325	.167	-.003					.013			
41								.253										
34										-.056								
30			.344			.354				.217	-.057			-.115	-.105	-.028		
15		.379		.399		.374			.418	.395		.050	-.072	-.162	-.020	-.049		
9										.380								
3	.398	.431	.439	.455		.469			.466	.226	.163			.048	-.015	-.115	.001	.063
1						.492												
0																-.113	.006	
$\alpha=0.4^\circ; \beta=-4^\circ$																		
P.L.	.507	.474	.445		.413	.397	.281		.243		.149			.133	.072	-.021	-.079	.034
57 $\frac{1}{2}$.397								.133				
51						.439	.300											
48						.327												
45								.424	.258	.068					.072			
41								.345										
34										.004								
30			.445			.440				.277	-.023		-.047	-.046	-.021			
15		.474		.492		.463			.490	.460		.081	-.050	-.145	-.005	-.079		
9										.447								
3	.507	.503	.503	.524		.540			.494	.152	.103			.003	-.042	-.096	-.078	.034
1						.544												
0																-.108	-.047	
-1						.325												
-3	.271	.328	.344	.361		.375			.397	.235	.170			.034	-.024	-.130	-.036	-.022
-7										.310								
-15		.279		.283		.286			.332	.322		.009	-.107	-.158	-.100	.009		
-30			.246			.264				.159	-.095		-.174	-.173	-.025			
-34										-.098								
-41									.168									
-45								.217	.068	-.075					-.039			
-48							.137											
-51						.225	.108											
-57 $\frac{1}{2}$.194									.001			
-F.L.	.271	.279	.246	.267	.194	.102		.085		.010				.001	-.039	-.025	.009	-.022

TABLE XII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED VEE-WINDSHIELD CANOPY - Continued

(b) $M=2.01$

$\frac{x}{l}$ ϕ, deg	.008	.044	.076	.124	.140	.164	.212	.244	.260	.295	.311	.324	.345	.356	.440	.600	.828	.988	
$\alpha=0.4^\circ; \beta=-8^\circ$																			
P.L.	.590	.577	.554		.512	.498	.377		.333		.229				.207	.136	.023	-.089	-.050
57½						.498									.207				
51						.553	.407												
48							.433												
45								.527	.361		.148					.136			
41									.446										
34											.089								
30			.554				.531				.342	.024		.036	.023	.023			
25		.577		.581		.549			.567		.533		.119	-.023	-.127	-.005	-.089		
9											.507								
3	.590	.566	.552	.581		.599			.513	.051	-.006				-.079	-.108	-.116	-.142	-.050
1							.572												
0																	-.134	-.121	
-1							.077												
-3	.159	.210	.140	.256		.220			.166	.158	.114				-.001	-.062	-.139	-.123	-.049
-9											.248								
-15		.192		.220		.204			.242		.246		-.027	-.132	-.146	-.185	-.046		
-30			.168				.182				.105	-.117		-.202	-.211	-.054			
-34											-.132								
-41									.095										
-45								.121	-.018		-.132				-.084				
-48							.051												
-51						.126	.022												
-57½						.099									-.053				
-P.L.	.159	.192	.168		.108	.099	.026		.016		-.046				-.053	-.084	-.054	-.046	-.049
$\alpha=6.5^\circ; \beta=0$																			
P.L.	.277	.281	.267		.265	.270	.194		.165		.079				.055	-.007	-.081	-.082	.015
57½						.270									.055				
51						.301	.202												
48							.219												
45								.268	.169		.007				-.007				
41									.224										
34											-.050								
30			.267				.291				.157	-.082		-.099	-.086	-.081			
25		.281		.309		.290			.319		.296		.004	-.109	-.185	-.051	-.082		
9											.280								
3	.277	.315	.319	.334		.343			.333	.110	.057				-.024	-.071	-.126	-.033	.015
1							.354												
0																	-.129	-.027	

TABLE XII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED VEE-WINDSHIELD CANOPY - Concluded

(b) $M=2.01$

α/β	.00R	.04L	.076	.12L	.140	.16L	.212	.24L	.260	.295	.311	.32L	.345	.356	.440	.600	.828	.988
$\alpha=6.5^\circ; \beta=-4^\circ$																		
P.L.	.367	.366	.357		.349	.353	.270		.234		.144				.116	.043	-.044	-.134
57 $\frac{1}{2}$.353									.116			-.007
51						.391	.288											
48							.305											
45								.352	.248		.072				.043			
42									.305									
34											.021							
30			.357				.365			.207	-.040		-.035	-.039	-.044			
15		.368		.390		.365			.380	.354		.033	-.087	-.175	-.077	-.134		
9										.338								
3	.367	.380	.376	.390		.402			.357	.032	-.006			-.104	-.100	-.115	-.078	-.007
1							.399											
0																-.128	-.076	
-1							.208											
-3	.174	.222	.234	.254		.256			.264	.124	.076			-.029	-.077	-.144	-.063	-.081
-9										.226								
-15		.193		.223		.216			.249	.237		-.026	-.128	-.196	-.060	-.022		
-30			.180			.218				.107	-.112		-.152	-.130	-.089			
-34										-.103								
-42									.150									
-45								.167	.094		-.044				-.044			
-48							.112											
-51						.219	.126											
-57 $\frac{1}{2}$.192								.008				
-P.L.	.174	.193	.180		.167	.192	.129		.107	.031				.008	-.044	-.089	-.022	-.081
$\alpha=6.5^\circ; \beta=-8^\circ$																		
P.L.	.438	.465	.460		.428	.435	.355		.313		.215				.180	.100	.003	-.154
57 $\frac{1}{2}$.435									.180			-.091
51						.477	.382											
48							.398											
45								.437	.345		.145				.100			
42									.396									
34											.105							
30			.460			.450				.270	.031		.037	.019	.003			
15		.465		.477		.446		.452		.419		.069	-.061	-.132	-.083	-.154		
9										.391								
3	.438	.446	.422	.439		.452			.380	-.055	-.073			-.143	-.174	-.163	-.138	-.091
1						.416												
0																-.154	-.173	
-1							-.012											
-3	.057	.023	-.041	.060		.007			.046	.032	.005			-.089	-.152	-.157	-.214	-.104
-9										.170								
-15		.101		.140		.127		.159		.170		-.062	-.155	-.207	-.112	-.064		
-30			.089			.136				.055	-.138		-.202	-.181	-.085			
-34										-.147								
-42								.072										
-45							.093	.016		-.102					-.087			
-48						.059												
-51						.125	.045											
-57 $\frac{1}{2}$.107								-.047				
-P.L.	.057	.101	.089		.098	.107	.060		.044	-.026				-.047	-.087	-.085	-.064	-.104

TABLE XIII. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED ROUND-WINDSHIELD CANOPY

(a) $M=1.41$

α/γ θ, deg	.002	.036	.088	.124	.160	.212	.260	.311	.356	.400	.600	.828	.988
$\alpha=0.4^\circ; \beta=-4^\circ$													
P.L.	.728	.512	.427	.366	.331	.275	.131	.059	.030	-.050	-.023	-.146	.025
57½					.331	.151	.026	.030					
45			.427		.332	.295	.147	-.010	-.061	-.050			
30		.512	.480	.454	.378	.354	.186	-.028	-.119	-.066	-.023		
15		.614	.576	.546	.497	.426	.228	-.056	-.148	-.167	-.053	-.146	
3				.631		.458	.220	-.106	-.193	-.211	-.162	-.061	.025
0	.728		.602		.577		.189		-.217		-.180	-.047	.026
-3				.569		.369	.126	-.154	-.244	-.222	-.197	-.050	.003
-15		.424	.347	.310	.257	.180	.026	-.200	-.253	-.240	-.061	-.031	
-30		.243	.227	.210	.146	.126	.004	-.167	-.244	-.248	-.030		
-45			.198		.120	.103	-.017	-.152	-.199	-.152			
-57½					.131		-.008	-.106	-.095				
-P.L.	.683	.243	.198	.148	.131	.089	.015	-.071	-.095	-.152	-.030	-.031	.003
$\alpha=0.4^\circ; \beta=-8^\circ$													
P.L.	.721	.629	.540	.471	.432	.374	.178	.130	.101	.007	.001	-.229	-.038
57½					.432	.231	.099	.101					
45			.540		.434	.391	.226	.064	.018	.007			
30		.629	.596	.550	.483	.449	.269	.047	-.048	-.201	.001		
15		.670	.648	.619	.583	.513	.302	.007	-.102	-.124	-.057	-.229	
3				.602		.451	.222	-.113	-.205	-.273	-.311	-.119	-.038
0	.721		.539		.527		.148		-.265		-.335	-.114	-.017
-3				.479		.280	.048	-.220	-.312	-.326	-.344	-.159	-.037
-15		.297	.192	.146	.107	.033	-.101	-.290	-.320	-.287	-.104	-.047	
-30		.100	.073	.083	.021	.003	-.089	-.237	-.303	-.240	-.042		
-45			.089		.017	.008	-.094	-.216	-.257	-.190			
-57½					.037		-.077	-.159	-.146				
-P.L.	.627	.100	.089	.045	.037	.010	-.049	-.124	-.146	-.190	-.042	-.047	-.037
$\alpha=6.5^\circ; \beta=-4^\circ$													
P.L.	.603	.441	.389	.359	.321	.273	.085	.059	.024	-.088	-.099	-.176	-.005
57½					.321		.004	.030	.024				
45			.389		.310	.283	.144	-.004	-.042	-.088			
30		.441	.426	.397	.330	.314	.158	-.039	-.132	-.096	-.099		
15		.499	.463	.439	.401	.346	.161	-.108	-.203	-.179	-.103	-.176	
3				.480		.375	.105	-.185	-.265	-.264	-.181	-.089	-.005
0	.603		.457		.435		.071		-.279		-.116	-.093	-.005
-3				.424		.282	.016	-.219	-.296	-.247	-.139	-.129	-.075
-15		.327	.273	.238	.200	.134	-.007	-.225	-.277	-.228	-.054	-.058	
-30		.203	.198	.188	.133	.127	.004	-.166	-.244	-.186	-.071		
-45			.181		.128	.118	.000	-.132	-.163	-.173			
-57½					.146		.015	-.090	-.088				
-P.L.	.551	.203	.181	.165	.146	.111	.036	-.062	-.098	-.173	-.071	-.058	-.075
$\alpha=6.5^\circ; \beta=-8^\circ$													
P.L.	.579	.544	.491	.458	.410	.360	.208	.124	.083	-.037	-.074	-.310	-.047
57½					.410		.007	.096	.083				
45			.491		.398	.360	.216	.041	.023	-.037			
30		.544	.525	.480	.416	.396	.023	.229	-.074	-.044	-.074		
15		.544	.523	.495	.468	.416	.216	-.062	-.168	-.150	-.126	-.210	
3				.442		.362	.096	-.207	-.293	-.262	-.394	-.156	-.047
0	.579		.386		.376		.025		-.244		-.170	-.201	-.040
-3				.332		.184	-.064	-.289	-.371	-.354	-.239	-.294	-.084
-15		.202	.129	.096	.061	-.003	-.171	-.296	-.325	-.261	-.063	-.058	
-30		.067	.072	.074	.031	.026	-.076	-.222	-.267	-.218	-.055		
-45			.080		.043	.038	-.066	-.144	-.211	-.206			
-57½					.066		-.002	-.135	-.133				
-P.L.	.504	.067	.080	.074	.066	.042	.000	-.108	-.133	-.206	-.055	-.050	-.084

CONFIDENTIAL

CONFIDENTIAL

NACA RM L55H23

TABLE XIII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED ROUND-WINDSHIELD CANOPY - Continued

(a) $M=1.41$

α, deg	β, deg	.002	.036	.088	.124	.160	.212	.260	.311	.356	.440	.600	.828	.988
$\alpha = -6.0^\circ; \beta = 0.3^\circ$														
P.L.		.830	.451	.329	.265	.217	.150	.164	-.030	-.043	-.073	.030	-.006	.079
57 $\frac{1}{2}$.217		.159	-.078	-.043				
45				.329		.217	.195	.041	-.119	-.182	-.073			
30			.451	.392	.363	.279	.248	.090	-.101	-.185	-.153	.038		
15			.644	.571	.519	.461	.363	.175	-.094	-.161	-.138	-.018	-.006	
3					.762		.581	.301	-.020	-.130	-.135	-.155	.009	.079
0		.830		.786		.735		.324		-.119		-.127	.017	.095
$\alpha = -3.0^\circ; \beta = 0.3^\circ$														
P.L.		.774	.408	.312	.257	.217	.157	.163	-.029	-.044	-.096	.007	-.031	.058
57 $\frac{1}{2}$.217		.158	-.066	-.044				
45				.312		.215	.193	.045	-.107	-.164	-.096			
30			.408	.366	.343	.265	.241	.086	-.106	-.189	-.143	.007		
15			.582	.516	.471	.414	.328	.148	-.118	-.185	-.162	-.004	-.031	
3					.688		.513	.239	-.070	-.172	-.167	-.153	-.011	.058
0		.774		.708		.661		.257		-.162		-.135	-.003	.070
$\alpha = 0^\circ; \beta = 0.3^\circ$														
P.L.		.719	.375	.301	.253	.219	.165	.167	-.024	-.043	-.112	-.021	-.047	.044
57 $\frac{1}{2}$.219		.161	-.056	-.043				
45				.301		.214	.192	.052	-.097	-.147	-.112			
30			.375	.344	.325	.254	.234	.083	-.109	-.192	-.144	-.021		
15			.529	.463	.432	.381	.301	.125	-.134	-.205	-.181	-.021	-.047	
3					.619		.452	.183	-.110	-.204	-.190	-.138	-.024	.044
0		.719		.635		.592		.198		-.197		-.124	-.018	.057
$\alpha = 3.0^\circ; \beta = 0.3^\circ$														
P.L.		.666	.345	.289	.250	.222	.173	.152	-.021	-.044	-.126	-.053	-.064	.029
57 $\frac{1}{2}$.222		.145	-.049	-.044				
45				.289		.212	.192	.056	-.088	-.133	-.126			
30			.345	.326	.308	.240	.225	.077	-.112	-.194	-.147	-.053		
15			.477	.417	.392	.345	.272	.104	-.151	-.223	-.198	-.043	-.064	
3					.553		.393	.132	-.149	-.236	-.208	-.122	-.035	.029
0		.666		.569		.526		.144		-.230		-.110	-.031	.044
$\alpha = 6.0^\circ; \beta = 0.3^\circ$														
P.L.		.609	.319	.278	.246	.221	.177	.154	-.016	-.046	-.142	-.087	-.082	.013
57 $\frac{1}{2}$.221		.148	-.044	-.046				
45				.278		.208	.187	.057	-.082	-.121	-.142			
30			.319	.307	.290	.224	.213	.070	-.116	-.197	-.156	-.087		
15			.427	.370	.350	.308	.241	.079	-.169	-.243	-.213	-.062	-.082	
3					.487		.330	.082	-.187	-.262	-.223	-.108	-.049	.013
0		.609		.502		.454		.090		-.260		-.100	-.043	.035

CONFIDENTIAL

TABLE XIII. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED ROUND-WINDSHIELD CANOPY - Continued

(a) $M=1.41$

x/l	.002	.036	.088	.124	.160	.212	.260	.311	.356	.440	.600	.828	.988
β, deg													
$\alpha=9.0^\circ; \beta=0.3^\circ$													
P.L.	.553	.297	.267	.243	.220	.181	.156	-.008	-.045	-.151	-.124	-.098	-.009
57½					.220		.151	-.036	-.045				
45			.267	.204	.183	.057	-.077	-.111	-.151				
30		.297	.291	.274	.212	.201	.062	-.118	-.197	-.165	-.124		
15		.383	.329	.315	.273	.213	.056	-.185	-.260	-.222	-.080	-.098	
3				.425		.275	.036	-.217	-.284	-.230	-.099	-.062	-.009
0	.553		.438		.392		.042		-.283		-.091	-.054	.009
$\alpha=12.0^\circ; \beta=0.3^\circ$													
P.L.	.496	.276	.254	.237	.216	.179	.159	-.002	-.043	-.152	-.164	-.106	-.037
57½					.216		.154	-.025	-.043				
45			.254		.199	.178	.057	-.071	-.102	-.152			
30		.276	.276	.256	.198	.192	.057	-.117	-.196	-.170	-.164		
15		.345	.294	.280	.241	.189	.040	-.194	-.270	-.229	-.094	-.106	
3				.367		.229	.000	-.240	-.296	-.230	-.094	-.077	-.037
0	.496		.382		.334		.004		-.257		-.087	-.068	-.011

(b) $M=2.01$

$\alpha=0.4^\circ; \beta=-4^\circ$

P.L.	.725	.466	.390	.350	.305	.237	.174	.100	.089	.033	.020		.020
57½					.305			.079	.089				
45			.390		.302	.283	.186	.066	.007	.033			
30		.466	.448	.418	.360	.330	.232	.085	-.007	.027	.020		
15		.603	.551	.528	.493	.434	.293	.063	-.021	-.039	.003		
3				.635		.555	.304	.038	-.061	-.090	-.103	-.057	.020
0	.725		.612		.598		.279		-.072		-.102	-.036	.014
-3				.541		.426	.186	-.036	-.125	-.148	-.192	-.142	-.050
-15		.347	.263	.216	.185	.129	.032	-.117	-.157	-.150	-.136		
-30		.143	.083	.079	.050	.039	-.016	-.107	-.153	-.158	-.054		
-45			.063		.032	.024	-.035	-.111	-.150	-.105			
-57½					.032			-.088	-.073				
-P.L.	.628	.143	.063	.056	.032	.004	-.023	-.064	-.073	-.105	-.054		-.060
$\alpha=0.4^\circ; \beta=-8^\circ$													
P.L.	.709	.579	.502	.466	.419	.338	.261	.176	.158	.090	.061		-.073
57½					.419			.152	.158				
45			.502		.408	.383	.273	.139	.079	.090			
30		.579	.566	.528	.465	.439	.320	.155	.055	.104	.061		
15		.665	.636	.614	.578	.524	.377	.121	.023	.001	.028		
3				.633		.570	.322	.043	-.058	-.111	-.158	-.123	-.073
0	.709		.576		.572		.264		-.090		-.181	-.112	-.061
-3				.636		.555	.304	.037	-.063	-.090	-.102	-.058	.020
-15		.605	.552	.530	.494	.435	.295	.063	-.021	-.038	.005		
-30		.467	.450	.420	.364	.332	.234	.087	-.006	.029	.022		
-45			.391		.304	.284	.189	.067	.009	.033			
-57½					.309			.080	.091				
-P.L.	.724	.467	.391	.352	.309	.238	.176	.103	.091	.033	.022		.020

TABLE XIII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED ROUND-WINDSHIELD CANOPY - Continued

(b) $M=2.01$

α/β	.002	.036	.088	.124	.160	.212	.260	.311	.356	.440	.600	.828	.988
β, deg													
$\alpha=6.5^\circ; \beta=-4^\circ$													
P.L.	.557	.385	.340	.315	.298	.235	.177	.096	.072	.002	-.033	-.150	-.033
57½					.298			.076	.072				
45			.340		.281	.261	.173	.060	.011	.002			
30		.385	.384	.357	.306	.284	.193	.057	-.027	.014	-.033		
15		.479	.434	.413	.381	.332	.214	.009	-.066	-.082	-.361	-.150	
3			.469		.399	.184	-.043	-.122	-.140	-.125	-.078	-.033	
0	.557		.455		.434		.159	-.129		-.116	-.081	-.056	
-3			.425		.327	.115	-.080	-.146	-.140	-.124	-.111	-.115	
-15		.328	.256	.229	.204	.159	.059	-.096	-.142	-.134	-.054	-.035	
-30		.181	.162	.165	.124	.106	.043	-.059	-.117	-.094	-.071		
-45			.148		.106	.100	.035	-.052	-.089	-.083			
-57½					.119		-.034	-.032					
-P.L.	.517	.181	.148	.128	.119	.086	.045	-.016	-.032	-.083	-.071	-.035	-.115
$\alpha=6.5^\circ; \beta=-8^\circ$													
P.L.	.530	.478	.442	.423	.397	.325	.255	.164	.136	.050	.003	-.148	-.068
57½					.397			.143	.136				
45			.442		.379	.346	.249	.126	.075	.050			
30		.478	.487	.454	.395	.375	.270	.116	.024	.075	.003		
15		.527	.501	.482	.449	.406	.279	.055	-.033	-.050	-.044	-.148	
3			.455		.405	.193	-.042	-.121	-.177	-.226	-.149	-.068	
0	.530		.413		.402		.138	-.154		-.212	-.161	-.086	
-3			.378		.270	.070	-.112	-.180	-.196	-.218	-.160	-.084	
-15		.234	.169	.129	.101	.056	-.023	-.147	-.179	-.163	-.068	-.046	
-30		.087	.052	.065	.043	.028	-.024	-.107	-.154	-.130	-.068		
-45			.061		.035	.036	-.019	-.095	-.125	-.112			
-57½					.049		-.071	-.069					
-P.L.	.460	.087	.061	.054	.049	.032	-.003	-.053	-.069	-.112	-.068	-.046	-.084
$\alpha=-6.0^\circ; \beta=0.3^\circ$													
P.L.	.900	.445	.321	.269	.219	.154	.100	.040	.042	.002	.045	-.002	.077
57½					.219		.085	.007	.042				
45			.321		.231	.214	.117	-.002	-.067	.002			
30		.445	.386	.271	.313	.276	.177	.033	-.048	-.057	.045		
15		.679	.587	.547	.512	.428	.386	.047	-.030	-.032	-.027	-.002	
3			.608		.669	.396	.102	-.010	-.030	-.055	.025	.077	
0	.900		.811		.801		.405		.002	-.043	.030	.087	
$\alpha=-3.0^\circ; \beta=0.3^\circ$													
P.L.	.804	.390	.291	.243	.203	.148	.096	.036	.033	-.009	.020	-.022	.060
57½					.203			.010	.033				
45			.291		.210	.197	.108	.001	-.061	-.009			
30		.390	.344	.335	.278	.244	.155	.025	-.056	-.056	.020		
15		.600	.517	.477	.446	.379	.243	.022	-.048	-.055	-.016	-.022	
3			.710		.610	.337	.064	-.038	-.055	-.067	.010	.060	
0	.804		.723		.698		.351	-.028		-.057	.017	.069	

TABLE XIII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED ROUND-WINDSHIELD CANOPY - Concluded

(b) $M=2.01$

x/λ ϕ, deg	.002	.036	.088	.124	.160	.212	.260	.311	.356	.440	.600	.828	.968
$\alpha=0.4^\circ; \beta=0^\circ$													
P.L.	.715	.357	.281	.245	.207	.156	.105	.044	.033	-.012	-.004		.041
57½					.207			.022	.033				
45			.281		.207	.194	.112	.005	-.047	-.012			
30		.357	.322	.316	.260	.233	.150	.023	-.054	-.041	-.004		
15		.536	.462	.431	.402	.345	.214	.007	-.063	-.071	-.012		
3				.622		.529	.278	.025	-.067	-.082	-.086	-.010	.041
0	.715		.628		.604		.285		-.066		-.079	-.005	.049
$\alpha=3.0^\circ; \beta=0.3^\circ$													
P.L.	.628	.306	.247	.213	.190	.145	.092	.030	.018	-.033	-.030	-.045	.026
57½					.190		.069	.011	.018				
45			.247		.185	.175	.097	-.003	-.048	-.033			
30		.306	.285	.275	.225	.201	.005	.126	-.067	-.046	-.030		
15		.464	.398	.368	.341	.287	.169	-.020	-.081	-.087	-.026	-.045	
3				.536		.450	.216	-.010	-.090	-.092	-.088	-.014	.026
0	.628		.546		.523		.226		-.085		-.081	-.011	.034
$\alpha=6.5^\circ; \beta=0^\circ$													
P.L.	.555	.282	.239	.215	.201	.155	.104	.035	.014	-.046	-.064	-.081	-.009
57½					.201			.016	.014				
45			.239		.186	.175	.098	-.002	-.045	-.046			
30		.282	.273	.258	.212	.192	.116	-.004	-.076	-.047	-.064		
15		.412	.351	.329	.300	.249	.139	-.042	-.105	-.112	-.057	-.081	
3				.457		.373	.154	-.056	-.128	-.119	-.098	-.040	-.009
0	.555		.466		.438		.158		-.124		-.095	-.037	-.001
$\alpha=9.0^\circ; \beta=0.3^\circ$													
P.L.	.491	.243	.211	.196	.187	.145	.098	.030	.006	-.058	-.081	-.082	-.030
57½					.187		.075	.011	.006				
45			.211		.171	.157	.084	-.011	-.049	-.058			
30		.243	.240	.226	.183	.168	-.019	.097	-.086	-.053	-.081		
15		.354	.296	.277	.251	.207	.104	-.064	-.121	-.127	-.070	-.082	
3				.389		.311	.110	-.082	-.144	-.128	-.090	-.046	-.030
0	.491		.400		.377		.117		-.141		-.089	-.046	-.028
$\alpha=12.0^\circ; \beta=0.3^\circ$													
P.L.	.431	.218	.198	.191	.185	.148	.102	.033	.006	-.062	-.097	-.089	-.048
57½					.165		.078	.017	.006				
45			.198		.170	.152	.083	-.008	-.043	-.062			
30		.218	.226	.210	.170	.158	-.025	.088	-.089	-.054	-.097		
15		.308	.260	.241	.215	.177	.082	-.078	-.133	-.137	-.090	-.089	
3				.327		.255	.068	-.109	-.159	-.135	-.084	-.062	-.048
0	.431		.339		.316		.071		-.157		-.085	-.063	-.046

TABLE XIV.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED FLAT-WINDSHIELD CANOPY

(a) $M=1.41$

α/β	$-\alpha$	0.002	0.006	0.035	0.072	0.137	0.181	0.206	0.210	0.228	0.264	0.306	0.365	0.444	0.606	0.862	0.990
$\alpha=0.4^\circ; \beta=0^\circ$																	
$P_{x/L}$.601	.443		.059				.027	.005	-.082	-.119		-.140			.067
$33\frac{1}{2}$.059	.041					-.090	-.119					
$22\frac{1}{2}$.443	.291		.047				-.065		-.241	-.175	-.140			
$13\frac{1}{2}$.481														
12	.704				.211	.122			.003		-.213		-.198	-.141			
9			.594														
$7\frac{1}{2}$.553													
6					.525												
5						.233											
3	.789				.519	.410			-.254	-.186	-.177	-.228	-.261	-.128	-.057	.067	
0			.611					.209		-.205		-.226		-.215			
$\alpha=0.4^\circ; \beta=-4^\circ$																	
$P_{x/L}$.720	.556		.159				.111	.086	-.016	-.058		-.149			.053
$33\frac{1}{2}$.159	.149					-.011	-.058					
$22\frac{1}{2}$.556	.400		.171				.031		-.173	-.138	-.149			
$13\frac{1}{2}$.596														
12	.726				.334	.245			.114		-.158		-.126	-.194			
9			.664														
$7\frac{1}{2}$.614													
6					.512												
5						.293											
3	.796				.530	.431			-.135	-.148	-.230	-.261	-.247	-.221	-.080	.053	
0			.599					.206		-.229	-.270	-.200					
-3	.771				.501	.387			-.356	-.257	-.213	-.261	-.258	-.165	-.121	.041	
-5						.196											
-6						.364											
$-7\frac{1}{2}$.458													
-9			.495														
-12	.655				.083	-.016			-.113		-.269		-.233	-.119			
$-13\frac{1}{2}$.323														
-22		.317	.177		-.076					-.167		-.306	-.163	-.112			
$-33\frac{1}{2}$					-.036	-.063					-.160	-.177					
$-P_{x/L}$.440	.317		-.036					-.048	-.069	-.111	-.177		-.112		.041	
$\alpha=0.4^\circ; \beta=-8^\circ$																	
$P_{x/L}$.790	.652		.265				.199	.170	.053	.011		-.130			.004
$33\frac{1}{2}$.265	.258					.070	.011					
$22\frac{1}{2}$.652	.464		.292				.127		-.103	-.086	-.130			
$13\frac{1}{2}$.683														
12	.719				.434	.360			.208		-.110		-.173	-.238			
9			.706														
$7\frac{1}{2}$.650													
6					.550												
5						.340											
3	.789				.532	.444			-.043	-.117	-.262	-.345	-.240	-.257	-.144	.004	
0			.586					.202		-.242	-.406		-.196				
-3	.766				.478	.367			-.375	-.350	-.343	-.341	-.253	-.272	-.149	.005	
-5						.172											
-6						.295											
$-7\frac{1}{2}$.358													
-9			.427														
-12	.607				-.024	-.132			-.220		-.312		-.303	-.167			
$-13\frac{1}{2}$.029														
-22		.183	.060		-.156					-.271		-.356	-.205	-.116			
$-33\frac{1}{2}$					-.126	-.163					-.224	-.235					
$-P_{x/L}$.246	.183		-.176					-.122	-.141	-.201	-.235		-.116		.005	

TABLE XIV. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED FLAT-WINDSHIELD CANOPY - Continued

(a) $M=1.41$

α, β	$\alpha=0^\circ$	$\alpha=2^\circ$	$\alpha=4^\circ$	$\alpha=6^\circ$	$\alpha=8^\circ$	$\alpha=10^\circ$	$\alpha=12^\circ$	$\alpha=14^\circ$	$\alpha=16^\circ$	$\alpha=18^\circ$	$\alpha=20^\circ$
$\alpha=6.5^\circ; \beta=0^\circ$											
P_{x1}	.497	.388	.081		.033	.011	-.074	-.120	-.184		.104
$33\frac{1}{2}$.081	.060		-.073	-.120				
$22\frac{1}{2}$.388	.260	.053		-.052	-.228	-.211	-.184		
$13\frac{1}{2}$.446									
12	.560		.407	.162	.074		-.026	-.243	-.208	-.154	
9		.506									
$7\frac{1}{2}$.445								
6				.330							
5					.131						
3	.633		.397	.293	.165	-.303	-.249	-.228	-.259	-.224	-.186
0		.496				.112	-.271	-.255	-.222	-.195	
$\alpha=6.5^\circ; \beta=-4^\circ$											
P_{x1}	.594	.468	.169		.103	.082	-.019	-.068	-.214		.056
$33\frac{1}{2}$.169	.148			-.009	-.068			
$22\frac{1}{2}$.468	.360	.160		.029	-.176	-.173	-.214		
$13\frac{1}{2}$.524									
12	.570		.392	.268	.182	.070	-.205		-.249	-.266	
9		.569									
$7\frac{1}{2}$.504								
6				.368							
5					.190						
3	.633		.400	.284	.156	-.357	-.216	-.304	-.317	-.285	-.182
0		.466			.104	-.288	-.306		-.186		-.051
-3	.616		.377	.267	.158	-.401	-.313	-.264	-.284	-.268	-.188
-5				.093							
-6				.249							
$-7\frac{1}{2}$.353								
-9		.417									
-12	.517		.403	.033	-.057	-.138	-.296		-.209	-.122	
$-13\frac{1}{2}$.273									
$-22\frac{1}{2}$.280	.149	-.060		-.143	-.287	-.224	-.144		
$-33\frac{1}{2}$			-.007	-.032			-.140	-.173			
$-P_{x1}$.360	.280	-.007			-.037	-.058	-.129	-.173	-.144	.028
$\alpha=6.5^\circ; \beta=-8^\circ$											
P_{x1}	.642	.573	.263		.182	.160	.047	-.010	-.182		.015
$33\frac{1}{2}$.263	.242			.064	-.010			
$22\frac{1}{2}$.573	.455	.263		.110	-.118	-.127	-.182		
$13\frac{1}{2}$.594									
12	.546		.394	.360	.284	.148	-.168		-.235	-.320	
9		.597									
$7\frac{1}{2}$.531								
6				.413							
5					.234						
3	.624		.394	.297	.159	-.114	-.195	-.339	-.424	-.407	-.178
0		.470			.099	-.311	-.455		-.200		-.049
-3	.600		.351	.259	.155	-.421	-.417	-.420	-.394	-.359	-.185
-5				.069							
-6				.178							
$-7\frac{1}{2}$.252								
-9		.349									
-12	.462		.393	-.074	-.178	-.247	-.336		-.248	-.131	
$-13\frac{1}{2}$.056									
$-22\frac{1}{2}$.171	.043	-.169		-.227	-.331	-.216	-.110		
$-33\frac{1}{2}$			-.087	-.118			-.197	-.219			
$-P_{x1}$.194	.171	-.087			-.099	-.119	-.179	-.219	-.110	-.002

TABLE XIV.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED FLAT-WINDSHIELD CANOPY - Continued

(b) M=2.01																	
α, \deg	β, \deg	-0.02	.006	.035	.072	.137	.181	.206	.210	.228	.264	.306	.365	.444	.606	.862	.990
$\alpha=0.4^\circ; \beta=0^\circ$																	
P.L.		.532	.401		.090				.058	.043	.001	-.023		-.083			.054
33½					.090	.065					-.015	-.023					
22½			.401	.315		.087				.011		-.110	-.085	-.083			
13½			.460														
12				.425	.255	.182			.105		-.056		-.118	-.077			
9			.622														
7½				.544													
6					.488												
5						.362											
3		.792			.569	.480	.385		-.074	-.060	-.083	-.092	-.112	-.087	-.037	.054	
0			.626					.363		-.090		-.089		-.091		.044	
$\alpha=0.4^\circ; \beta=-4^\circ$																	
P.L.		.619	.497		.166				.123	.106	.053	.029		-.068			.019
33½					.166	.116					.047	.029					
22½			.497	.411		.180				.088		-.059	-.045	-.068			
13½			.569														
12				.527	.342	.275			.185		-.012		-.101	-.103			
9			.686														
7½				.623													
6					.538												
5						.380											
3		.783			.573	.490	.388		-.008	-.008	-.095	-.131	-.136	-.132	-.077	.019	
0			.630					.350		-.103		-.132		-.132		-.001	
-3		.778			.551	.463	.377		-.096	-.108	-.094	-.112	-.141	-.134	-.109	-.009	
-5						.336											
-6					.425												
-7½			.664														
-9			.556														
-12				.237	.167	.096			.023		-.103		-.142	-.080			
-13½			.315														
-22½		.288	.213		.001				-.067		-.164	-.124	-.077				
-33½					.011	-.018				.077	-.072						
-P.L.		.398	.288		.011				-.007	-.021	-.052	-.072		-.077		-.009	
$\alpha=0.4^\circ; \beta=-8^\circ$																	
P.L.		.671	.585		.263				.206	.188	.126	.096		-.026			-.046
33½					.263	.247					.125	.096					
22½			.585	.510		.288				.184		.007	.012	-.026			
13½			.654														
12				.623	.447	.379			.277		.038		-.059	-.093			
9			.732														
7½				.687													
6					.587												
5						.427											
3		.760			.586	.508	.406		.066	.030	-.073	-.140	-.188	-.222	-.145	-.046	
0			.644					.349		-.069		-.169		-.214		-.066	
-3		.751			.546	.450	.367		-.101	-.163	-.167	-.187	-.202	-.236	-.149	-.062	
-5						.216											
-6					.378												
-7½				.432													
-9			.521														
-12				-.036	.020	.027			-.028		-.139		-.176	-.133			
-13½			.036														
-22½		.175	.111		-.081				-.138		-.217	-.158	-.097				
-33½					-.063	-.103				-.133	-.122						
-P.L.		.222	.175		-.063				-.068	-.079	-.104	-.122		-.097		-.062	

TABLE XIV. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED FLAT-WINDSHIELD CANOPY - Concluded

(b) M=2.01																
x/l	.002	.006	.035	.072	.137	.181	.206	.210	.228	.264	.306	.365	.444	.606	.862	.990
θ, deg	$\alpha = 6.5^\circ; \beta = 0^\circ$															
P.L.		.431	.332		.089				.053	.038	-.008	-.028		-.115		.044
33 $\frac{1}{2}$.089	.068					-.013	-.028				
22 $\frac{1}{2}$.332	.266		.073			.007		-.104	-.093	-.115			
13 $\frac{1}{2}$.378													
12				.336	.190	.126			.058	-.089		-.142	-.108			
9			.504													
7 $\frac{1}{2}$.429												
6					.361											
5						.244										
3		.605			.422	.343	.259		-.118	-.111	-.130	-.130	-.133	-.077	-.042	.044
0			.490					.245		-.132		-.125		-.078		-.021
$\alpha = 6.5^\circ; \beta = -4^\circ$																
P.L.		.500	.411		.154				.110	.094	.037	.014		-.090		-.004
33 $\frac{1}{2}$.154	.140					.043	.014				
22 $\frac{1}{2}$.411	.349		.157				.075		-.057	-.060	-.090		
13 $\frac{1}{2}$.466													
12				.426	.268	.210			.130	-.053		-.127	-.152			
9		.551														
7 $\frac{1}{2}$.494												
6					.405											
5						.268										
3		.598			.426	.353	.265		-.058	-.067	-.141	-.178	-.173	-.123	-.096	-.004
0			.488				.234		-.148		-.170			-.114		-.006
-3		.588		.406	.325	.252			-.144	-.154	-.146	-.152	-.163	-.123	-.144	-.026
-5					.220											
-6					.291											
-7 $\frac{1}{2}$.347												
-9			.439													
-12				.166	.099	.039			-.023		-.135		-.149	-.083		
-13 $\frac{1}{2}$.249													
-22 $\frac{1}{2}$.236	.173		-.013				-.065		-.150	-.131	-.116		
-33 $\frac{1}{2}$.017	-.008					-.066	-.073				
-P.L.		.313	.236		.017				-.005	-.019	-.055	-.073		-.116		-.026
$\alpha = 6.5^\circ; \beta = -8^\circ$																
P.L.		.522	.480		.225				.171	.160	.095	.069		-.059		-.019
33 $\frac{1}{2}$.225	.218					.108	.069				
22 $\frac{1}{2}$.480	.422		.244				.153		-.007	-.017	-.059		
13 $\frac{1}{2}$.517													
12				.496	.353	.294			.203	-.019		-.126	-.138			
9		.565														
7 $\frac{1}{2}$.535												
6					.439											
5						.300										
3		.560			.431	.363	.276		-.002	-.045	-.131	-.192	-.235	-.207	-.118	-.019
0			.493				.226			-.129		-.231		-.207		-.027
-3		.557		.400	.315	.244			-.147	-.203	-.214	-.217	-.231	-.231	-.144	-.032
-5					.204											
-6					.255											
-7 $\frac{1}{2}$.315												
-9			.406													
-12				-.065	-.038	-.035			-.073		-.169		-.176	-.100		
-13 $\frac{1}{2}$.009													
-22 $\frac{1}{2}$.136	.081		-.086				-.127		-.187	-.156	-.097		
-33 $\frac{1}{2}$					-.044	-.073					-.105	-.109				
-P.L.		.160	.136		-.044				-.051	-.062	-.109		-.097		-.032	

TABLE XV.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED VEE-WINDSHIELD CANOPY

(a) $M=1.41$

x/l ϕ, deg	.010	.030	.083	.133	.176	.224	.251	.260	.269	.293	.300	.360	.440	.600	.856	.992
$\alpha=0.4^\circ; \beta=0^\circ$																
P.L.	.426	.347	.352	.318	.304	.092		-.014				-.045	-.114		-.015	.090
33 $\frac{1}{2}$								-.078					-.114			
30					.304											
29						.024										
27 $\frac{1}{2}$.212										
24									-.085							
22 $\frac{1}{2}$.318				.186	-.076		-.293	-.330	-.246	-.015		
12			.352	.393	.421	.424		.391		.209	-.064	-.425	-.266	-.176		
4 $\frac{1}{2}$.295								
3		.347	.419		.507		.282	.078				-.061	-.184	-.293		.090
1 $\frac{1}{2}$.465										
1		.373	.437	.478												
0								.130				-.033	-.176			.077
$\alpha=0.4^\circ; \beta=-4^\circ$																
P.L.	.535	.455	.466	.424	.402	.185		.062				.026	-.051		-.014	.062
33 $\frac{1}{2}$.015					-.051			
30					.402											
29						.140										
27 $\frac{1}{2}$.296										
24									-.128							
22 $\frac{1}{2}$.424				.230	-.039		-.184	-.236	-.198	-.014		
12			.466	.492	.500	.479		.434		.234	-.045	-.425	-.211	-.186		
4 $\frac{1}{2}$.315								
3		.455	.511		.573		.259	-.100				-.134	-.216	-.269		.062
1 $\frac{1}{2}$.480										
1		.476	.510	.547												
0								.094				-.047	-.203			.083
-1		.234	.287	.327												
-1 $\frac{1}{2}$.371										
-3		.221	.306		.391		.271	.142				-.053	-.195	-.310		.046
-4 $\frac{1}{2}$.243								
-12			.226	.275	.313	.335		.323		.162	-.095	-.430	-.359	-.148		
-22 $\frac{1}{2}$.204				.142	-.113		-.353	-.426	-.276	.009		
-24									-.126							
-27 $\frac{1}{2}$.139										
-29							-.098									
-30					.191											
-33 $\frac{1}{2}$								-.187				-.176				
-P.L.	.273	.221	.226	.204	.191	-.009		-.097				-.120	-.176	.009		.046

TABLE XV.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD LOCATED VEE WINDSHIELD CANOPY. Continued

(a) $M=1.41$

x/l θ, deg	.010	.030	.083	.133	.176	.224	.251	.260	.269	.293	.300	.360	.440	.600	.856	.992
$\alpha=0.4^\circ; \beta=-8^\circ$																
P.L.	.629	.570	.589	.525	.500	.287		.153			.109	.022		-.004		.006
33½								.116				.022				
30					.500											
29						.264										
27½						.392										
24								-.097								
22½				.525				.290	.033		-.078	-.138	-.134	-.004		
12			.589	.597	.585	.538		.480		.268	-.020	-.382	-.249	-.189		
4½								.338								
3		.570	.583		.617		.245	-.273			-.360	-.264	-.258			.006
1½						.445										
1		.558	.548	.560												
0								-.019			-.113	-.243				.085
-1		-.070	-.078	-.051												
-1½						.086										
-3		.092	.109		-.001		.057	.014			-.109	-.246	-.317			-.002
-4½								-.034								
-12			.101	.155	.196	.228		.241		.127	-.118	-.403	-.394	-.202		
-22½				.078				.087	-.154		-.383	-.494	-.371	.039		
-24								-.133								
-27½						.060										
-29						-.217										
-30					.065											
-33½								-.299				-.235				
-P.L.	.079	.092	.101	.078	.065	-.110		-.174			-.192	-.235		.039		-.002
$\alpha=6.5^\circ; \beta=0^\circ$																
P.L.	.321	.268	.288	.285	.308	.118		.016			-.025	-.105		-.221		.102
33½								-.037				-.105				
30					.308											
29						.084										
27½						.208										
24								-.121								
22½				.285				.180	-.101		-.238	-.249	-.253	-.221		
12			.288	.335	.350	.341		.321		.159	-.087	-.464	-.254	-.199		
4½								.198								
3		.268	.327		.392		.193	-.041			-.139	-.251	-.335			.102
1½						.363										
1		.288	.340	.381												
0								.040			-.110	-.237		-.117		.139

TABLE IV. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED RE-WINDSHIELD CANOPY - Continued

(a) $M=1.41$

x/l	.010	.030	.083	.133	.176	.224	.251	.260	.269	.293	.300	.360	.440	.600	.856	.992
θ, deg	$\alpha=6.5^\circ; \beta=-4^\circ$															
P.L.	.426	.377	.396	.369	.392	.196		.092				.044	-.046		-.269	.068
33 $\frac{1}{2}$.053					-.046			
30					.392											
29						.192										
27 $\frac{1}{2}$.295										
24								-.055								
22 $\frac{1}{2}$.369				.247	-.037			-.142	-.167	-.200	-.269	
19		.396	.434	.431	.398			.360		.185	-.071	-.438	-.286	-.247		
14 $\frac{1}{2}$.212								
9	.377	.416		.466		.180	-.235					-.196	-.275	-.285		.068
1 $\frac{1}{2}$.362											
1	.388	.407	.436													
0								.017				-.138	-.268		-.174	.068
-1	.141	.200	.187													
-1 $\frac{1}{2}$.222											
-3	.144	.221		.297		.187	.025					-.128	-.254	-.351		.033
-4 $\frac{1}{2}$.151									
-12		.176	.227	.258	.267		.262		.122	-.118	-.470	-.331	-.140			
-22 $\frac{1}{2}$.183				.147	-.149		-.343	-.345	-.294	-.202			
-24							-.144									
-27 $\frac{1}{2}$.126											
-29					-.032											
-30				.215												
-33 $\frac{1}{2}$							-.125					-.160				
-P.L.	.174	.144	.176	.183	.216	.032	-.061				-.095	-.160		-.202		.033
θ, deg	$\alpha=6.5^\circ; \beta=-8^\circ$															
P.L.	.517	.484	.500	.431	.461	.259		.166				.123	.019		-.232	.029
33 $\frac{1}{2}$.153					.019			
30					.461											
29						.307										
27 $\frac{1}{2}$.390										
24								.023								
22 $\frac{1}{2}$.431				.304	.055		-.047	-.084	-.140	-.232		
19		.500	.530	.501	.450			.404		.213	-.048	-.371	-.324	-.280		
14 $\frac{1}{2}$.238								
9	.484	.485		.504		.191	-.346					-.409	-.428	-.317		.029
1 $\frac{1}{2}$.328											
1	.467	.438	.438													
0								-.130				-.198	-.309		-.299	.034
-1	-.171	-.149	-.112													
-1 $\frac{1}{2}$.015											
-3	.019	-.014		-.103		-.025	-.102					-.193	-.318	-.344		.025
-4 $\frac{1}{2}$							-.154									
-12		.061	.116	.158	.182		.202		.097	-.125	-.473	-.414	-.158			
-22 $\frac{1}{2}$.088				.107	-.185		-.390	-.435	-.322	-.176			
-24							-.147									
-27 $\frac{1}{2}$.048											
-29					-.135											
-30				.119												
-33 $\frac{1}{2}$							-.213					-.211				
-P.L.	-.009	.019	.061	.088	.119	-.051	-.134				-.168	-.211		-.176		.025

TABLE XV.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED VEE-WINDSHIELD CANOPY - Continued

(b) $M=2.01$

x/l θ, deg	.010	.030	.083	.133	.176	.224	.251	.260	.269	.293	.300	.360	.440	.600	.856	.992
$\alpha=0.4^\circ; \beta=0^\circ$																
P.L.	.388	.329	.320	.286	.297	.103		.044			.017	-.018		-.104		.077
33 $\frac{1}{2}$								-.008				-.018				
30					.297											
29						.079										
27 $\frac{1}{2}$.287										
24								.073								
22 $\frac{1}{2}$.286				.289	.105		-.068	-.143	-.113	-.104		
12			.320	.354	.379	.396		.433		.375	.183	-.172	-.168	-.060		
4 $\frac{1}{2}$.362								
3		.329	.406		.464		.346	.129			.030	-.007	-.078	-.120		.077
1 $\frac{1}{2}$.473										
1		.370	.443	.477												
0								.181				.006		-.126		.094
$\alpha=0.4^\circ; \beta=-4^\circ$																
P.L.	.520	.442	.432	.385	.404	.201		.133			.104	.055		-.085		.036
33 $\frac{1}{2}$.091				.055				
30					.404											
29						.189										
27 $\frac{1}{2}$.381										
24								.128								
22 $\frac{1}{2}$.385				.362	.159		-.021	-.069	-.034	-.085		
12			.432	.467	.485	.493		.519		.442	.237	-.150	-.141	-.059		
4 $\frac{1}{2}$.422								
3		.442	.508		.557		.394	.015			-.071	-.016	-.080	-.091		.036
1 $\frac{1}{2}$.513										
1		.473	.521	.550												
0								.152				-.014		-.116		.030
-1		.248	.317	.331												
-1 $\frac{1}{2}$.363										
-3		.228	.311		.384		.318	.191			.060	-.009	-.085	-.125		-.027
-4 $\frac{1}{2}$.286								
-12			.220	.257	.287	.306		.340		.295	.127	-.165	-.150	-.105		
-22 $\frac{1}{2}$.197				.220	.056		-.101	-.163	-.170	-.090		
-24								.023								
-27 $\frac{1}{2}$.201										
-29						-.006										
-30					.194											
-33 $\frac{1}{2}$								-.088				-.084				
-P.L.	.259	.228	.220	.197	.194	.022		-.026			-.050	-.084		-.090		-.027

TABLE X. PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED RE-WINDSHIELD CANOPY - Continued

(b) $M=2.01$

α/γ θ, deg	.010	.030	.083	.133	.176	.224	.251	.260	.269	.293	.300	.360	.440	.600	.856	.992
$\alpha=0.4^\circ; \beta=-8^\circ$																
P.L.	.620	.562	.544	.450	.492	.285		.217			.187	.123		-.041		-.035
33 $\frac{1}{2}$.169				.123				
30					.492											
29						.305										
27 $\frac{1}{2}$.471											
24								.174								
22 $\frac{1}{2}$.450				.423	.206		.049	.008	.027	-.041		
12			.544	.581	.590	.587		.603		.505	.280	-.119	-.124	-.086		
4 $\frac{1}{2}$.475								
3		.562	.599		.640		.452	-.012			-.109	-.141	-.143	-.143		-.035
1 $\frac{1}{2}$.532										
1		.552	.566	.584												
0								.045				-.104		-.142		-.009
-1		.091	.031	.050												
-1 $\frac{1}{2}$.143										
-3		.125	.227		.232		.152	.108			.030	-.068	-.140	-.144		-.034
-4 $\frac{1}{2}$.197								
-12			.126	.165	.195	.213		.245		.214	.069	-.172	-.148	-.149		
-22 $\frac{1}{2}$.101				.148	.009		-.127	-.202	-.196	-.136		
-24								-.023								
-27 $\frac{1}{2}$.122										
-29						-.072										
-30					.095											
-33 $\frac{1}{2}$								-.149				-.179				
-P.L.	.124	.125	.126	.101	.095	-.056		-.095			-.111	-.179		-.136		-.034
$\alpha=6.5^\circ; \beta=0^\circ$																
P.L.	.287	.252	.254	.239	.254	.113		.054			.030	-.016		-.123		.029
33 $\frac{1}{2}$.022				-.016				
30					.254											
29						.095										
27 $\frac{1}{2}$.251										
24								.033								
22 $\frac{1}{2}$.239				.230	.067		-.086	-.122	-.087	-.123		
12			.254	.299	.316	.324		.347		.298	.130	-.187	-.175	-.094		
4 $\frac{1}{2}$.266								
3		.252	.321		.361		.253	.029			-.049	-.054	-.109	-.091		.029
1 $\frac{1}{2}$.347										
1		.275	.332	.357												
0								.101				-.038		-.099		.041

CONFIDENTIAL

47

TABLE XV.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED VEE-WINDSHIELD CANOPY - Concluded

(b) $M=2.01$

α/β β, deg	.010	.030	.083	.133	.176	.224	.251	.260	.269	.293	.300	.360	.440	.600	.856	.992
$\alpha=6.5^\circ; \beta=-4^\circ$																
P.L.	.404	.334	.346	.304	.338	.179		.124			.103	.043		-.089		.037
33 $\frac{1}{2}$.113				.043				
30					.338											
29						.211										
27 $\frac{1}{2}$.346										
24								.092								
22 $\frac{1}{2}$.304				.291	.109		-.030	-.056	-.037	-.089		
12		.346	.398	.406	.406			.420		.353	.173	-.170	-.170	-.142		
4 $\frac{1}{2}$.322								
3		.354	.407	.444		.294	-.065				-.140	-.188	-.181	-.116		.037
1 $\frac{1}{2}$.384											
1		.378	.407	.422												
0								.076					-.066	-.128		.029
-1		.158	.195	.203												
-1 $\frac{1}{2}$.257										
-3		.150	.225		.279		.219	.096			-.003	-.055	-.116	-.122		-.023
-4 $\frac{1}{2}$.203								
-12			.167	.203	.229	.242		.270		.235	.083	-.192	-.196	-.079		
-22 $\frac{1}{2}$.165				.177	.031		-.115	-.172	-.135	-.131		
-24								.003								
-27 $\frac{1}{2}$.186										
-29						.016										
-30					.181											
-33 $\frac{1}{2}$								-.053				-.059				
-P.L.	.165	.150	.167	.165	.181	.049		-.001			-.023	-.059		-.131		-.023
$\alpha=6.5^\circ; \beta=-8^\circ$																
P.L.	.493	.454	.423	.330	.379	.230		.177			.167	.102		-.050		-.031
33 $\frac{1}{2}$.203				.102				
30					.379											
29						.304										
27 $\frac{1}{2}$.436										
24								.154								
22 $\frac{1}{2}$.330				.248	.159		.053	.012	.013	-.050		
12		.423	.504	.497	.484			.491		.402	.209	-.142	-.133	-.131		
4 $\frac{1}{2}$.366								
3		.454	.482	.504		.340	-.067				-.135	-.173	-.169	-.177		-.031
1 $\frac{1}{2}$.389											
1		.450	.439	.442												
0								-.045					-.195	-.158		-.015
-1		-.023	-.064	-.050												
-1 $\frac{1}{2}$.040											
-3		.046	.089		.064	.029	-.016			-.071	-.163	-.207	-.151		-.024	
-4 $\frac{1}{2}$.046									
-12			.073	.112	.139	.159		.186		.161	.031	-.195	-.155	-.110		
-22 $\frac{1}{2}$.082				.119	-.009		-.140	-.197	-.199	-.117		
-24								-.041								
-27 $\frac{1}{2}$.111										
-29						-.053										
-30				.100												
-33 $\frac{1}{2}$								-.113				-.101				
-P.L.	.022	.046	.073	.082	.100	-.015		-.056			-.078	-.101		-.117		-.024

CONFIDENTIAL

TABLE X-1. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED ROUND-WINDSHIELD CANOPY

(a) $M=1.41$

x/y ϕ , deg	.010	.030	.068	.096	.133	.176	.224	.260	.302	.360	.440	.600	.856	.992
$\alpha=0.4^\circ; \beta=0^\circ$														
P.L.	.587	.491	.372	.308	.237	.124	-.017	-.061	-.095	-.157	-.152	-.156	-.069	.073
33 $\frac{1}{2}$.124	-.026		-.139	-.157				
22 $\frac{1}{2}$.308		.163	.007		-.199	-.238	-.164	-.156		
12		.491		.442	.350	.235	.036	-.072	-.188	-.292	-.235	-.148	-.069	
3	.687	.648	.594	.567	.545		.060		-.198		-.253	-.187	-.060	.073
0		.647		.589			.100	-.009		-.262		-.154		.081
$\alpha=0.4^\circ; \beta=-4^\circ$														
P.L.	.738	.613	.501	.442	.357	.233	.075	.015	-.024	-.097	-.108	-.158	-.144	.059
33 $\frac{1}{2}$.233	.057		-.069	-.097				
22 $\frac{1}{2}$.442		.267	.097		-.119	-.165	-.118	-.158		
12		.613		.563	.459	.334	.133	.006	-.134	-.246	-.203	-.162	-.144	
3	.738	.685	.612	.582	.559		.103		-.193		-.263	-.287	-.086	.059
0		.617		.561			.089	-.005		-.290		-.220		.058
-3	.601	.577	.529	.514	.490		-.006		-.229		-.294	-.212	-.129	.044
-12		.346		.297	.218	.118	-.074	-.358	-.254	-.332	-.278	-.125	-.042	
-22 $\frac{1}{2}$.176		.054	-.083		-.274	-.304	-.200	-.133		
-33 $\frac{1}{2}$.016	-.118		-.206	-.212				
-P.L.	.601	.346	.232	.176	.114	.016	-.059	-.133	-.158	-.212	-.193	-.133	-.042	.044
$\alpha=0.4^\circ; \beta=-8^\circ$														
P.L.	.746	.718	.625	.578	.467	.351	.176	.100	.055	-.028	-.053	-.134	-.269	.009
33 $\frac{1}{2}$.351	.162		.011	-.028				
22 $\frac{1}{2}$.578		.367	.193		-.030	-.099	-.065	-.134		
12		.718		.657	.551	.417	.215	.077	-.081	-.200	-.173	-.203	-.269	
3	.746	.688	.583	.550	.528		.109		-.216		-.332	-.457	-.179	.009
0		.544		.482			.035	-.039		-.345		-.448		.011
-3	.472	.463	.415	.412	.388		-.096		-.299		-.414	-.370	-.160	.008
-12		.180		.127	.061	-.029	-.195	-.262	-.336	-.390	-.341	-.167	-.087	
-22 $\frac{1}{2}$.039		-.058	-.179		-.348	-.374	-.234	-.130		
-33 $\frac{1}{2}$						-.087	-.200		-.274	-.267				
-P.L.	.472	.180	.086	.039	-.001	-.087	-.174	-.202	-.221	-.267	-.229	-.130	-.087	.008
$\alpha=6.5^\circ; \beta=-4^\circ$														
P.L.	.583	.502	.426	.406	.327	.236	.103	.037	-.003	-.097	-.128	-.210	-.126	.058
33 $\frac{1}{2}$.236	.120		-.048	-.097				
22 $\frac{1}{2}$.406		.253	.094		-.115	-.170	-.153	-.210		
12		.502		.497	.397	.278	.060	-.046	-.182	-.273	-.237	-.253	-.126	
3	.583	.592	.498	.464	.452		.012		-.267		-.289	-.196		.058
0		.541		.446			-.013	-.101		-.334		-.177		.058
-3	.472	.491	.425	.410	.384		-.088		-.284		-.261	-.172		.043
-12		.275		.260	.183	.089	-.092	-.181	-.270	-.344	-.237	-.121	-.060	
-22 $\frac{1}{2}$.166		.065	-.059		-.251	-.276	-.209	-.160		
-33 $\frac{1}{2}$.049	-.065		-.176	-.199				
-P.L.	.472	.275	.198	.166	.136	.049	-.055	-.101	-.128	-.199	-.205	-.160	-.060	.043

TABLE XVI. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REWARD-LOCATED ROUND-WINDSHIELD CANOPY - Continued

(a) $M=1.41$

x/λ θ, deg	.010	.030	.068	.096	.133	.176	.224	.260	.302	.360	.440	.600	.856	.992
$\alpha=6.5^\circ; \beta=-8^\circ$														
P.L.	.589	.584	.532	.534	.422	.331	.193	.114	.069	-.029	-.072	-.182	-.234	.007
$33\frac{1}{2}$.331	.221		.026	-.029				
$22\frac{1}{2}$.534		.344	.178		-.050	-.108	-.106	-.182		
12		.584		.575	.467	.344	.130	.012	-.142	-.248	-.224	-.274	-.234	
3	.589	.591	.466	.422	.418		.006		-.307		-.414	-.342		.007
0		.467		.361			-.060	-.144		-.405		-.335		.015
-3	.362	.350	.313	.306	.276		-.181		-.362		-.390	-.294		.011
-12		.122		.104	.038	-.042	-.204	-.275	-.338	-.386	-.281	-.115	-.106	
-22 $\frac{1}{2}$.046		-.036	-.142		-.320	-.326	-.231	-.128		
-33 $\frac{1}{2}$						-.042	-.146		-.233	-.245				
-P.L.	.362	.122	.075	.046	.033	-.042	-.128	-.160	-.184	-.245	-.228	-.128	-.106	.011
$\alpha=-6.0^\circ; \beta=0.3^\circ$														
P.L.	.827	.557	.395	.311	.226	.095	-.045	-.089	-.106	-.148	-.118	-.096	-.038	.101
$33\frac{1}{2}$.095	-.030		-.166	-.148				
$22\frac{1}{2}$.311		.163	.004		-.211	-.253	-.133	-.096		
12		.557		.493	.399	.275	.063	-.035	-.156	-.255	-.216	-.111	-.038	
3	.827	.754	.715	.696	.681		.158		-.120		-.203	-.148		.101
0		.768		.728			.215	.067		-.195		-.150		.106
$\alpha=-3.0^\circ; \beta=0.3^\circ$														
P.L.	.760	.516	.373	.294	.220	.101	-.037	-.077	-.103	-.157	-.134	-.123	-.050	.090
$33\frac{1}{2}$.101	-.023		-.156	-.157				
$22\frac{1}{2}$.294		.156	.001		-.212	-.247	-.147	-.123		
12		.516		.460	.365	.249	.041	-.060	-.176	-.274	-.221	-.123	-.050	
3	.760	.702	.653	.631	.617		.105		-.158		-.222	-.144		.090
0		.711		.664			.160	.013		-.228		-.145		.094
$\alpha=3.0^\circ; \beta=0.3^\circ$														
P.L.	.613	.431	.326	.271	.220	.124	-.011	-.063	-.093	-.161	-.164	-.176	-.073	.080
$33\frac{1}{2}$.124	-.014		-.138	-.161				
$22\frac{1}{2}$.271		.142	-.004		-.203	-.237	-.173	-.176		
12		.431		.396	.305	.195	.004	-.104	-.213	-.303	-.225	-.148	-.073	
3	.613	.603	.534	.505	.492		.010		-.227		-.248	-.138		.080
0		.607		.538			.057	-.083		-.283		-.139		.095
$\alpha=6.0^\circ; \beta=0.3^\circ$														
P.L.	.543	.387	.299	.266	.214	.134	.004	-.049	-.081	-.155	-.173	-.202	-.079	.104
$33\frac{1}{2}$.134	.010		-.125	-.155				
$22\frac{1}{2}$.266		.141	-.005		-.193	-.232	-.186	-.202		
12		.387		.379	.281	.175	-.011	-.120	-.228	-.312	-.231	-.159	-.079	
3	.543	.557	.482	.455	.439		-.029		-.253		-.245	-.134		.104
0		.560		.482			.012	-.123		-.299		-.135		.137

TABLE XVI. PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED ROUND-WINDSHIELD CANOPY - Continued

(a) $M=1.41$

x/l ϕ, deg	.010	.030	.068	.096	.133	.176	.224	.260	.302	.360	.440	.600	.856	.992
$\alpha=9.0^\circ; \beta=0.3^\circ$														
P.L.	.474	.344	.275	.260	.205	.139	.021	-.030	-.064	-.146	-.173	-.227	-.084	.140
33 $\frac{1}{2}$.139	.040		-.110	-.146				
22 $\frac{1}{2}$.260		.145	-.001		-.180	-.225	-.195	-.227		
12		.344		.353	.264	.160	-.022	-.130	-.238	-.315	-.240	-.164	-.084	
3	.474	.511	.437	.440	.393		-.061		-.272		-.234	-.129		.140
0		.516		.435			-.025	-.154		-.309		-.131		.173
$\alpha=12.0^\circ; \beta=0.3^\circ$														
P.L.	.405	.301	.248	.245	.193	.135	.031	-.021	-.057	-.138	-.175	-.256	-.097	.164
33 $\frac{1}{2}$.135	.059		-.099	-.138				
22 $\frac{1}{2}$.245		.146	-.002		-.175	-.222	-.207	-.256		
12		.301		.334	.245	.142	-.034	-.142	-.251	-.324	-.258	-.175	-.097	
3	.405	.457	.398	.365	.346		-.094		-.294		-.227	-.131		.164
0		.466		.389			-.064	-.186		-.317		-.134		.179

(b) $M=2.01$

$\alpha=0.4^\circ; \beta=-4^\circ$														
P.L.	.696	.535	.436	.400	.330	.241	.129	.080	.048	-.003	-.017	-.078	-.128	.018
33 $\frac{1}{2}$.241	.116		.010	-.003				
22 $\frac{1}{2}$.400		.291	.171		-.007	-.065	-.038	-.078		
12		.535		.555	.464	.383	.213	.109	-.010	-.102	-.099	-.091	-.128	
3	.696	.694	.619	.604	.605		.200		-.045		-.123	-.139	-.082	.018
0		.660		.586			.190	.082		-.124		-.137		.017
-3	.560	.580	.530	.535	.527		.122		-.082		-.130	-.138	-.120	-.013
-12		.290		.304	.245	.180	.044	-.030	-.108	-.162	-.147	-.095	-.025	
-22 $\frac{1}{2}$.156		.094	.009		-.120	-.170	-.124	-.095		
-33 $\frac{1}{2}$.045	-.042		-.109	-.106				
-P.L.	.560	.290	.197	.156	.122	.045	-.031	-.058	-.077	-.106	-.108	-.095	-.025	-.013
$\alpha=0.4^\circ; \beta=-8^\circ$														
P.L.	.716	.635	.551	.530	.436	.351	.223	.165	.123	.057	.039	-.050	-.129	-.059
33 $\frac{1}{2}$.351	.211		.084	.057				
22 $\frac{1}{2}$.530		.393	.258		.057	-.004	.014	-.050		
12		.635		.660	.561	.470	.290	.170	.036	-.071	-.068	-.084	-.129	
3	.716	.703	.620	.595	.606		.214		-.051		-.154	-.206	-.156	-.059
0		.614		.537			.159	.064		-.153		-.227		-.049
-3	.456	.487	.452	.466	.457		.072		-.119		-.185	-.226	-.166	-.061
-12		.169		.174	.133	.078	-.039	-.099	-.157	-.198	-.177	-.145	-.079	
-22 $\frac{1}{2}$.043		.002	-.067		-.167	-.206	-.160	-.114		
-33 $\frac{1}{2}$						-.038	-.109		-.153	-.146				
-P.L.	.456	.169	.086	.043	.030	-.038	-.091	-.110	-.124	-.146	-.144	-.114	-.079	-.061

TABLE XVI.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED ROUND-WINDSHIELD CANOPY. Continued

(b) $M=2.01$

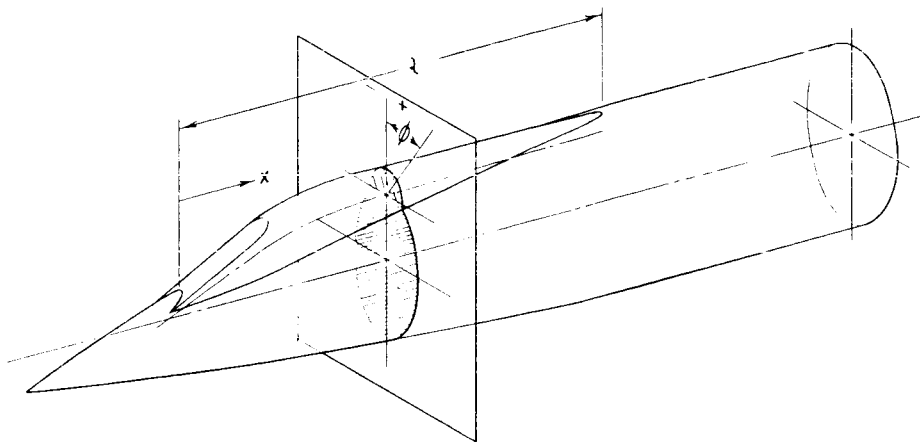
x/l β, deg	.010	.030	.068	.096	.133	.176	.224	.260	.302	.360	.440	.600	.856	.992
$\alpha=6.5^\circ; \beta=-4^\circ$														
P.L.	.519	.410	.338	.323	.259	.200	.109	.077	.043	-.018	-.039	-.113	-.139	.001
33 $\frac{1}{2}$.200	.116		.007	-.018				
22 $\frac{1}{2}$.323		.246	.136		-.034	-.078		-.067	-.113	
12		.410		.456	.366	.291	.144	.049	-.063	-.147	-.139	-.144	-.139	
3	.519	.548	.474	.451	.449		.097		-.110		-.170	-.152	-.103	.001
0		.525		.438			.087	-.002		-.170		-.137		.012
-3	.405	.449	.398	.394	.383		.033		-.134		-.154	-.133	-.149	-.002
-12		.205		.237	.175	.118	-.001	-.064	-.133	-.182	-.161	-.097	-.054	
-22 $\frac{1}{2}$.125		.074	-.009		-.126	-.164	-.130	-.130		
-33 $\frac{1}{2}$.050	-.030		-.097	-.106				
-P.L.	.405	.205	.144	.125	.102	.050	-.018	-.046	-.066	-.106	-.110	-.130	-.054	-.002
$\alpha=6.5^\circ; \beta=-8^\circ$														
P.L.	.532	.486	.418	.423	.331	.271	.177	.150	.115	.040	.012	-.082	-.153	-.021
33 $\frac{1}{2}$.271	.208		.074	.040				
22 $\frac{1}{2}$.423		.342	.215		.027	-.023	-.018	-.082		
12		.486		.542	.444	.363	.202	.097	-.026	-.119	-.117	-.134	-.153	
3	.532	.553	.466	.437	.446		.103		-.120		-.207	-.226	-.133	-.021
0		.470		.383			.053	-.023		-.202		-.223		-.026
-3	.299	.344	.311	.318	.306		-.020		-.178		-.225	-.247	-.157	-.049
-12		.086		.103	.065	.018	-.081	-.134	-.183	-.212	-.185	-.106	-.100	
-22 $\frac{1}{2}$.029		-.008	-.073		-.172	-.200	-.155	-.117		
-33 $\frac{1}{2}$						-.022	-.091		-.137	-.141				
-P.L.	.299	.086	.044	.029	.025	-.022	-.075	-.094	-.111	-.141	-.140	-.117	-.100	-.049
$\alpha=-6.0^\circ; \beta=0.3^\circ$														
P.L.	.818	.532	.395	.319	.260	.146	.051	.018	-.004	-.039	-.038	-.043	-.021	.097
33 $\frac{1}{2}$.146	.037		-.050	-.039				
22 $\frac{1}{2}$.319		.233	.124		-.053	-.112	-.070	-.043		
12		.532		.533	.450	.372	.201	.104	-.005	-.093	-.087	-.056	-.021	
3	.818	.799	.747	.749	.748		.287		.009		-.056	-.088	-.018	.097
0		.834		.780			.334	.200		-.052		-.087		.102
$\alpha=-3.0^\circ; \beta=0.3^\circ$														
P.L.	.732	.473	.354	.292	.239	.138	.044	.009	-.016	-.050	-.055	-.073	-.040	.076
33 $\frac{1}{2}$.138	.035		-.054	-.050				
22 $\frac{1}{2}$.292		.207	.102		-.064	-.119	-.079	-.073		
12		.473		.479	.398	.327	.163	.071	-.033	-.115	-.109	-.071	-.040	
3	.732	.725	.668	.668	.665		.227		-.027		-.083	-.097	-.032	.076
0		.758		.694			.273	.146		-.083		-.096		.084

TABLE XVII. PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED ROUND-WINDSHIELD CANOPY - Concluded

(b) $M=2.01$

x/λ β, deg	.010	.030	.068	.096	.133	.176	.224	.260	.302	.360	.440	.600	.856	.992
$\alpha=0^\circ; \beta=0.3^\circ$														
P.L.	.651	.421	.314	.272	.224	.139	.042	.007	-.019	-.055	-.063	-.092	-.048	.064
33 $\frac{1}{2}$.139	.034		-.053	-.055				
22 $\frac{1}{2}$.272		.191	.088		-.070	-.124	-.085	-.092		
12		.421		.437	.357	.287	.132	.044	-.053	-.131	-.121	-.081	-.048	
3	.651	.658	.594	.590	.586		.176		-.058		-.102	-.104	-.040	.064
0	.683			.615			.217	.100		-.107		-.106		.072
$\alpha=3.0^\circ; \beta=0.3^\circ$														
P.L.	.567	.370	.279	.249	.204	.135	.045	.007	-.020	-.063	-.072	-.111	-.058	.055
33 $\frac{1}{2}$.135	.031		-.052	-.063				
22 $\frac{1}{2}$.249		.175	.075		-.074	-.125	-.092	-.111		
12		.370		.395	.315	.249	.100	.020	-.073	-.144	-.132	-.093	-.058	
3	.567	.588	.524	.518	.512		.124		-.084		-.118	-.130	-.050	.055
0	.612			.540			.161	.051		-.131		-.104		.074
$\alpha=6.0^\circ; \beta=0.3^\circ$														
P.L.	.488	.323	.244	.223	.182	.125	.045	.013	-.015	-.064	-.075	-.128	-.068	.050
33 $\frac{1}{2}$.125	.036		-.051	-.064				
22 $\frac{1}{2}$.223		.161	.064		-.079	-.125	-.100	-.128		
12		.323		.359	.281	.215	.076	-.003	-.093	-.159	-.145	-.110	-.068	
3	.488	.528	.462	.453	.445		.082		-.109		-.131	-.094	-.056	.050
0	.550			.474			.112	.014		-.149		-.096		.072
$\alpha=9.0^\circ; \beta=0.3^\circ$														
P.L.	.417	.280	.215	.201	.165	.120	.045	.019	-.007	-.057	-.074	-.135	-.077	.029
33 $\frac{1}{2}$.120	.045		-.040	-.057				
22 $\frac{1}{2}$.201		.158	.060		-.078	-.118	-.100	-.135		
12		.280		.331	.255	.190	.060	-.019	-.104	-.167	-.151	-.124	-.077	
3	.417	.469	.412	.398	.387		.044		-.127		-.138	-.092	-.067	.029
0	.494			.416			.072	-.016		-.163		-.092		.056
$\alpha=12.0^\circ; \beta=0.3^\circ$														
P.L.	.348	.246	.191	.183	.150	.104	.039	.018	-.007	-.053	-.074	-.142	-.097	.019
33 $\frac{1}{2}$.104	.049		-.033	-.053				
22 $\frac{1}{2}$.183		.156	.057		-.079	-.116	-.102	-.142		
12		.246		.309	.232	.168	.040	-.035	-.117	-.177	-.158	-.147	-.097	
3	.348	.409	.368	.349	.337		.011		-.144		-.139	-.092	-.096	.019
0	.432			.365			.037	-.046		-.170		-.089		.046

TABLE XVII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED SPACER PLAT-WINDSHIELD CANOPY

(a) $M=1.41$

x/y	-0.15	0	.004	.005	.006	.023	.039	.059	.084	.095	.097	.099	.099	.099
θ, deg														
$\alpha=0.4^\circ; \beta=0^\circ$														
$P_{e,L}$.355		.259	.151		.112	.042	-.028	-.088	-.097	.011	.020	-.027	-.036
80	.355			.185			-.079	-.047		-.097				
60	.365			.260		.150	-.076	-.101	-.110		.011			
40	.467		.250	.253		.214	-.033	-.147		-.070		.020		
32			.333											
20		.542				.283	.203	-.197	-.186		.019		-.027	
11						.235								
0			.538			.283	.060	-.198			.016		-.036	
$\alpha=0.4^\circ; \beta=-4^\circ$														
$P_{e,L}$.414		.346	.271		.192	.139	.016	-.049	-.105	-.045	-.007	-.044	-.049
80	.414			.296			.019	-.001		-.105				
60	.433			.489		.252	.033	-.044	-.098		-.045			
40	.491		.337	.358		.291	.028	-.079		-.133		-.007		
32			.375											
20		.530				.303	.257	-.159	-.191		-.022		-.044	
11						.256								
0			.531			.266	.050	-.201			.003		-.049	
-11						.195								
-20		.551				.229	.091	-.236	-.237		.014		-.039	
-32			.253											
-40	.437		.070	.132		.112	-.132	-.267		-.051		.011		
-60	.196			.155		.042	-.178	-.179	-.129		.007			
-80	.271			.078		-.174	-.097		-.083					
$P_{e,U}$.271		.161	.115		.074	-.035	-.089	-.117	-.083	.007	.011	-.039	-.045

TABLE XVII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED SMALL PLAT-WINDSHIELD CANOPY - Continued

(a) $M=1.41$

$\frac{x}{l}$	-0.15	0	0.004	0.065	0.164	0.223	0.239	0.309	0.382	0.455	0.527	0.655	0.760	0.869	0.993
θ, deg															
$\alpha=0.4^\circ; \beta=-8^\circ$															
$P_{u,0}$.465			.431	.348		.272	.172	.066	-.022	-.105	-.143	-.065	-.092	-.093
80	.465			.393			.104	.038		-.105					
60	.557			.548			.342	.123	.017	.097		-.143			
40	.489			.411	.431		.342	.075	-.055		-.170		-.065		
32				.392											
20		.512				.303	.277		-.142	-.203		-.097		-.092	
11							.255								
0			.522				.240	.015	-.211				-.042		-.093
-11							.156								
-20		.557				.177	-.135		-.274	-.334		-.040		-.073	
-32				.178											
-40	.408			-.192	.022		-.013	-.183	-.345		-.091		-.045		
-60	-.010				.023		-.069	-.264	-.269	-.184		-.075			
-80	.171				-.028			-.252	-.173		-.202				
$-P_{u,0}$.171			.071	.032		-.044	-.116	-.172	-.220	-.202	-.075	-.045	-.073	-.090
$\alpha=6.5^\circ; \beta=0^\circ$															
$P_{u,0}$.294			.176	.159		.076	.014	-.060	-.115	-.129	-.030	-.006	-.042	-.036
80	.294				.148			-.055	.004		-.129				
60	.311				.172		.137	.019	-.115	-.173		-.030			
40	.311			.324	.189		.142	-.055	-.169		-.097		-.006		
32				.255											
20		.424				.176	.125		-.235	-.182		-.013		-.042	
11							.141								
0			.429				.181	-.009	-.215				-.006		-.036
$\alpha=6.5^\circ; \beta=-4^\circ$															
$P_{u,0}$.322			.253	.223		.142	.066	-.020	-.092	-.175	-.098		-.061	-.060
80	.322			.247	.227			.015			-.175				
60	.337			.241			.210	.056	-.087	-.195		-.098			
40	.301			.379	.256		.186	-.025	-.151		-.211				
32				.294											
20	.405					.185	.165		-.206	-.233		-.053		-.061	
11							.157								
0		.410					.165	-.017	-.215				-.041		-.060
-11							.100								
-20	.431					.132	.025		-.276	-.186		-.026		-.090	
-32				.182											
-40	.295			.144	.089		.053	-.115	-.237		-.073				
-60	.165				.073		.052	-.106	-.157	-.164		-.016			
-80	.235			.110	.065			-.129		-.088					
$-P_{u,0}$.235			.117	.098		.018	-.041	-.096	-.135	-.088	-.016		-.090	-.059
$\alpha=6.5^\circ; \beta=-8^\circ$															
$P_{u,0}$.342			.287	.323		.206	.123	.028	-.060	-.156	-.226	-.046	-.117	-.071
80	.342				.301			.079	.119		-.196				
60	.342				.315		.269	.111	-.077	-.216		-.226			
40	.257			.366	.299		.216	.000	-.170		-.281		-.046		
32				.301											
20		.378				.168	.162		-.216	-.279		-.109		-.117	
11							.146								
0			.392				.129	-.061	-.243				-.117		-.071
-11							.057								
-20	.433					.079	-.193		-.220	-.285		-.117		-.066	
-32				.105											
-40	.295			-.095	-.024		-.042	-.151	-.341		-.121				
-60	-.059				-.024		-.036	-.199	-.212	-.185		-.111			
-80	.157			.039	-.010			-.186			-.119				
$-P_{u,0}$.157			.043	.036		-.042	-.100	-.143	-.164	-.119	-.111		-.066	-.067

TABLE XVII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED SMALL FLAT-WINDSHIELD CANOPY - Continued

(b) $M=2.01$

x/y θ, deg	-015	0	.004	.065	.164	.223	.239	.309	.382	.455	.527	.655	.760	.869	.993
$\alpha=0.4^\circ; \beta=0^\circ$															
P.L.	.276			.276	.148		.085	.067	.013	-.022	-.037	-.004	.035	.008	-.010
80	.276			.179	.130		.070	-.042	.001		-.037				
60				.241	.206		.156	.016	-.076	-.040		-.004			
40	.376			.254	.241		.065	-.080		-.042		.035			
32				.323											
20		.476		.379		.298	.194	.112	-.074	-.125		.024		.008	
11						.259									
0			.545			.308	.117	-.069				.037		-.010	
$\alpha=0.4^\circ; \beta=-4^\circ$															
P.L.	.337			.243	.224		.154	.134	.070	.017	-.027	-.032	-.005	-.002	-.019
80	.337			.258	.227		.157	.046	.049		-.027				
60				.329	.326		.244	.078	-.004	-.018		-.032			
40	.348			.369	.317		.117	-.042		-.038		-.005			
32				.382											
20		.414		.405		.327	.257	.122	-.062	-.086		-.027		-.002	
11						.282									
0			.533			.305	.114	-.069				.021		-.019	
-11						.236									
-20		.510		.359		.266	.093	.085	-.096	-.148		.020		-.001	
-32				.268											
-40	.409			.074	.154		.006	-.123		-.043		.032			
-60				.146	.113		.063	-.051	-.132	-.074		-.026			
-80	.205			.098	.041		-.012	-.110	-.060		-.099				
-P.L.	.205			.106	.080		.024	-.007	-.047	-.078	-.099	-.026	.032	-.001	-.020
$\alpha=0.4^\circ; \beta=-8^\circ$															
P.L.	.392			.312	.310		.231	.204	.130	.064	-.011	-.040	-.098	-.031	-.048
80	.392			.331	.329		.253	.142	.107		-.011				
60				.405	.459		.328	.154	.062	-.002		-.040			
40	.336			.438	.393		.169	.004		-.053		-.098			
32				.424											
20		.349		.426		.350	.300	.148	-.064	-.085		-.123		-.031	
11						.304									
0			.492			.308	.054	-.092				-.028		-.048	
-11						.232									
-20		.525		.365		.255	-.039	.011	-.119	-.169		-.043		-.045	
-32				.253											
-40	.433			-.113	-.051		-.048	-.166		-.064		-.023			
-60				.060	.024		-.020	-.119	-.155	-.118		-.018			
-80	.091			.024	-.046		-.099	-.145	-.135		-.168				
-P.L.	.091			.045	.015		-.039	-.091	-.118	-.153	-.168	-.018	-.023	-.045	-.055

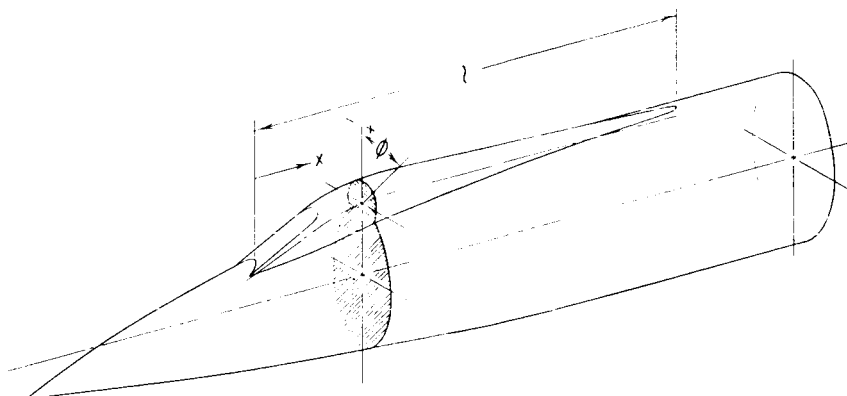
TABLE XVII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED SMALL FLAT-WINDSHIELD CANOPY - Concluded

(b) $M=2.01$

α/β	$\alpha=0^\circ$	$\alpha=10^\circ$	$\alpha=20^\circ$	$\alpha=30^\circ$	$\alpha=40^\circ$	$\alpha=50^\circ$	$\alpha=60^\circ$	$\alpha=70^\circ$	$\alpha=80^\circ$	$\alpha=90^\circ$
β, deg										
$\alpha=6.5^\circ; \beta=0^\circ$										
P.L.	.237	.140	.066	.045	.007	-.050	-.087	-.024	.007	-.012
80	.237	.153	.134	.074	-.011	-.024	-.087			
60		.200	.164	.136	-.004	-.060	-.076	-.024		
40	.257	.223	.179	.024	-.101		-.076		.007	
32		.259								
20	.341	.270	.192	.118	.040	-.114	-.122	-.091	-.012	
11				.156						
0		.409		.198	.034	-.114		.009	-.018	
$\alpha=6.5^\circ; \beta=-4^\circ$										
P.L.	.291	.185	.122	.092	.036	-.022	-.097	-.104	-.023	-.034
80	.291	.203	.197	.137	.051	.008	-.097			
60		.255	.253	.187	.047	-.032	-.087	-.104		
40	.228	.292	.230	.053	-.075		-.104	-.023		
32		.317								
20	.274	.287	.210	.161	.037	-.112	-.126	-.047	-.034	
11				.174						
0	.379		.191	.022	-.122			-.033	-.042	
-11			.135							
-20	.375	.246	.160	.019	.010	-.141	-.164	-.014	-.057	
-32		.194								
-40	.284	.089	.098		-.028	-.145	-.062	-.023		
-60		.122	.083	.054	-.064	-.115	-.094	-.010		
-80	.161	.092	.059	-.001	-.077	-.060	-.087			
-P.L.	.161	.089	.016	-.006	-.046	-.076	-.087	-.010	-.023	-.057
$\alpha=6.5^\circ; \beta=-8^\circ$										
P.L.	.344	.242	.189	.132	.082	.010	-.079	-.117	-.048	-.064
80	.344	.252	.273	.212	.120	.050	-.079			
60		.300	.337	.257	.102	.001	-.075	-.117		
40	.199	.337	.283	.082	-.050		-.140	-.048		
32		.329								
20	.202	.302	.218	.185	.055	-.120	-.151	-.116	-.064	
11				.186						
0	.330		.186	-.032	-.149			-.093	-.067	
-11			.129							
-20	.372	.238	.140	-.106	-.082	-.174	-.195	-.112	-.070	
-32		.146								
-40	.304	-.091	-.082		-.093	-.193	-.095	-.148		
-60		.046	.002	-.027	-.126	-.164	-.150	-.110		
-80	.047	.029	-.011	-.066	-.128	-.114	-.118			
-P.L.	.047	.036	-.033	-.059	-.088	-.109	-.118	-.110	-.148	-.070

CONFIDENTIAL

TABLE XVIII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED SMALL FLAT-WINDSHIELD CANOPY

(a) $M=1.41$

x/λ	-.011	0	.004	.052	.102	.142	.149	.193	.249	.324	.467	.662	.804	.985
z, deg														
$\alpha=0.4^\circ; \beta=0^\circ$														
P.L.	.298		.285	.152	.121	.071		-.005	-.031	-.060	-.111	-.060	-.066	-.022
80				.152				-.038		-.068	-.111			
60	.298			.159	.120	.073			-.108		-.104	-.060		
48			.356											
43	.388			.242	.166	.105				-.153	-.094		-.066	
40			.447											
32				.326										
29						.148								
26						.271								
20		.516		.338		.216	.100		-.072	-.202	-.075	-.040		
0			.529	.354	.308		.225	.089	-.049	-.224	-.063	-.040	-.044	-.022
$\alpha=0.4^\circ; \beta=-4^\circ$														
P.L.	.333		.337	.245	.211	.150		.058	.032	-.017	-.147	-.093	-.098	-.042
80				.245				.050		-.028	-.147			
60	.333			.265	.224	.173			.031		-.173	-.093		
48			.408											
43	.373			.336	.265	.201				-.126	-.197		-.098	
40			.450											
32				.368										
29						.238								
26						.318								
20		.488		.351		.263	.180		-.056	-.197	-.135	-.068		
0		.514	.346	.305		.227	.049	.126	-.239	-.099	-.064	-.068	-.042	
-20		.515	.314		.150	-.034		-.122	-.278	-.086	-.063			
-26					.191									
-29						-.006								
-32					.262									
-40			.407											
-43	.370			.096	.017	-.031				-.215	-.080		-.088	
-48			.242											
-60	.229			.024	-.015	-.054			-.223		-.080	-.071		
-80				.046				-.133		-.119	-.082			
-P.L.	.229	.201	.046	.022	-.012		-.082	-.143	-.121	-.067	-.071	-.088	-.037	

CONFIDENTIAL

TABLE XVIII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED SMALL FLAT-WINDSHIELD CANOPY - Continued

(a) $M=1.41$														
x/l	-0.11	0	$.004$	$.052$	$.102$	$.142$	$.149$	$.193$	$.249$	$.324$	$.467$	$.662$	$.804$	$.985$
θ, deg	$\alpha=0.4^\circ; \beta=-8^\circ$													
P.L.	.355	.379	.336	.307	.236	.178	.106	.034	-.140	-.332	-.514	-.689		
80			.336			.143		.014	-.140					
60	.355		.357	.318	.263		.044		-.257	-.332				
48		.416												
43	.336		.392	.338	.276				-.115	-.267		-.114		
40		.403												
32			.391											
29					.287									
26					.341									
20	.431		.360		.290	.225	-.051	-.218	-.337	-.414				
0	.482	.336	.298		.223	.014	-.123	-.288	-.172	-.110	-.111	-.089		
-20	.495		.284		.096	-.152	-.186	-.377	-.150	-.127				
-26				.121										
-29					-.235									
-32			.190											
-40		.354												
-43	.328		-.092	-.155	-.211				-.313	-.167		-.128		
-48		.086												
-60	.128		-.128	-.183	-.197		-.368		-.202	-.238				
-80		-.062				-.212		-.191	-.114					
-P.L.	.128	.096	-.062	-.073	-.101	-.166	-.194	-.217	-.114	-.238	-.128	-.074		
$\alpha=6.5^\circ; \beta=0^\circ$														
P.L.	.227	.220	.125	.104	.060		-.019	-.046	-.086	-.163	-.140	-.070	-.014	
80			.125				-.026		-.096	-.163				
60	.227		.136	.103	.063			-.107			-.140			
48		.280												
43	.290		.195	.123	.077				-.179	-.123		-.070		
40		.340												
32			.246											
29					.101									
26					.194									
20	.392		.252		.142	.045	-.095	-.220	-.086	-.054				
0	.406	.264	.214		.139	.012	-.149	-.236	-.077	-.050	-.039	-.014		
$\alpha=6.5^\circ; \beta=-4^\circ$														
P.L.	.246	.260	.203	.185	.132		.037	.009	-.048	-.205	-.253	-.093	-.027	
80			.203				.049		-.071	-.205				
60	.246		.223	.186	.143			-.054			-.253			
48		.314												
43	.266		.264	.199	.151				-.179	-.262		-.093		
40		.332												
32			.278											
29					.167									
26					.233									
20	.370		.262		.182	.110	-.085	-.244	-.152	-.087				
0	.395	.257	.211		.140	-.031	-.167	-.271	-.111	-.118	-.067	-.027		
-20	.395		.227		.074	-.075	-.135	-.256	-.098	-.091				
-26				.119										
-29					-.033									
-32			.187											
-40		.313												
-43	.278		.065	-.000	-.036				-.244	-.094		-.095		
-48		.195												
-60	.176		.022	-.006	-.039		-.164			-.054				
-80		.036					-.106		-.129	-.058				
-P.L.	.176	.155	.036	.019	-.013		-.061	-.103	-.121	-.098	-.054	-.095	-.028	

CONFIDENTIAL

TABLE XVIII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED SMALL FLAT-WINDSHIELD CANOPY - Continued

(a) $M=1.41$

α/β	-0.11	0	$.004$	$.052$	$.102$	$.142$	$.149$	$.193$	$.249$	$.324$	$.467$	$.662$	$.804$	$.985$
β, deg	$\alpha=6.5^\circ; \beta=-8^\circ$													
P.L.	.251	.288	.277	.273	.210		.095	.070	.007	-.198	-.276	-.120	-.053	
80			.277				.125		-.031	-.198				
60	.251		.293	.260	.212			.004			-.276			
48		.305												
43	.213		.294	.248	.204				-.197	-.372		-.120		
40		.268												
32			.285											
29					.191									
26				.243										
20		.312		.264	.196	.138		-.099	-.299	-.264	-.151			
0		.361	.245	.201		.129	-.066	-.234	-.348	-.242	-.172	-.074	-.053	
-20		.376		.201		.010	-.204		-.197	-.386	-.211	-.245		
-26				.042										
-29					-.246									
-32				.123										
-40		.271												
-43	.241		-.301	-.263	-.160				-.251	-.174		-.149		
-48		.069												
-60	.096		-.309	-.135	-.157			-.282			-.142			
-80		-.053					-.179	-.178	-.073					
-P.L.	.096	.071	-.053	-.064	-.087		-.150	-.168	-.172	-.073	-.142	-.149	-.047	

(b) $M=2.01$

α/β	$\alpha=0.4^\circ; \beta=0^\circ$													
P.L.	.229	.222	.142	.122	.089		.045	.022	-.001	-.067	-.064	-.030	-.010	
80			.142				.007		-.007	-.067				
60	.229		.172	.123	.102			-.049		-.030	-.064			
48		.284												
43	.318		.258	.191	.134				-.081	-.063		-.030		
40		.372												
32			.354											
29					.170									
26				.293										
20		.449	.375		.253	.144		-.022	-.112	-.038	-.015			
0		.499	.390	.338		.304	.114	.005	-.114	-.027	-.012	-.019	-.010	
α/β	$\alpha=0.4^\circ; \beta=-4^\circ$													
P.L.	.270	.282	.218	.204	.159		.101	.076	.049	-.063	-.076	-.047	-.029	
80			.218				.083		.035	-.063				
60	.270		.261	.220	.194			.029		-.074	-.076			
48		.336												
43	.321		.334	.287	.224				-.044	-.096		-.047		
40		.364												
32			.389											
29					.252									
26				.336										
20		.406	.390		.294	.215		-.023	-.100	-.114	-.044			
0		.474	.390	.340		.306	.121	-.007	-.123	-.073	-.027	-.044	-.029	
-20		.457	.351		.209	.061		-.047	-.150	-.056	-.040			
-26				.237										
-29					.050									
-32				.297										
-40		.355												
-43	.287		.210	.074	.016				-.154	-.054		-.056		
-48		.195												
-60	.164		.063	.006	-.009			-.135		-.052	-.061			
-80			.056				-.073	-.063	-.071					
-P.L.	.164	.145	.056	.039	.019		-.023	-.044	-.062	-.071	-.061	-.056	-.032	

CONFIDENTIAL

TABLE XVIII. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED SMALL FLAT-WINDSHIELD CANOPY - Continued

(b) $M=2.01$

α/β	-0.11	0	$.004$	$.052$	$.102$	$.142$	$.149$	$.193$	$.249$	$.324$	$.467$	$.662$	$.804$	$.985$
$\alpha=0.4^\circ; \beta=-8^\circ$														
$P_{z/L}$.301		.334	.286	.280	.235		.165	.136	.105	-.039	-.076	-.091	-.083
80				.286				.162		.088	-.039			
60	.301			.323	.302	.277			.103		-.116	-.076		
48			.308											
43	.308			.371	.359	.297				-.016	-.116		-.091	
40			.334											
32				.399										
29						.309								
26						.371								
20		.347		.393		.326	.263		-.012	-.094	-.166	-.105		
0		.417		.386	.341		.308	.135	-.031	-.143	-.163	-.097	-.076	-.083
-20		.442		.341		.193	.030		-.082	-.184	-.118	-.102		
-26					.207									
-29						-.093								
-32					.260									
-40			.343											
-43	.252			.004	-.030	-.071				-.185	-.109		-.103	
-48			.094											
-60	.094			-.035	-.102	-.128			-.193		-.112	-.141		
-80				-.020				-.130		-.169	-.091			
$P_{x/L}$.094	.072	-.070	-.036	-.049			-.087	-.116	-.127	-.091	-.141	-.103	-.076
$\alpha=6.5^\circ; \beta=0^\circ$														
$P_{z/L}$.161	.164	.108	.094	.062		.018	-.007	-.026	-.109	-.116	-.052	-.010	
80			.108				-.003		-.037	-.109				
60	.161		.133	.095	.077			-.053		-.052	-.116			
48		.211												
43	.218		.194	.140	.093				-.101	-.102		-.052		
40		.267												
32			.256											
29					.110									
26					.200									
20		.325	.266		.163	.079		.002	-.137	-.062	-.038			
0		.362	.280	.229		.197	.035	-.061	-.151	-.049	-.033	-.028	-.010	
$\alpha=6.5^\circ; \beta=-4^\circ$														
$P_{z/L}$.187	.211	.168	.163	.126		.073	.042	.019	-.094	-.136	-.072	-.023	
80			.168				.060		.007	-.094				
60	.187		.201	.175	.153			.009		-.082	-.136			
48		.248												
43	.210		.251	.217	.166				-.083	-.164		-.072		
40		.247												
32			.282											
29					.160									
26					.240									
20		.284	.276		.199	.141		-.005	-.140	-.152	-.082			
0		.341	.277	.228		.197	.041	-.076	-.163	-.083	-.107	-.058	-.023	
-20		.336	.265		.123	.065		-.013	-.147	-.066	-.071			
-26				.151										
-29					.002									
-32			.212											
-40		.261												
-43	.265		.166	.042	.001				-.133	-.067		-.058		
-48		.149												
-60	.170		.047	.008	-.006			-.123		-.044	-.075			
-80			.042					-.065		-.071	-.094			
$P_{x/L}$.170	.106	.042	.027	.005		-.033	-.053	-.065	-.054	-.075	-.058	-.023	

TABLE XVIII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED SMALL FLAT-WINDSHIELD CANOPY - Concluded

(b) $M=2.01$

x/l	-0.011	0	$.004$	$.052$	$.102$	$.142$	$.149$	$.193$	$.249$	$.324$	$.467$	$.662$	$.804$	$.985$
ϕ, deg	$\alpha=6.5^\circ; \beta=-8^\circ$													
P.L.	.200	.247	.224	.228	.188		.126	.091	.066	-.066	-.124	-.107	-.046	
80			.224				.128		.043	-.066				
60	.200		.242	.232	.214			.058		-.125	-.124			
48		.258												
43	.181		.256	.264	.215				-.063	-.177		-.107		
40		.200												
32			.271											
29					.207									
26				.255										
20	.223		.275		.218	.172		.002	-.139	-.187	-.106			
0		.286	.274	.231		.199	.055	-.096	-.190	-.170	-.145	-.078	-.046	
-20	.307		.234		.101	-.031		-.047	-.212	-.164	-.176			
-26				.120										
-29					-.143									
-32			.171											
-40		.241												
-43	.165		-.042	-.079	-.109				-.194	-.181		-.095		
-48		.050												
-60	.047		-.057	-.105	-.119			-.187		-.113	-.113			
-80			-.035				-.119		-.114	-.079				
-P.L.	.047	.037	-.035	-.043	-.055		-.088	-.107	-.112	-.079	-.113	-.095	-.046	

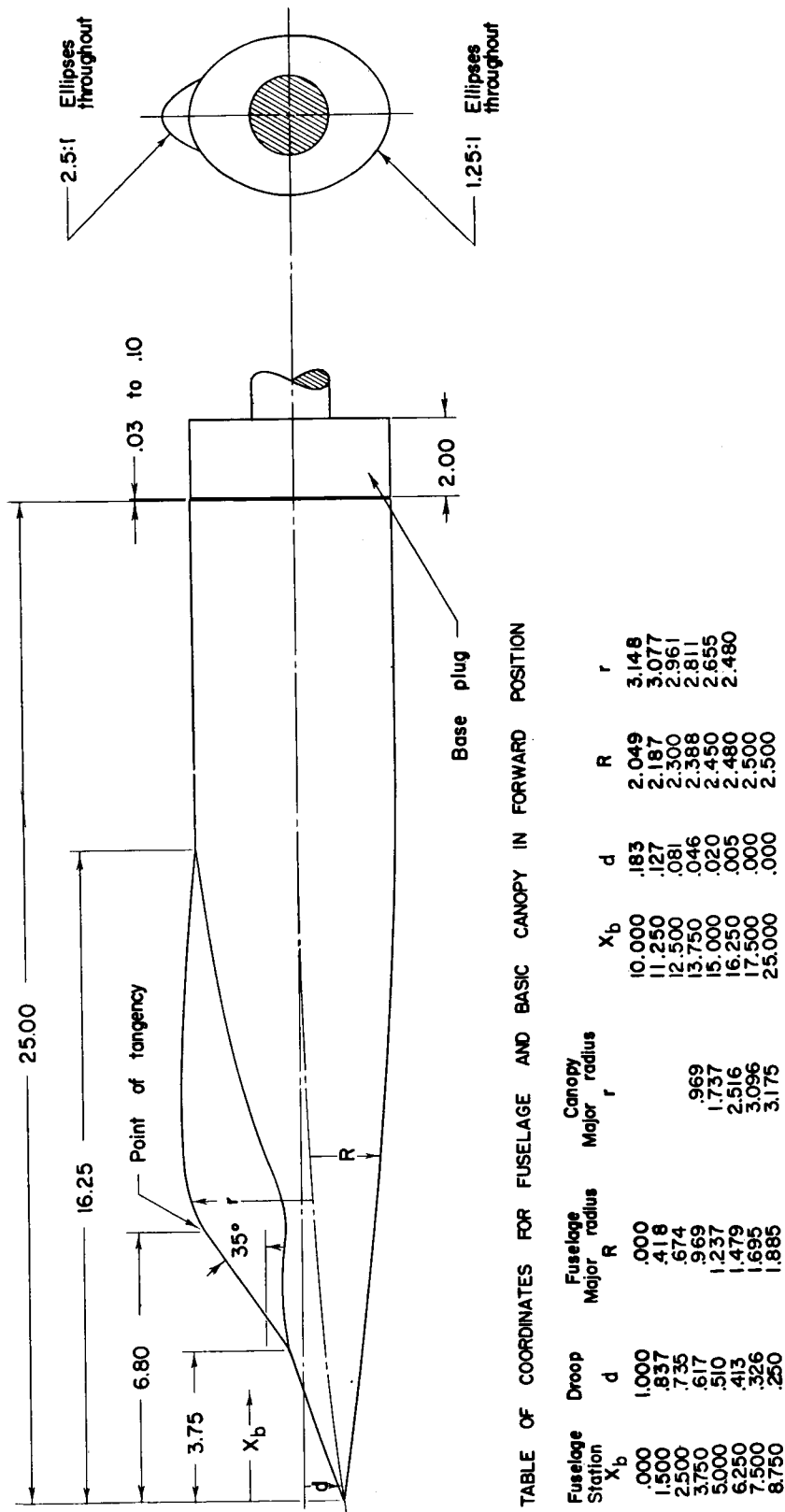


Figure 1.- Details of canopy-fuselage model showing round-windshield canopy in the forward location. All dimensions are in inches.

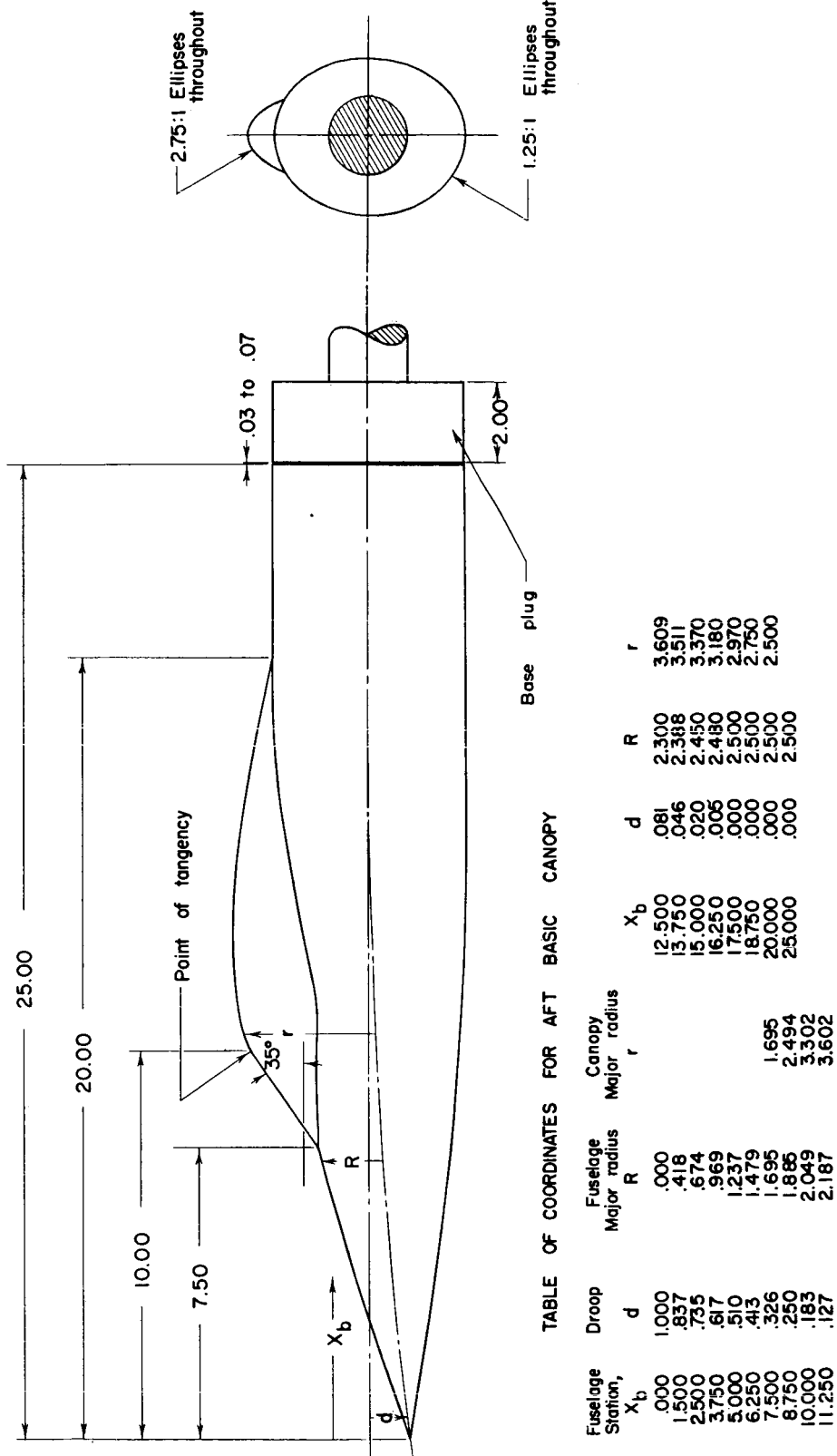
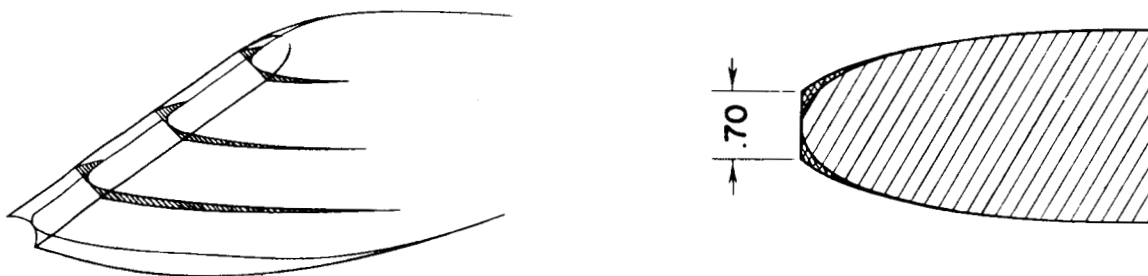
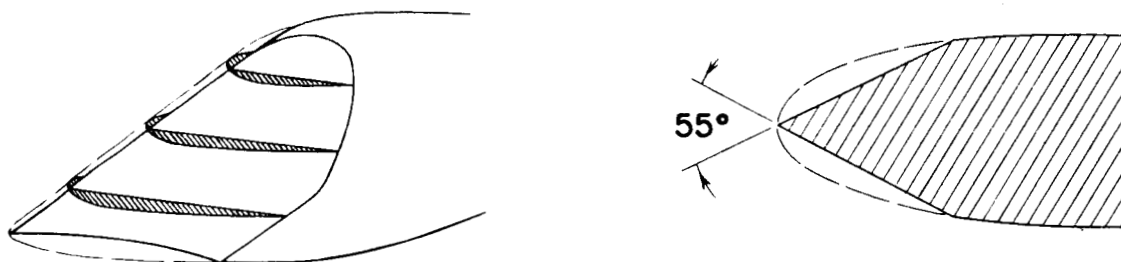


Figure 2.- Details of canopy-fuselage model showing round-windshield canopy in the rearward location. All dimensions are in inches.



Typical section in X-Y plane

(a) Method of development of flat-faced canopies from basic or round-faced canopies.



Typical section in X-Y plane

(b) Method of development of vee-faced canopies from basic or round-faced canopies.

Figure 3.- Method of development of flat and vee-windshield canopies from the basic or round-windshield canopy. All dimensions are in inches.

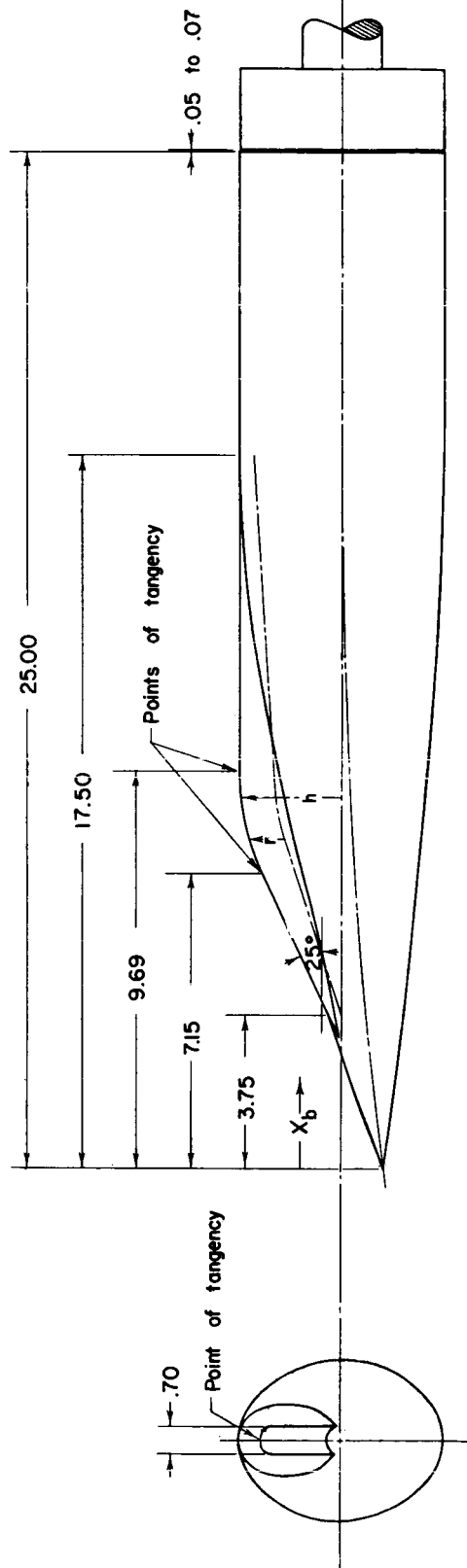


TABLE OF DIMENSIONS FOR SMALL CANOPY IN FORWARD POSITION

Fuselage Station X_b	Canopy Profile h	Canopy Radius r	Fuselage Station X_b	Canopy Profile h	Canopy Radius r
3.750	.325	.352	11.250	2.500	.733
5.000	.935	.529	12.500	2.500	.657
6.250	1.518	.706	13.750	2.500	.581
7.500	2.088	.870	15.000	2.500	.504
8.750	2.427	.849	16.250	2.500	.428
10.000	2.500	.809	17.500	2.500	.352

Note: Dimensions from table describe a round-faced canopy. Flat-faced configuration results from addition of the flat face and filling to fair to basic shape in a manner similar to that shown in figure 3.

Figure 4.- Details of canopy-fuselage model showing small forward-located flat-windshield canopy.
All dimensions are in inches.

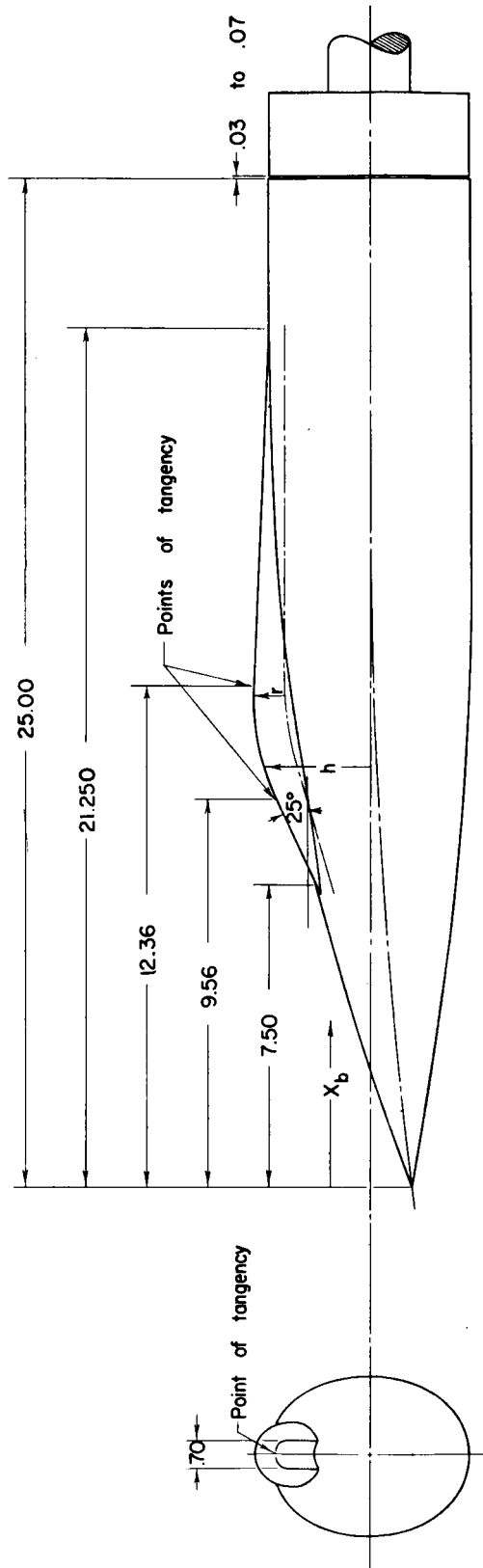
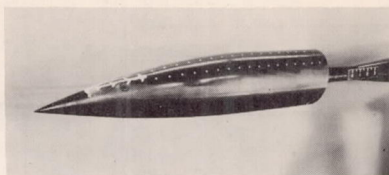


TABLE OF DIMENSIONS FOR SMALL CANOPY IN AFT POSITION

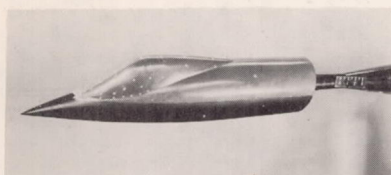
Fuselage Station X_b	Canopy Profile h	Canopy Radius r	Fuselage Station X_b	Canopy Profile h	Canopy Radius r
7.500	1.395	.390	15.000	2.773	.629
8.750	1.952	.592	16.250	2.718	.574
10.000	2.514	.772	17.500	2.664	.520
11.250	2.833	.758	19.375	2.582	.438
12.500	2.882	.738	21.250	2.500	.356
13.750	2.827	.683			

Figure 5.- Details of canopy-fuselage model showing small rearward-located flat-windshield canopy. All dimensions are in inches.

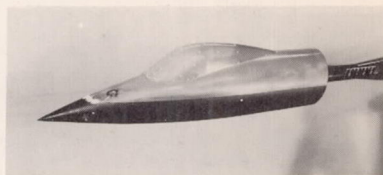


(a) Body alone.

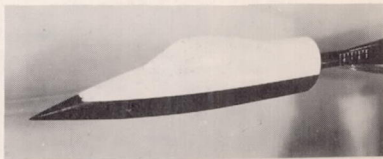
Forward-located canopies



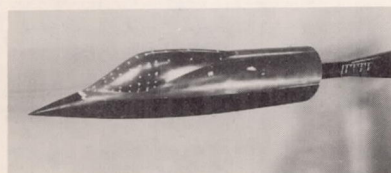
Rearward-located canopies



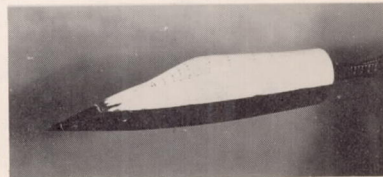
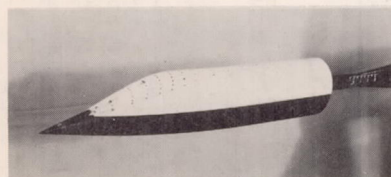
(b) Large flat-windshield configurations.



(c) Large vee-windshield configurations.



(d) Large round-windshield configurations.



(e) Small flat-windshield configurations.

L-89387

Figure 6.- Photographs of models.

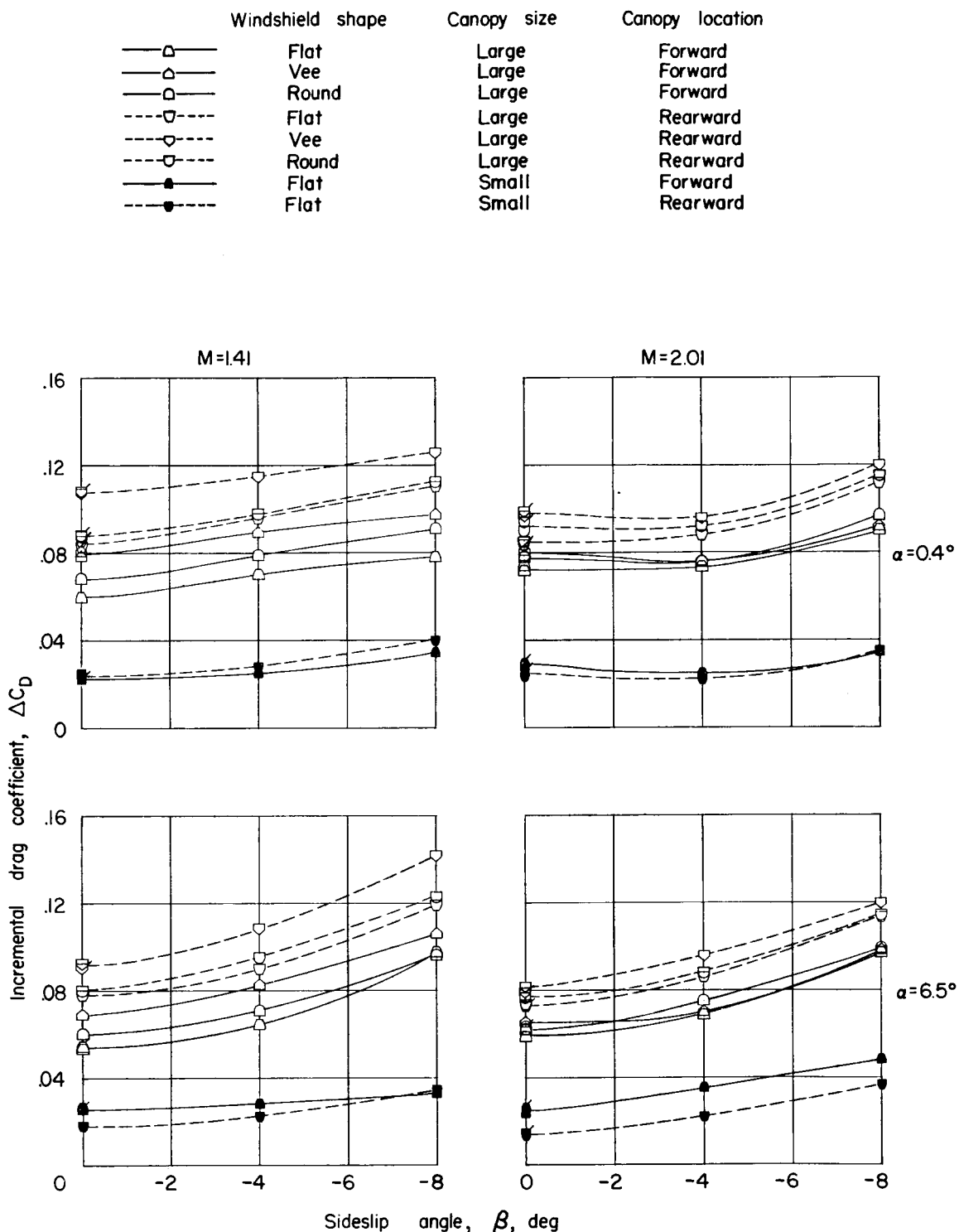


Figure 7.- Incremental drag coefficients for the several canopy configurations at various angles of sideslip for $M = 1.41$ and 2.01 and $\alpha = 0.4^\circ$ and 6.5° . Tailed symbols are check points.

- \square Large forward-located canopy
 ∇ Large rearward-located canopy
 \blacksquare Small forward-located canopy
 \blacksquare Small rearward-located canopy
- $\left. \begin{array}{l} \square \\ \nabla \\ \blacksquare \\ \blacksquare \end{array} \right\} \begin{array}{l} M=1.41 \\ \text{present tests} \end{array}$
- Bodies of revolution; $M=1.40$; ref. 7

K = Location of station of maximum cross-section area, percent of length

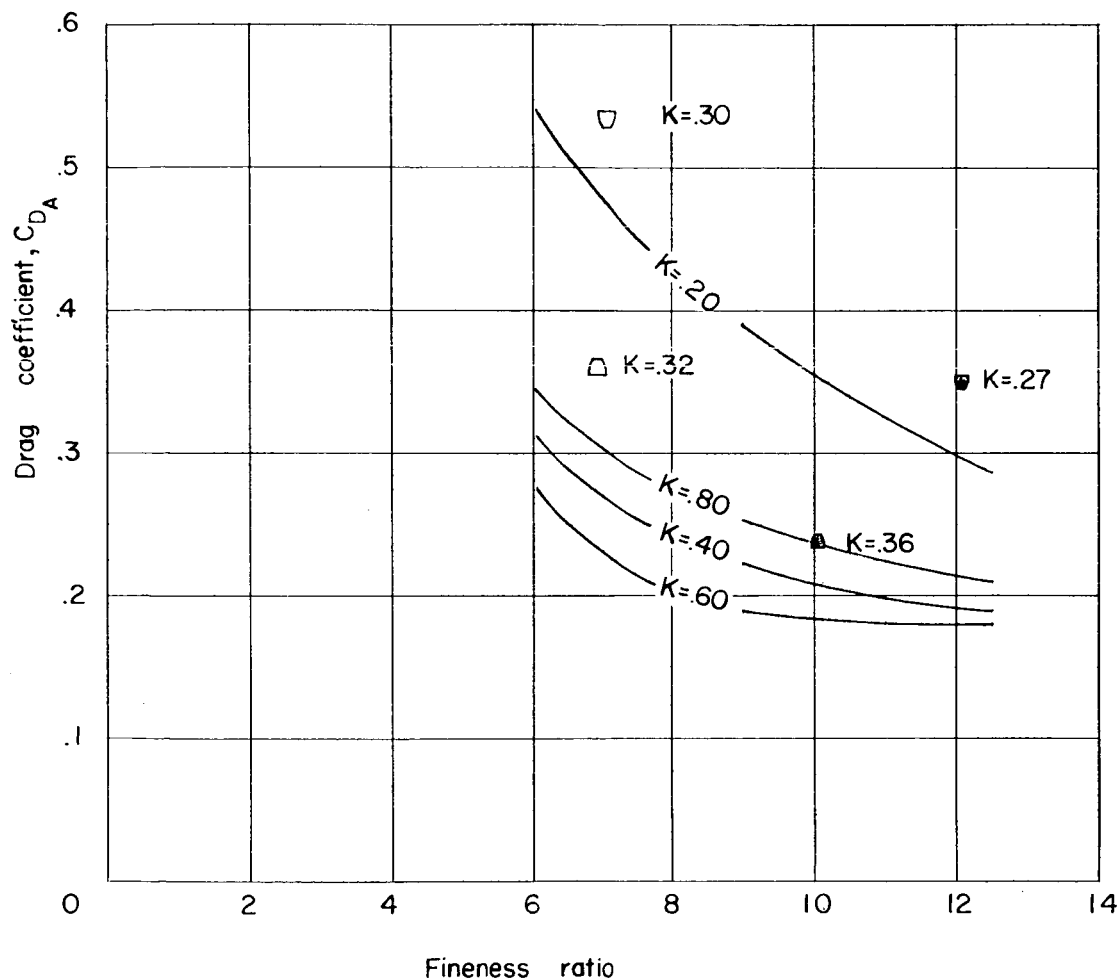
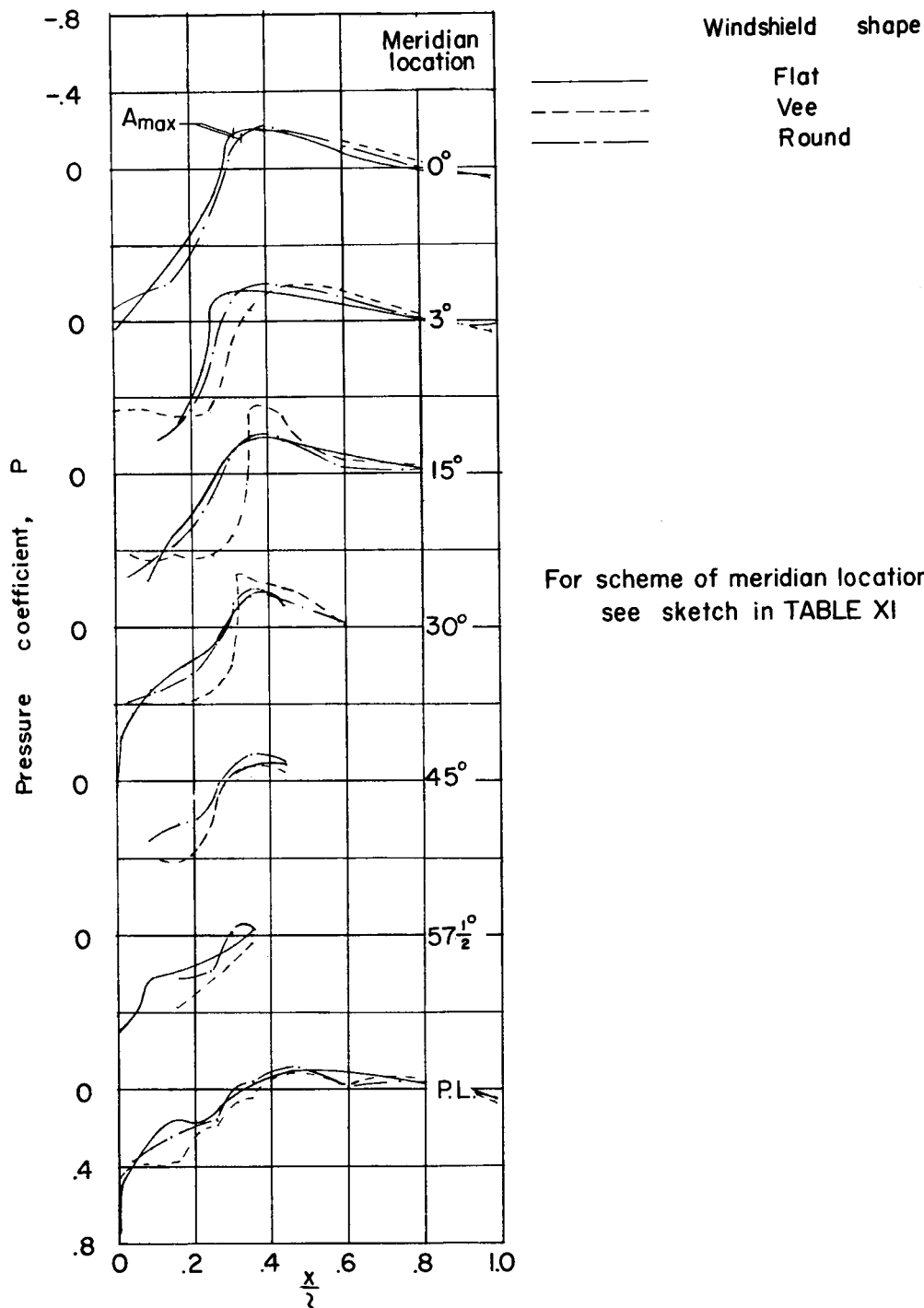
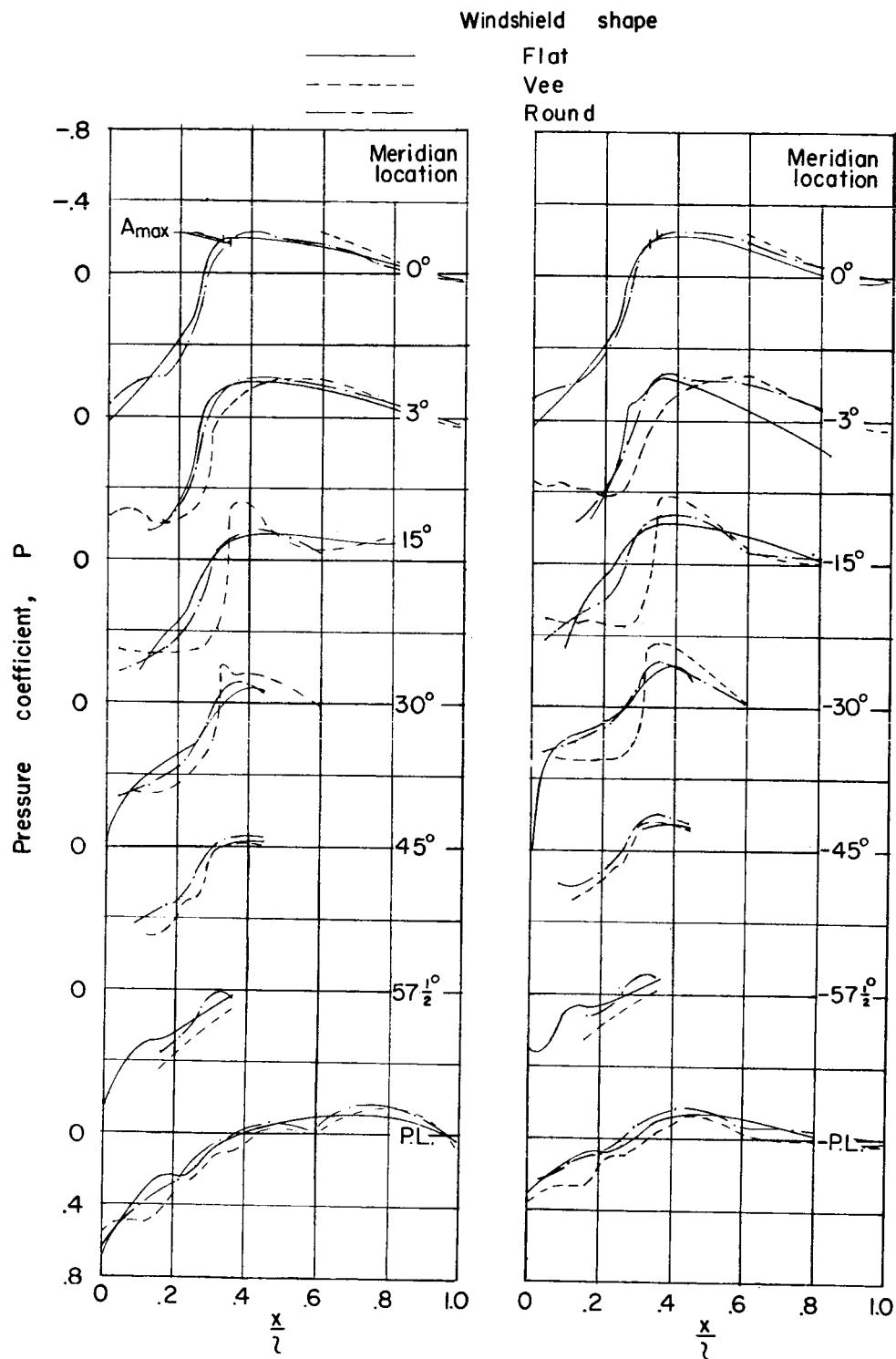


Figure 8.- Incremental drag coefficient C_{DA} (based on canopy maximum cross-section area) for flat-windshield canopies compared with drag coefficients C_{DA} for bodies of revolution having various locations of maximum diameter (ref. 7).



(a) $\alpha = 0.4^\circ$; $\beta = 0^\circ$.

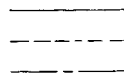
Figure 9.- Effect of windshield shape on pressure-coefficient distributions on large forward-located canopies at $M = 1.41$.



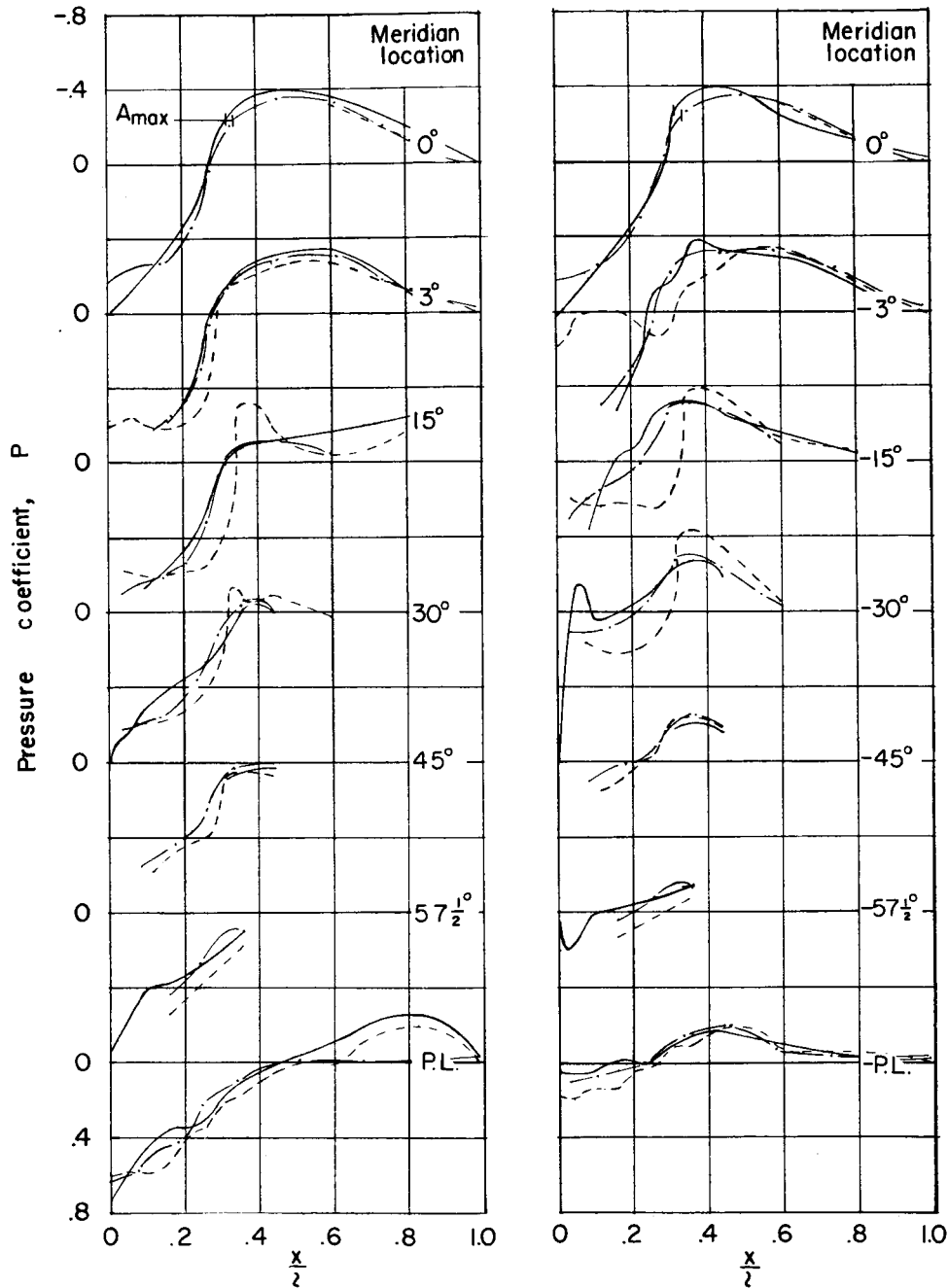
(b) $\alpha = 0.4^\circ$; $\beta = -4^\circ$.

Figure 9.- Continued.

Windshield shape



Flat
Vee
Round



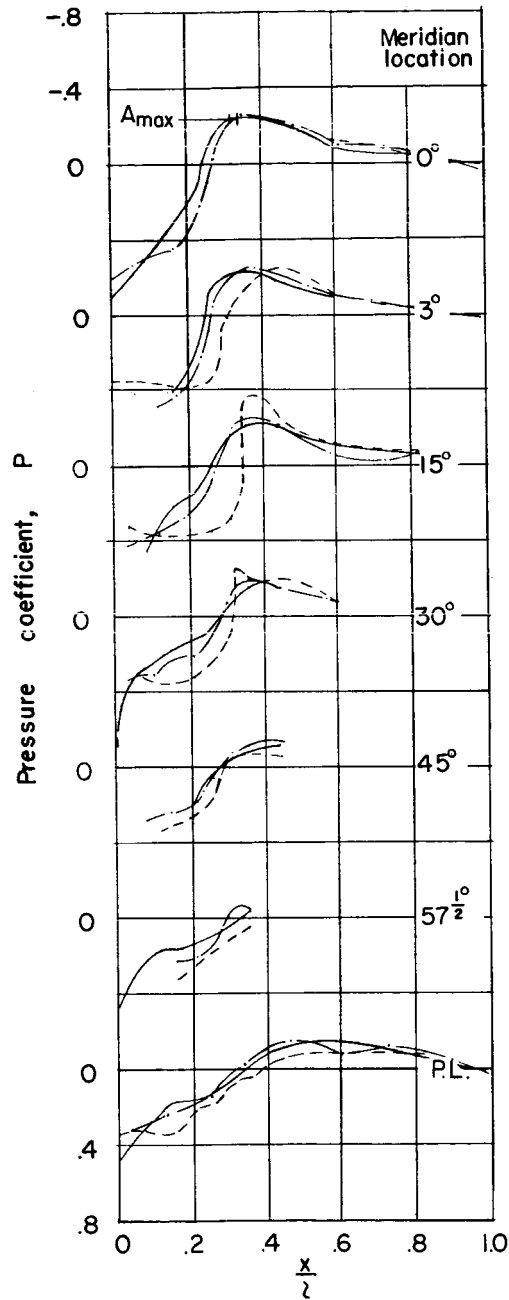
(c) $\alpha = 0.4^\circ$; $\beta = -8^\circ$.

Figure 9.- Continued.

CONFIDENTIAL

Windshield shape

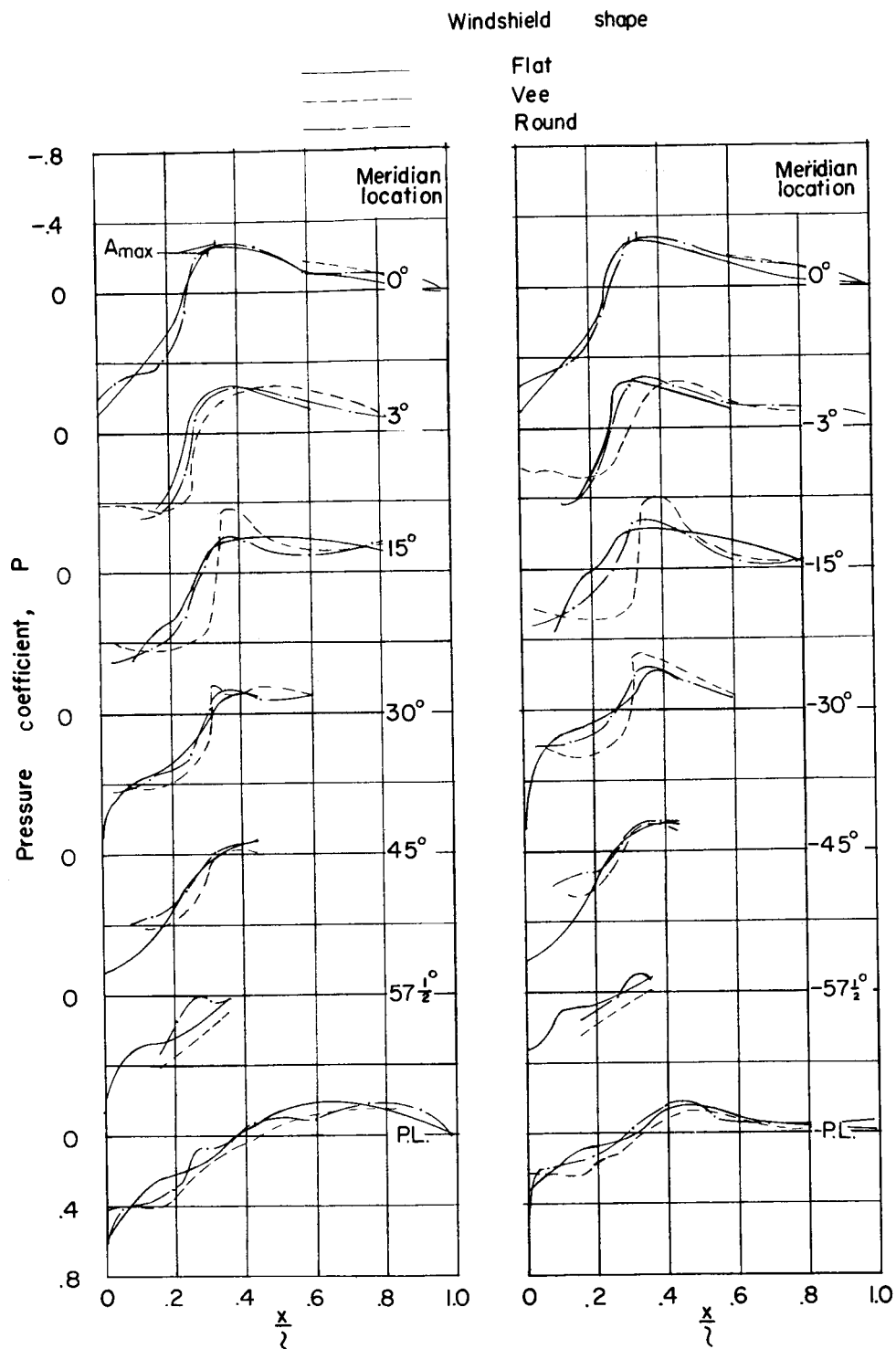
— Flat
 - - - Vee
 - - - Round



(d) $\alpha = 6.5^\circ$; $\beta = 0^\circ$.

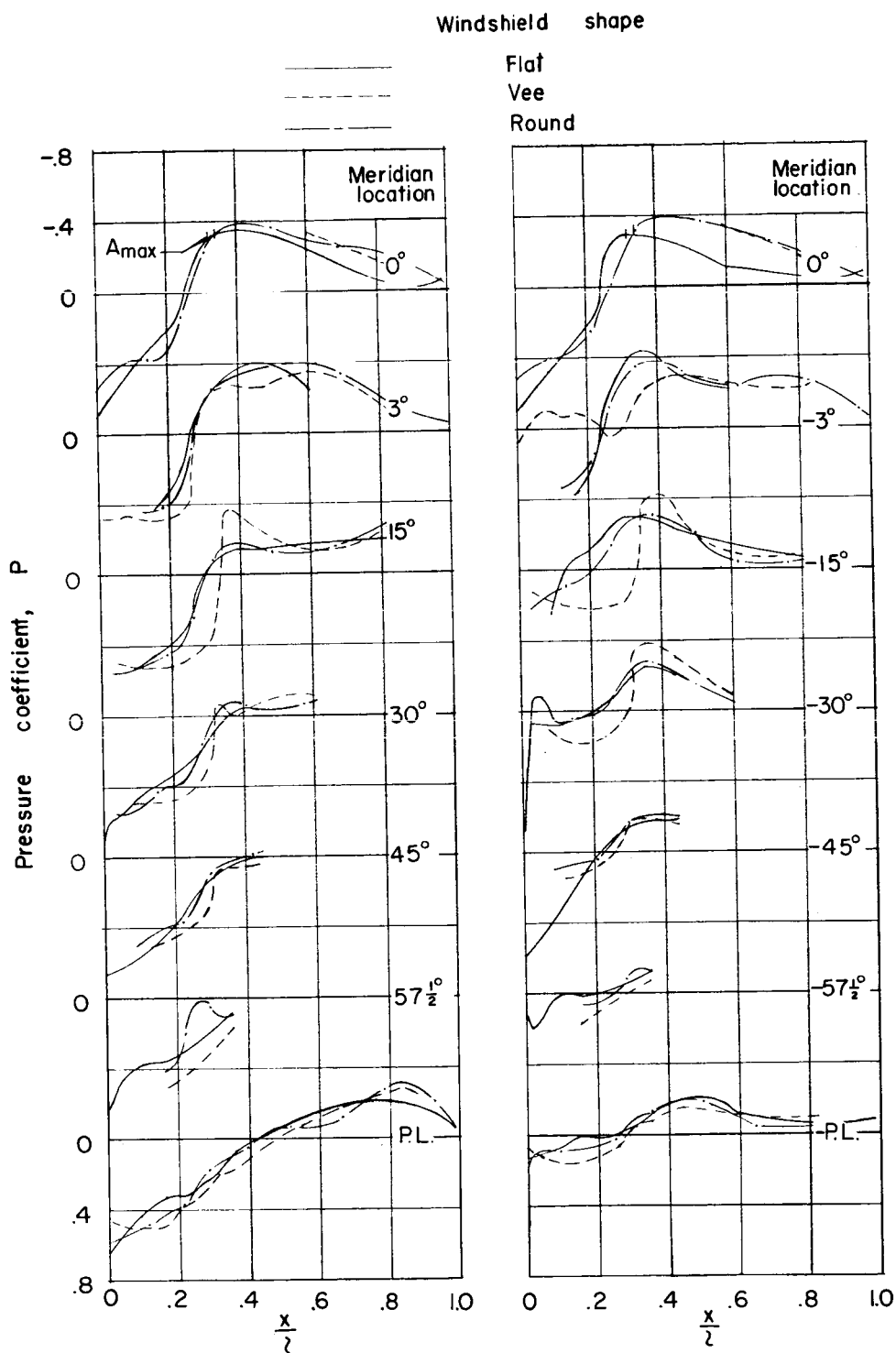
Figure 9.- Continued.

CONFIDENTIAL



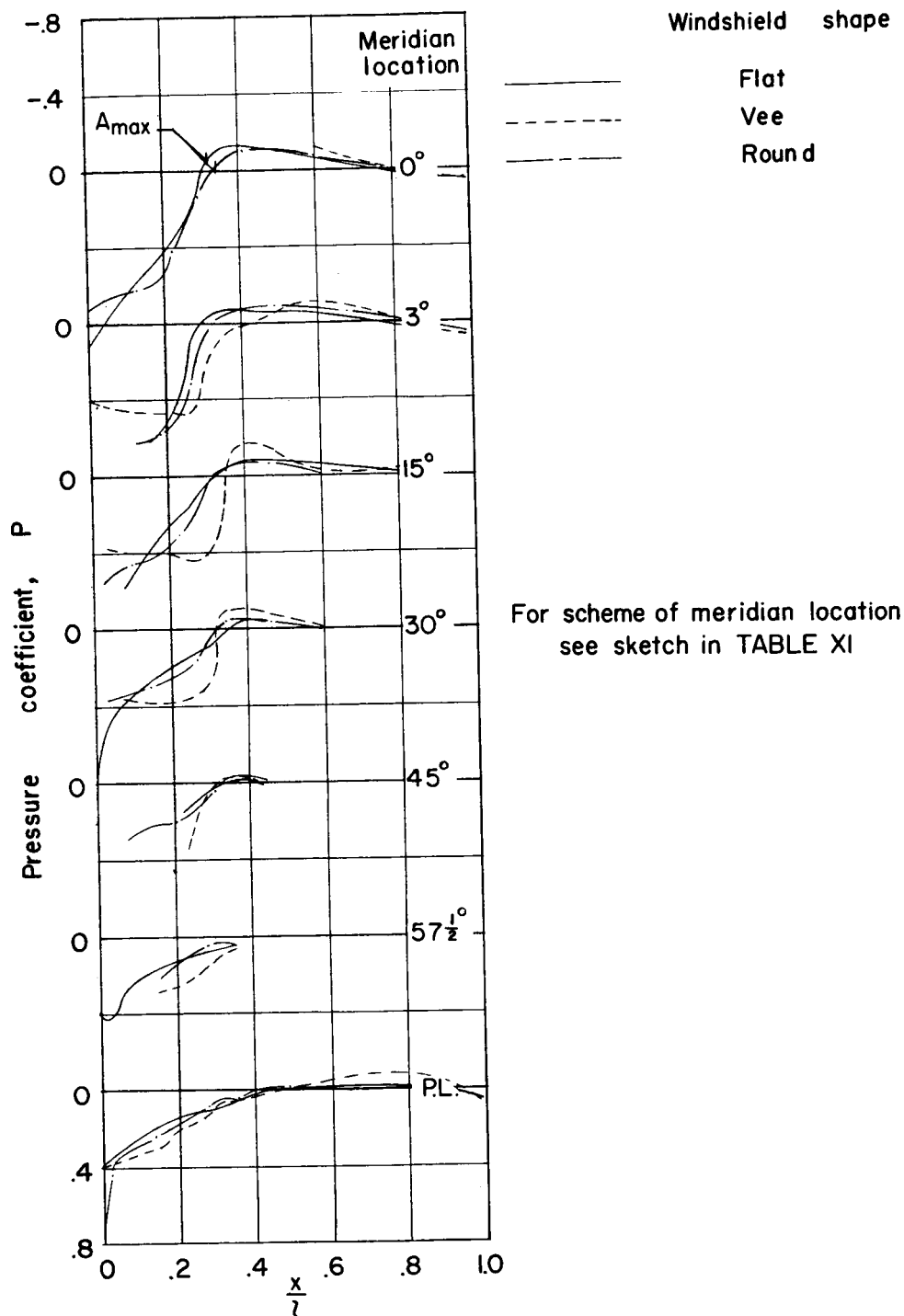
(e) $\alpha = 6.5^\circ$; $\beta = -4^\circ$.

Figure 9.- Continued.



(f) $\alpha = 6.5^\circ$; $\beta = -8^\circ$.

Figure 9.- Concluded.



(a) $\alpha = 0.4^\circ$; $\beta = 0^\circ$.

Figure 10.- Effect of windshield shape on pressure-coefficient distributions on large forward-located canopies at $M = 2.01$.

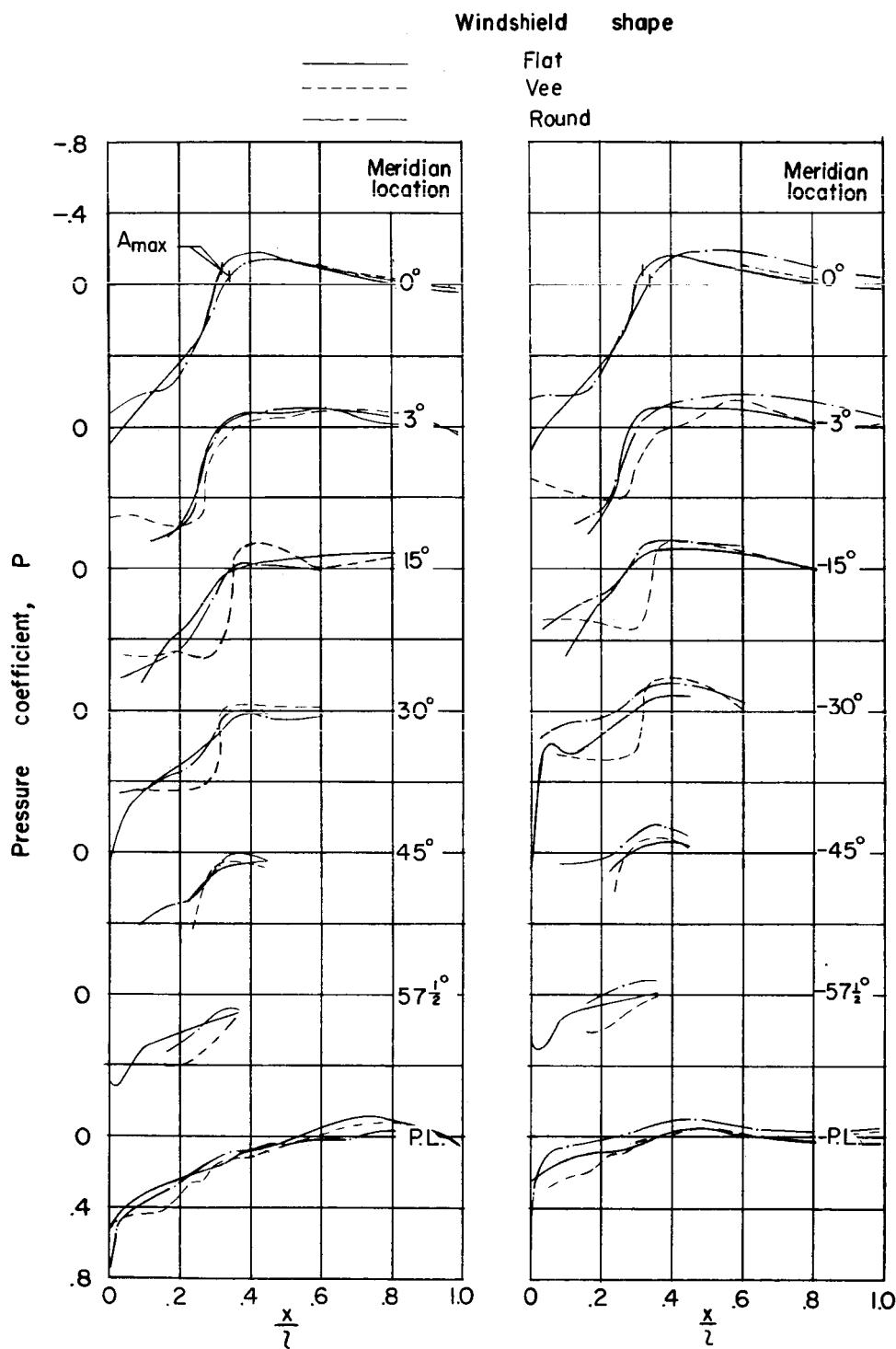
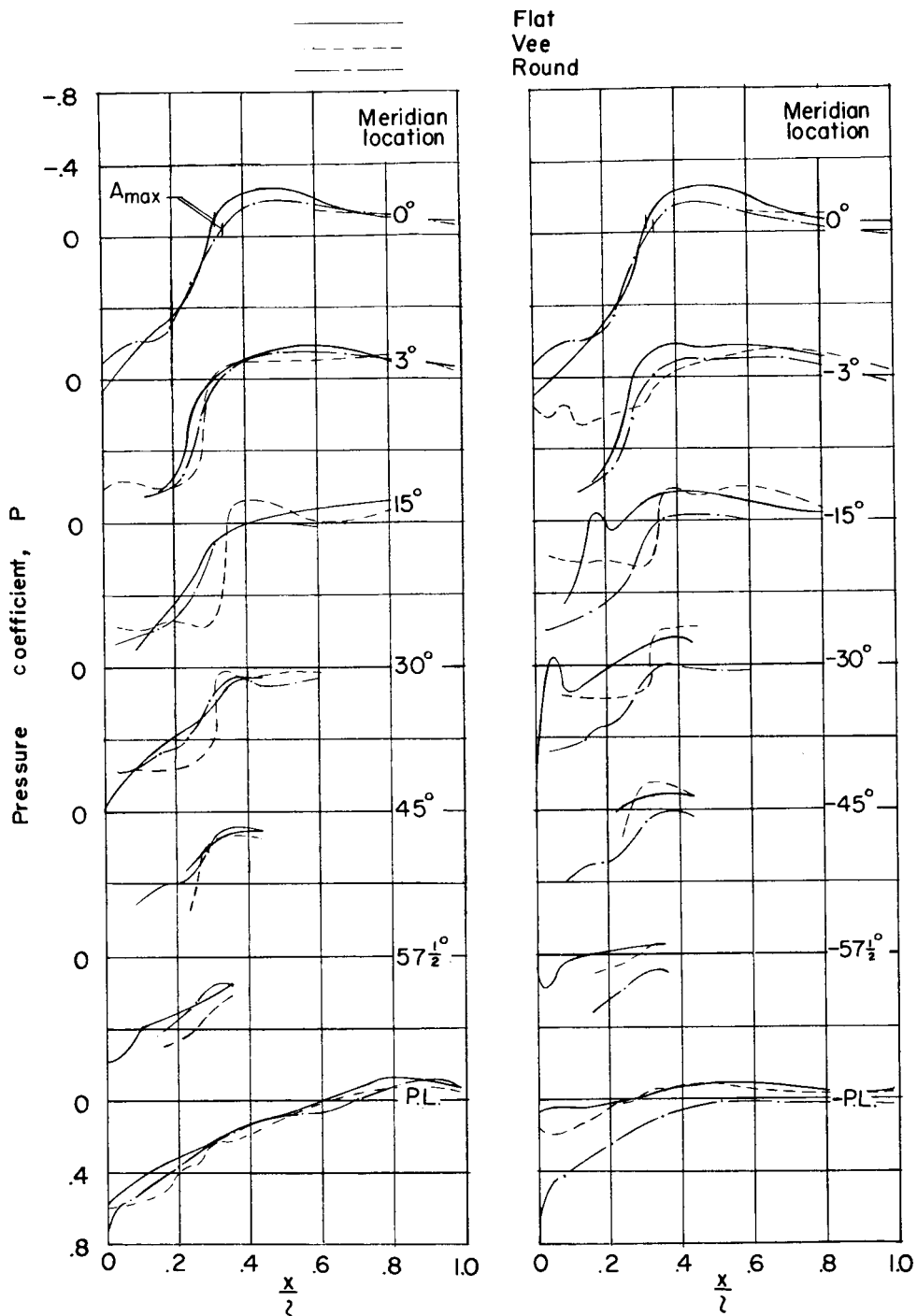


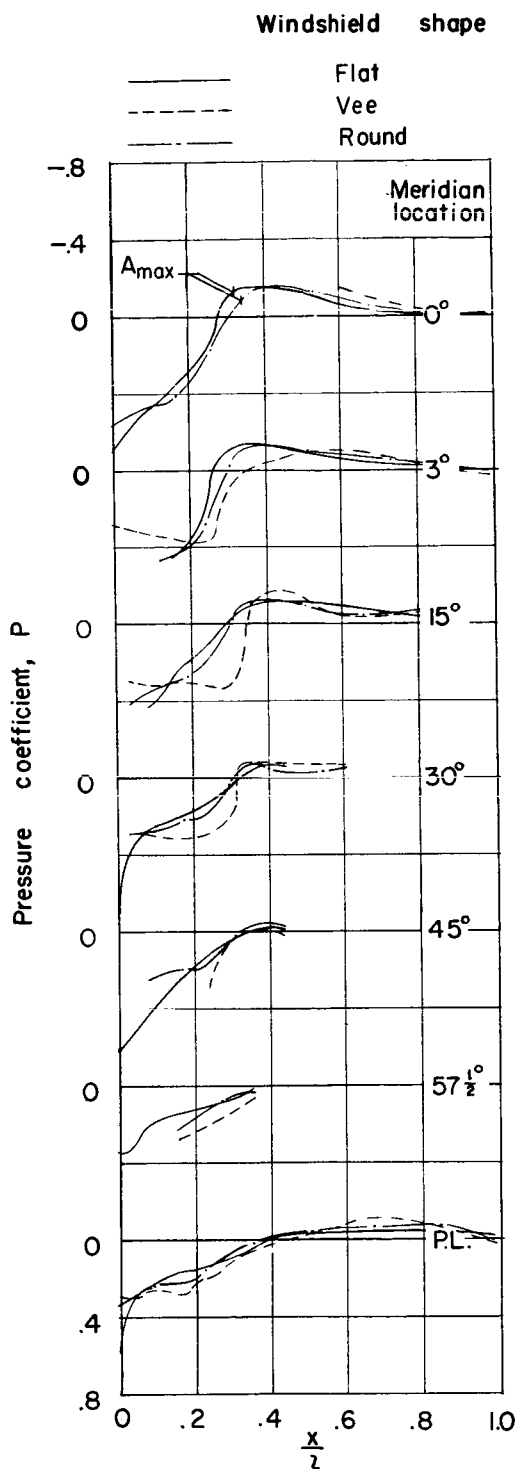
Figure 10.- Continued.

Windshield shape



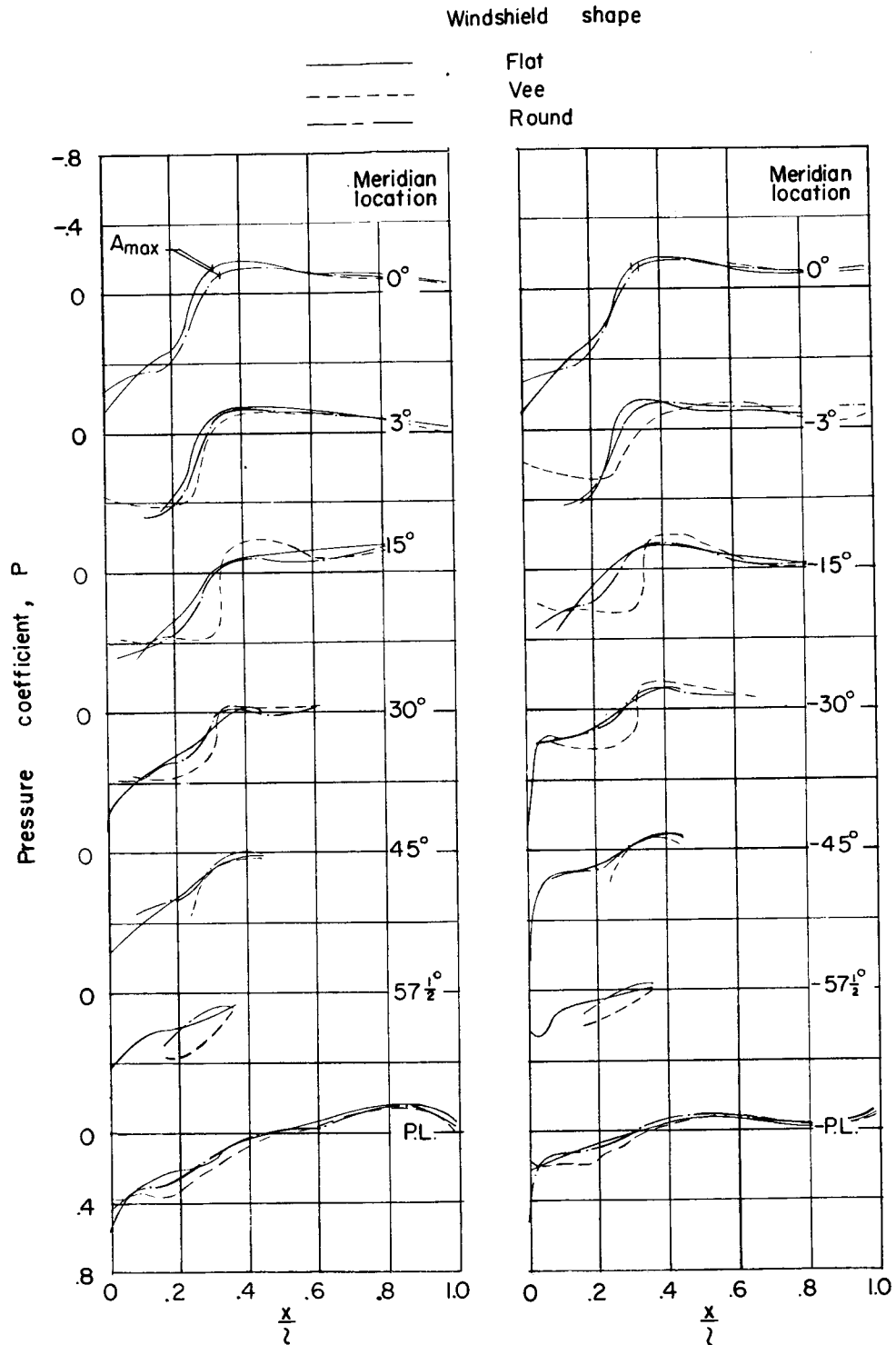
(c) $\alpha = 0.4^\circ$; $\beta = -8^\circ$.

Figure 10.- Continued.



(d) $\alpha = 6.5^\circ$; $\beta = 0^\circ$.

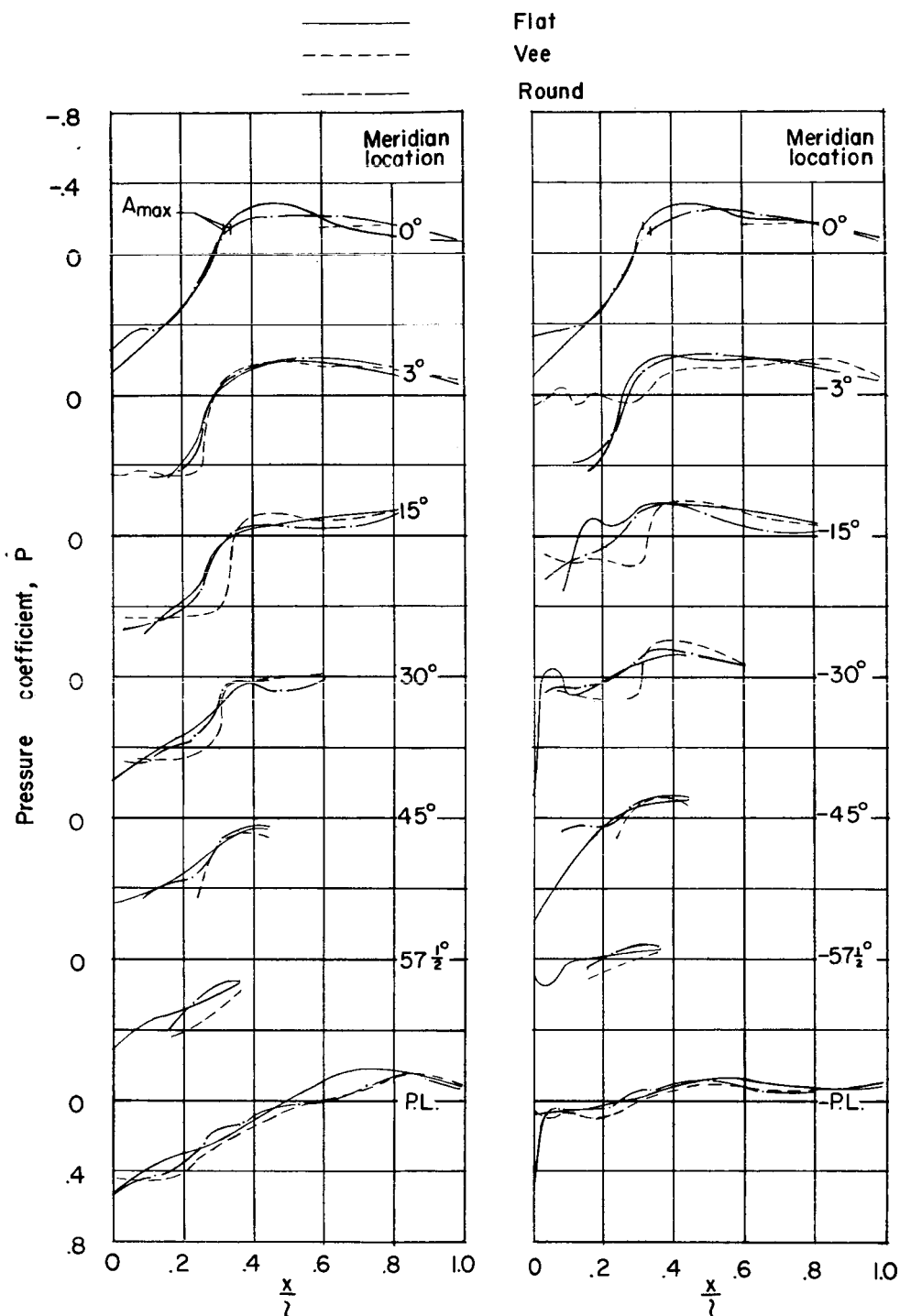
Figure 10.- Continued.



(e) $\alpha = 6.5^\circ$; $\beta = -4^\circ$.

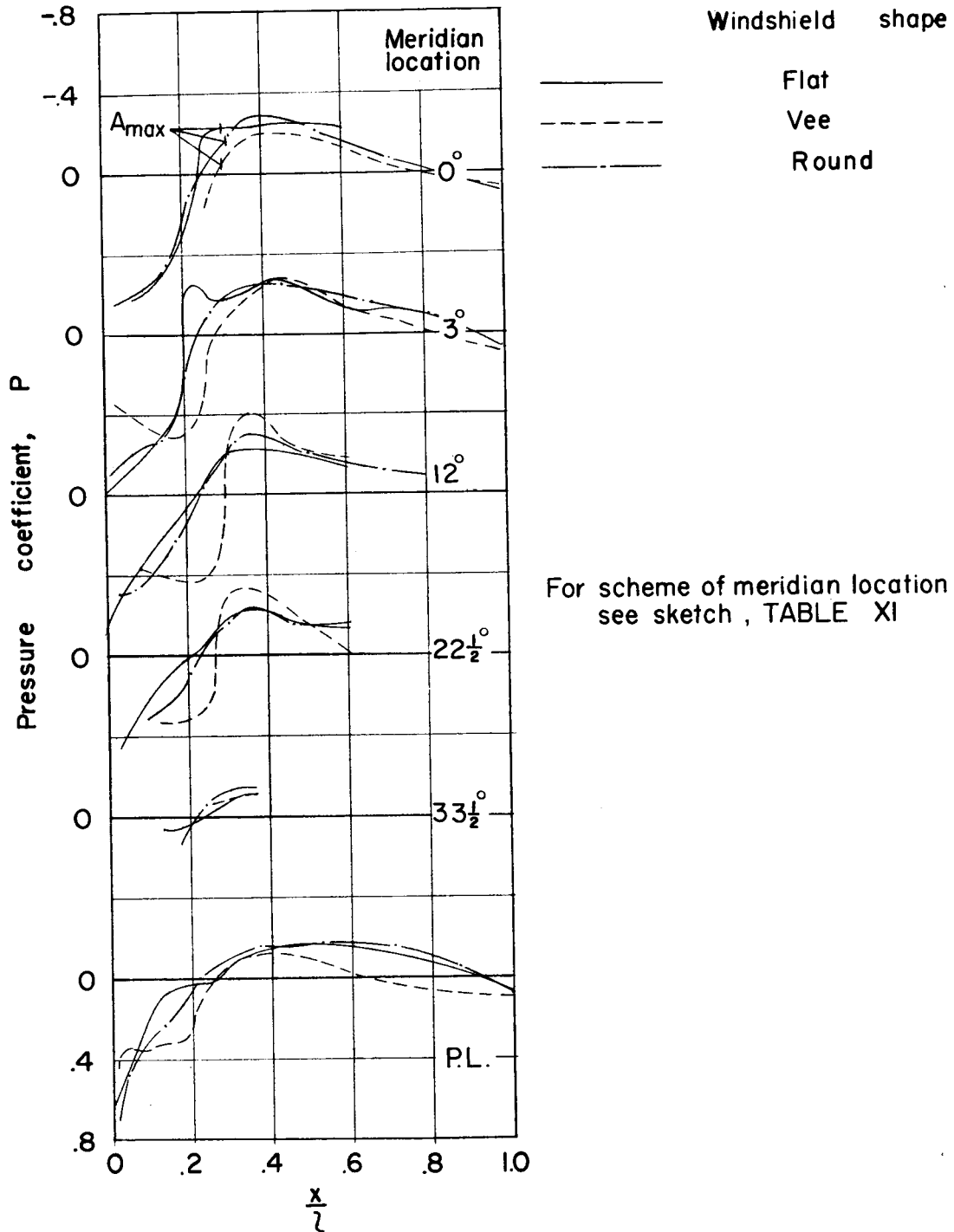
Figure 10.- Continued.

Windshield shape



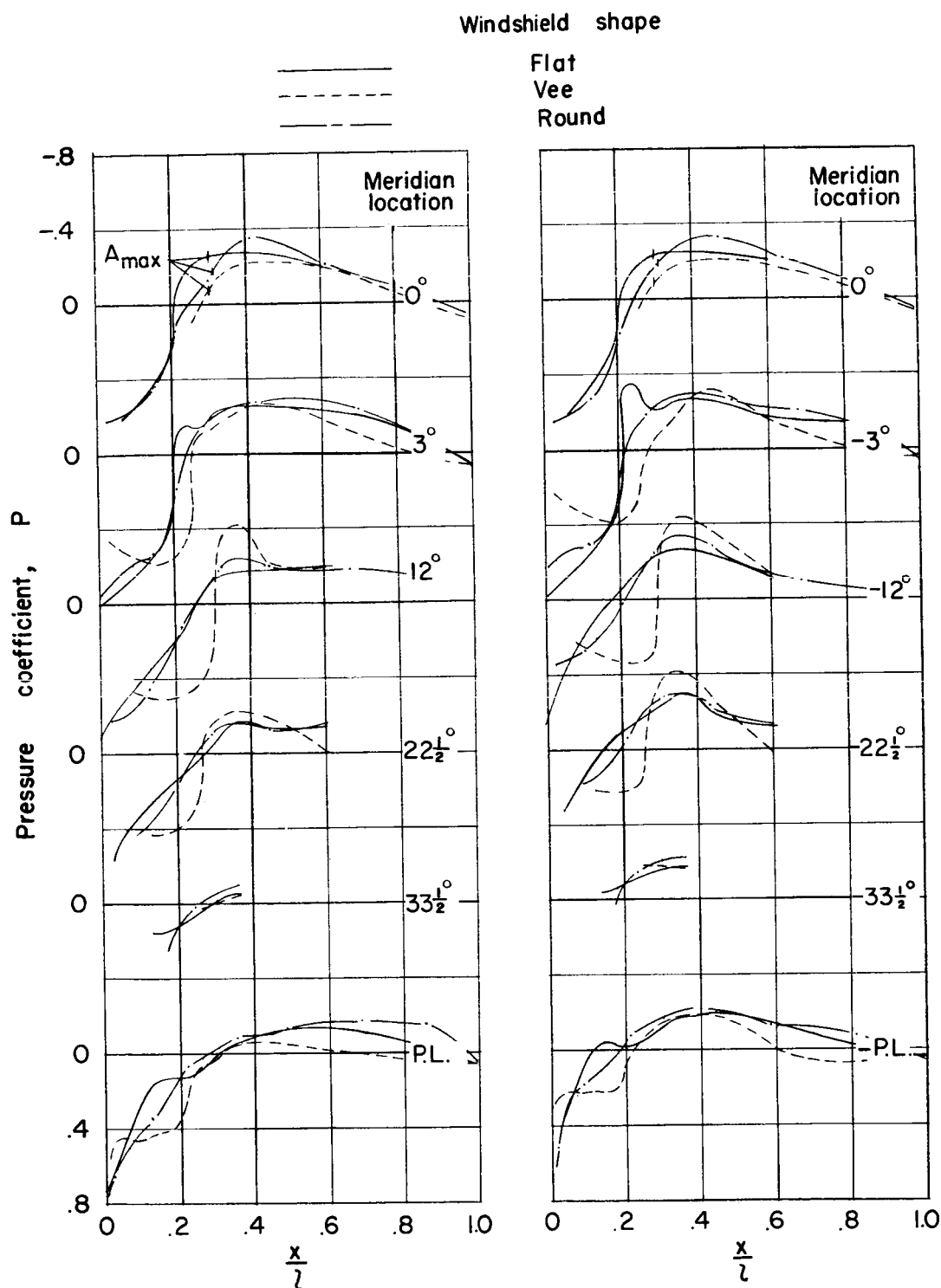
(f) $\alpha = 6.5^\circ$; $\beta = -8^\circ$.

Figure 10.- Concluded.



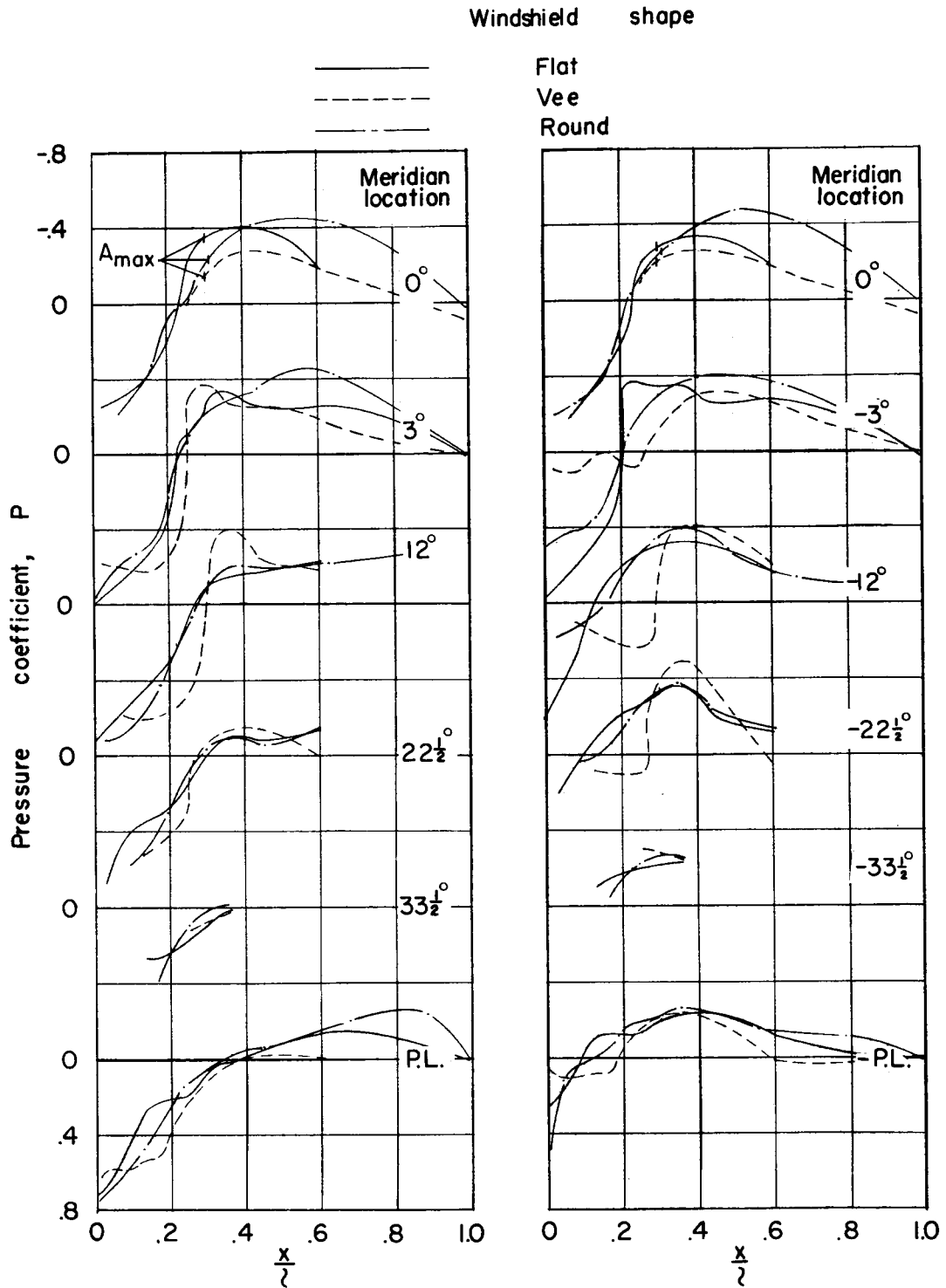
(a) $\alpha = 0.4^\circ$; $\beta = 0^\circ$.

Figure 11.- Effect of windshield shape on pressure-coefficient distributions on large rearward-located canopies at $M = 1.41$.



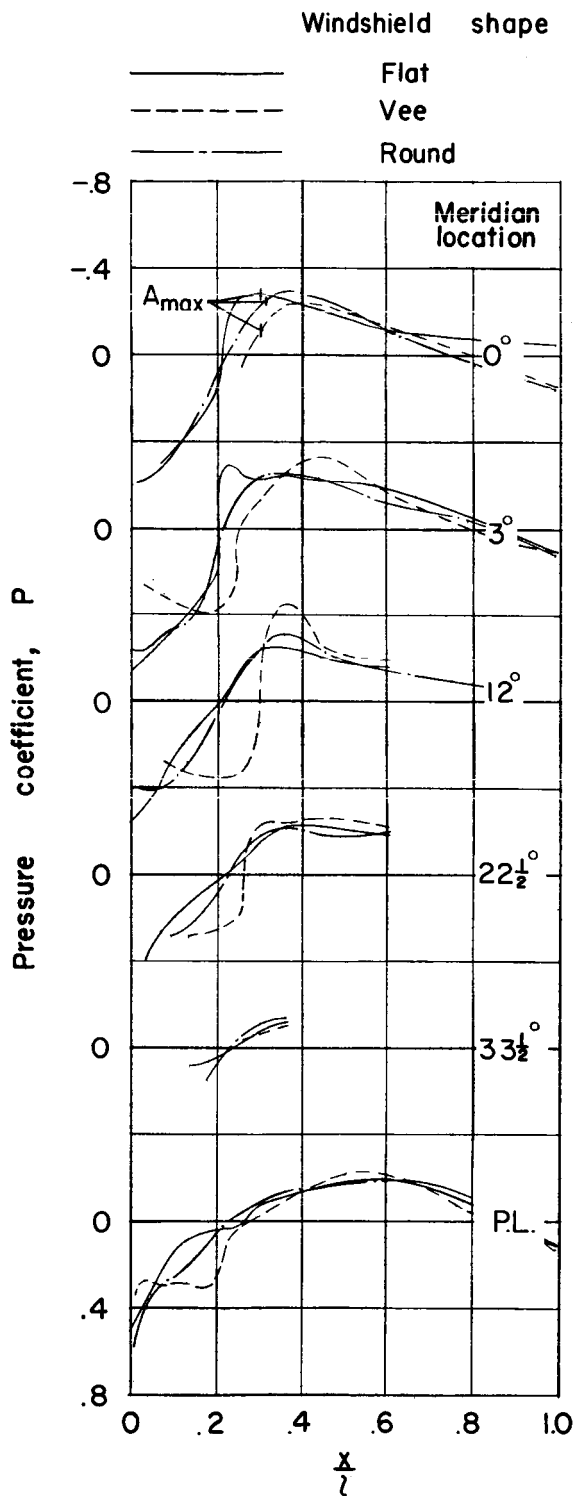
(b) $\alpha = 0.4^\circ$; $\beta = -4^\circ$.

Figure 11.- Continued.



(c) $\alpha = 0.4^\circ$; $\beta = -8^\circ$.

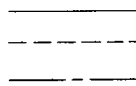
Figure 11.- Continued.



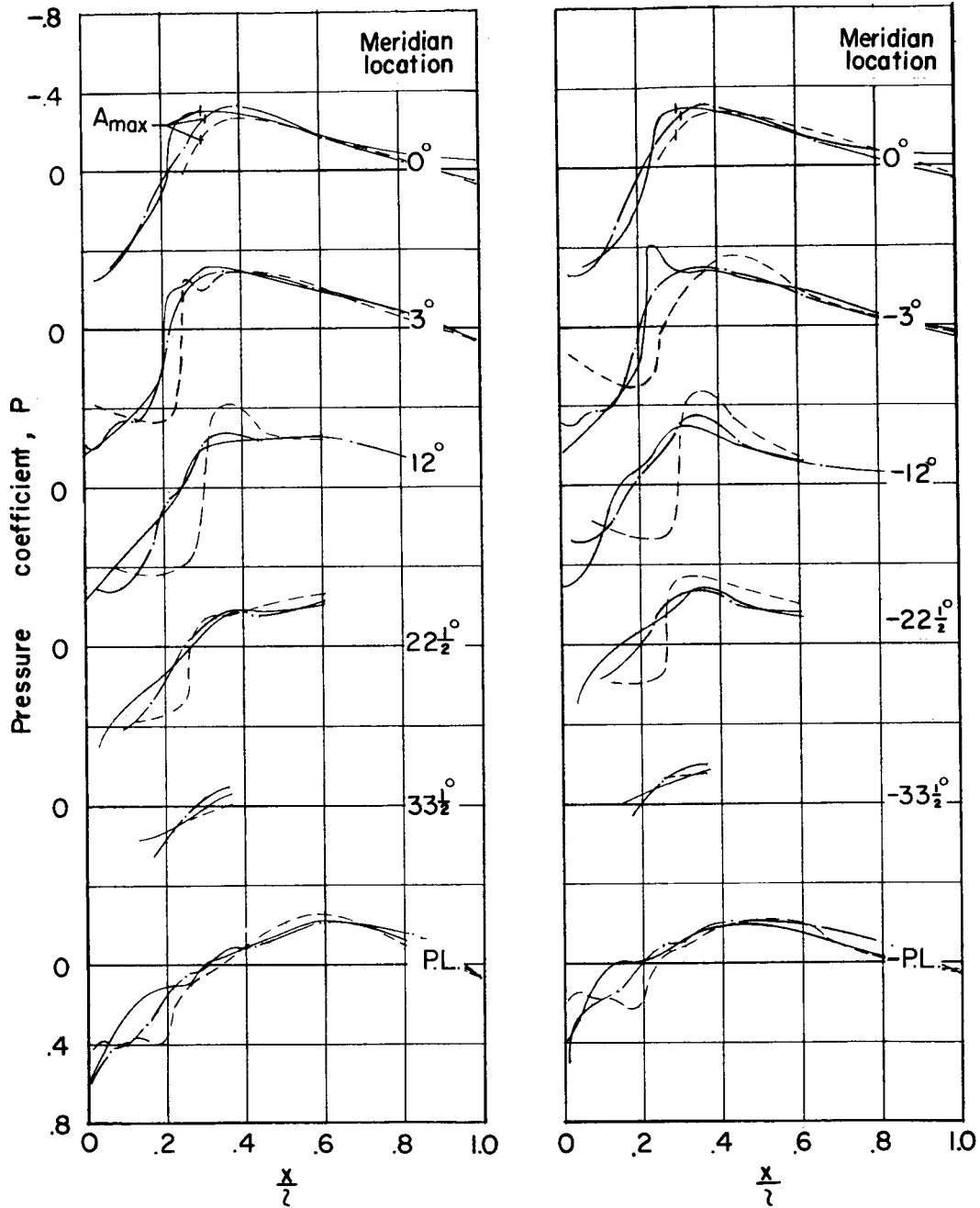
(d) $\alpha = 6.5^\circ$; $\beta = 0^\circ$.

Figure 11.- Continued.

Windshield shape

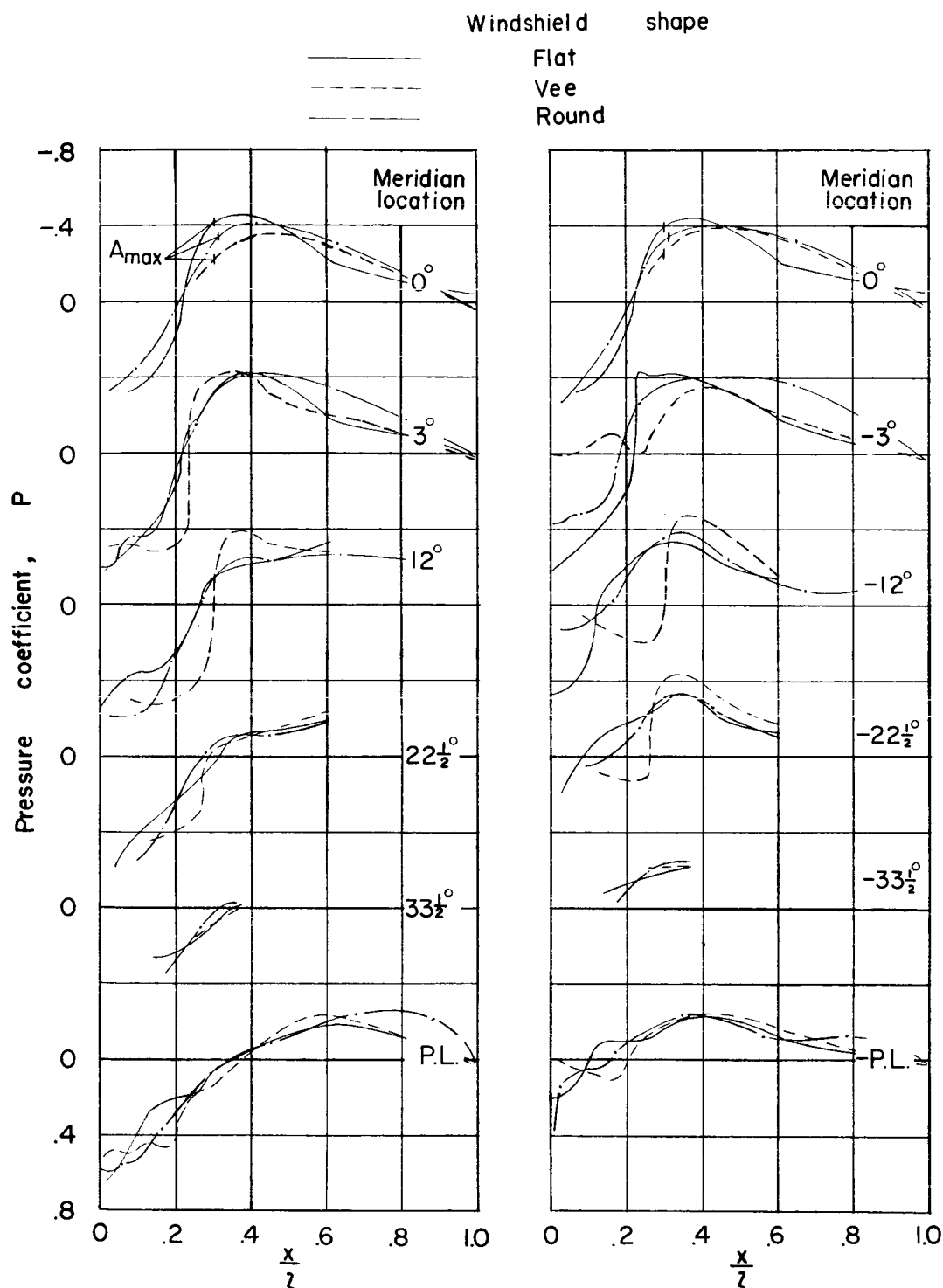


Flat
Vee
Round



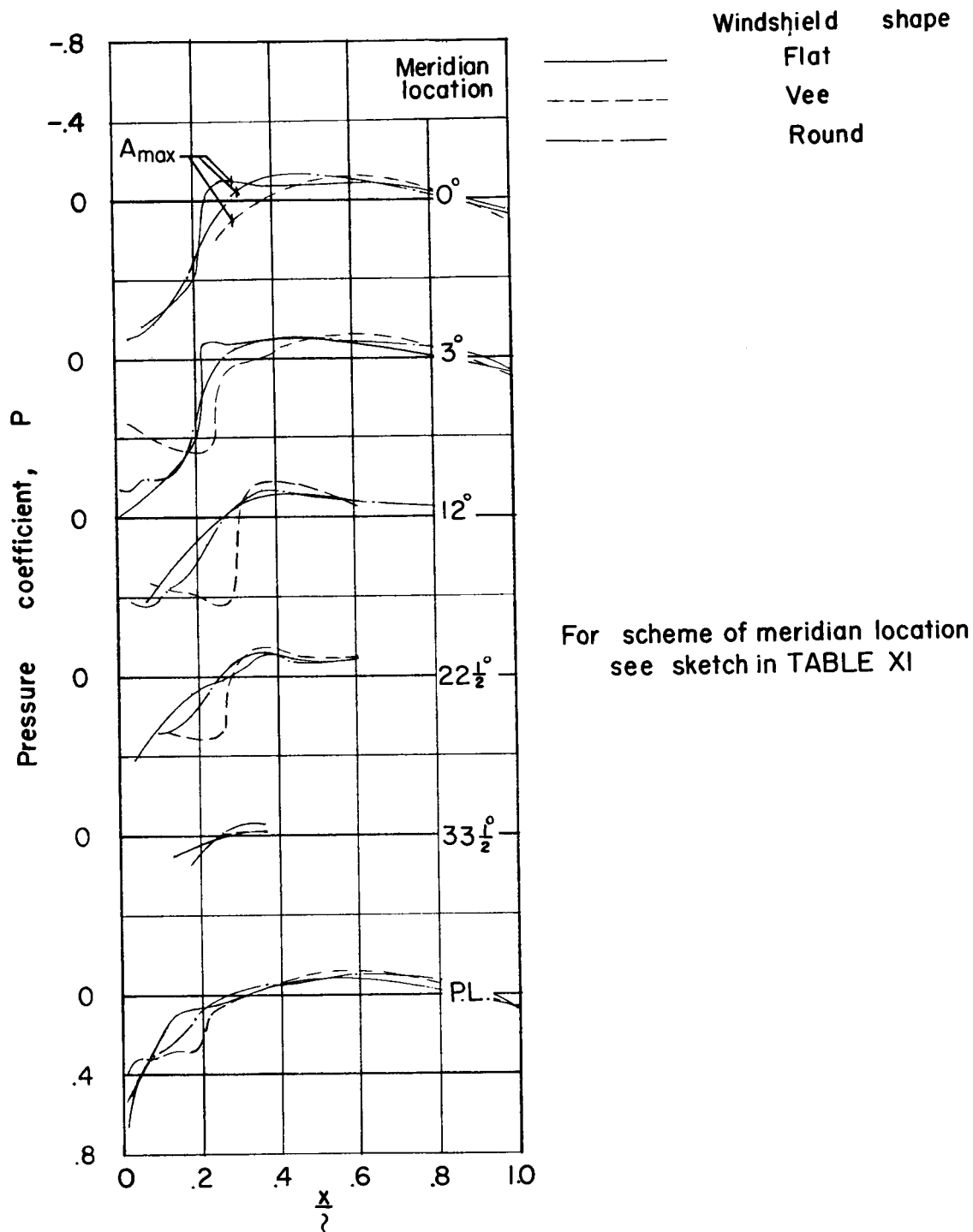
(e) $\alpha = 6.5^\circ$; $\beta = -4^\circ$.

Figure 11.- Continued.



(f) $\alpha = 6.5^\circ$; $\beta = -8^\circ$.

Figure 11.- Concluded.



(a) $\alpha = 0.4^\circ$; $\beta = 0^\circ$.

Figure 12.- Effect of windshield shape on pressure-coefficient distributions on large rearward-located canopies at $M = 2.01$.

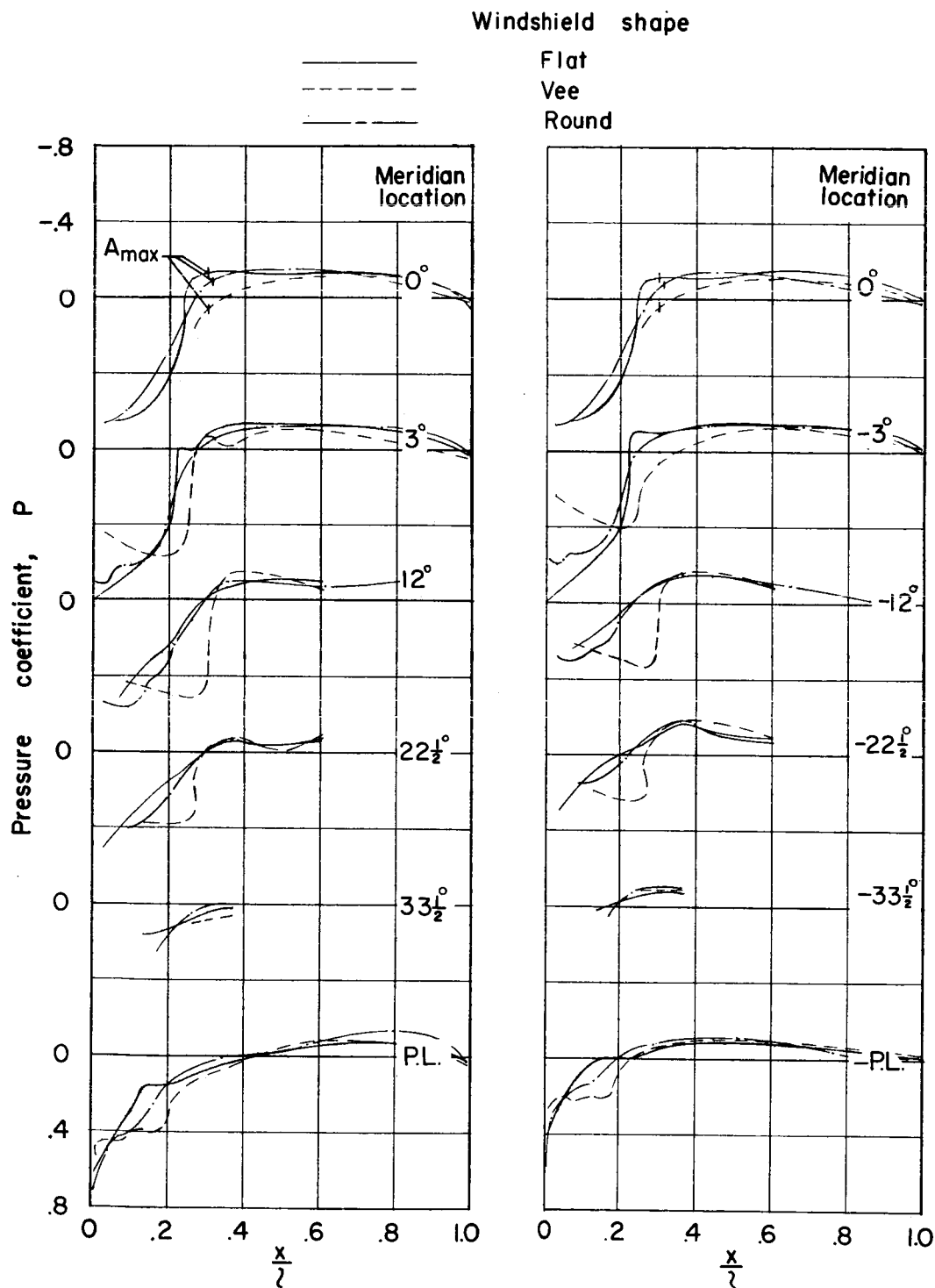
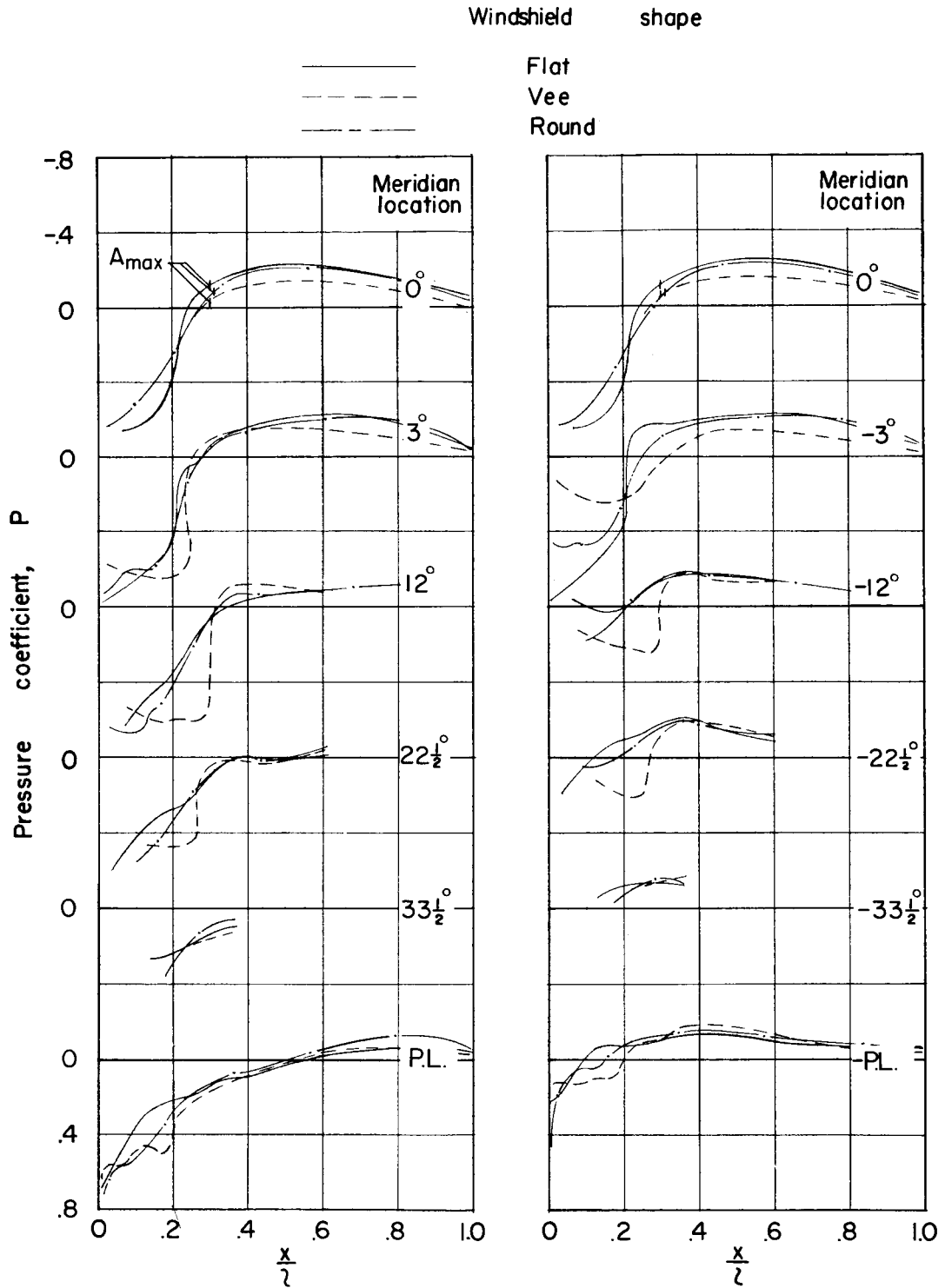


Figure 12.- Continued.

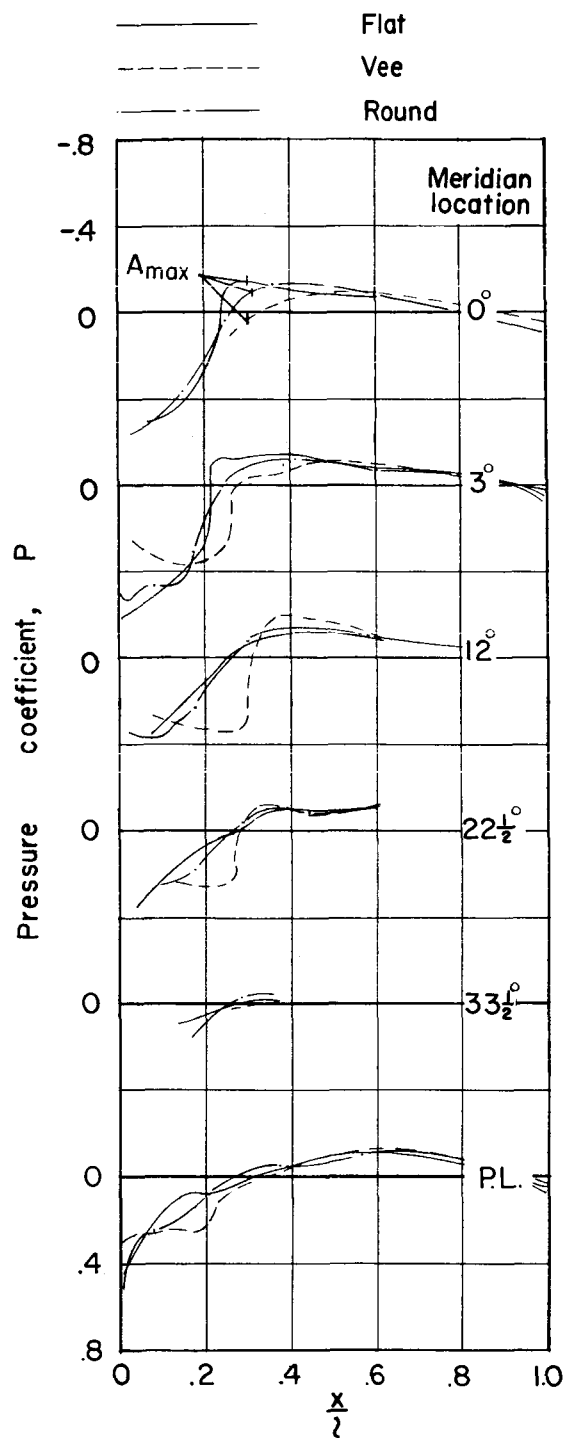


(c) $\alpha = 0.4^\circ$; $\beta = -8^\circ$.

Figure 12.- Continued.

UNCLASSIFIED

Windshield shape



(d) $\alpha = 6.5^\circ$; $\beta = 0^\circ$.

Figure 12.- Continued.

CONFIDENTIAL

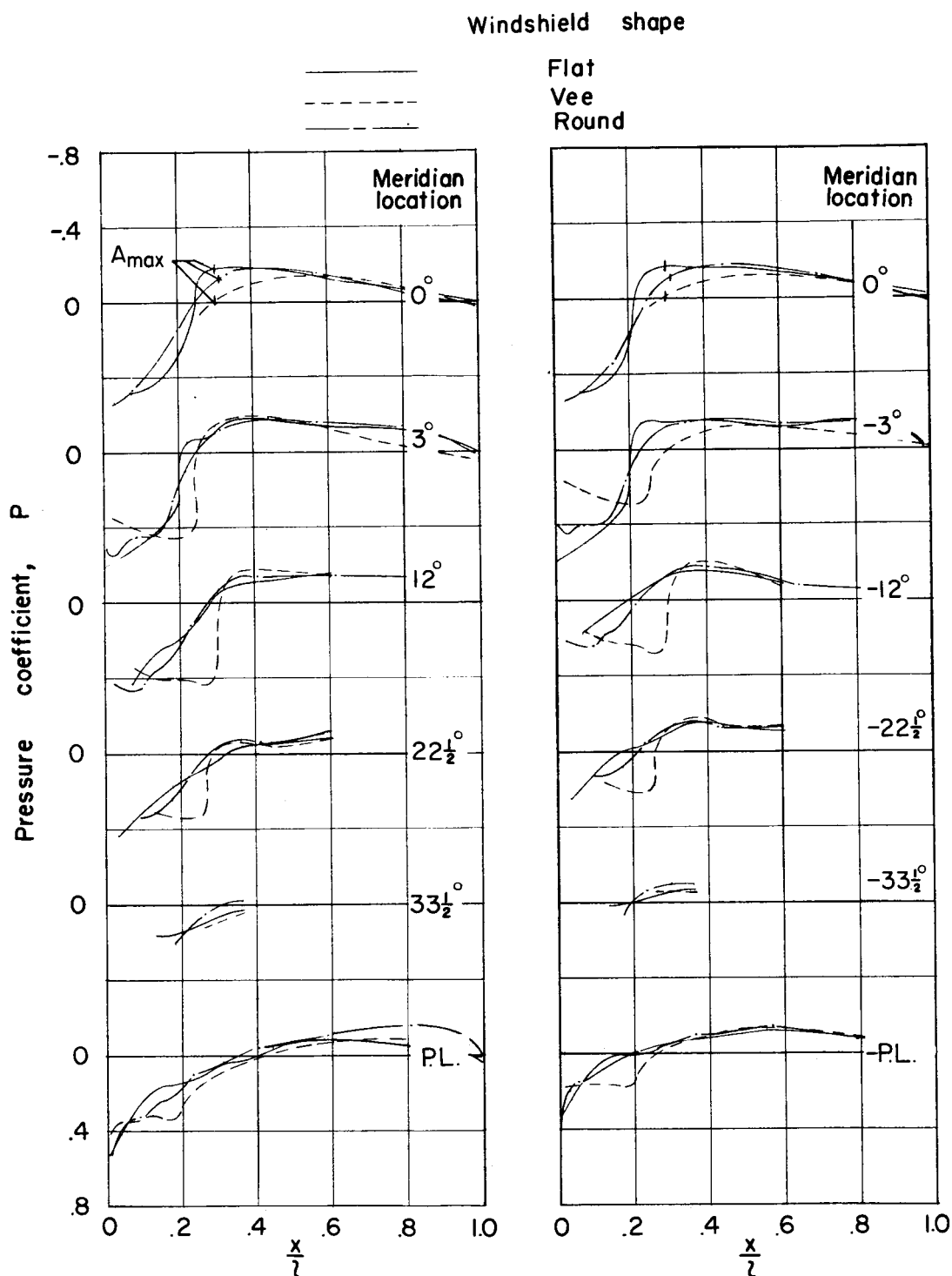
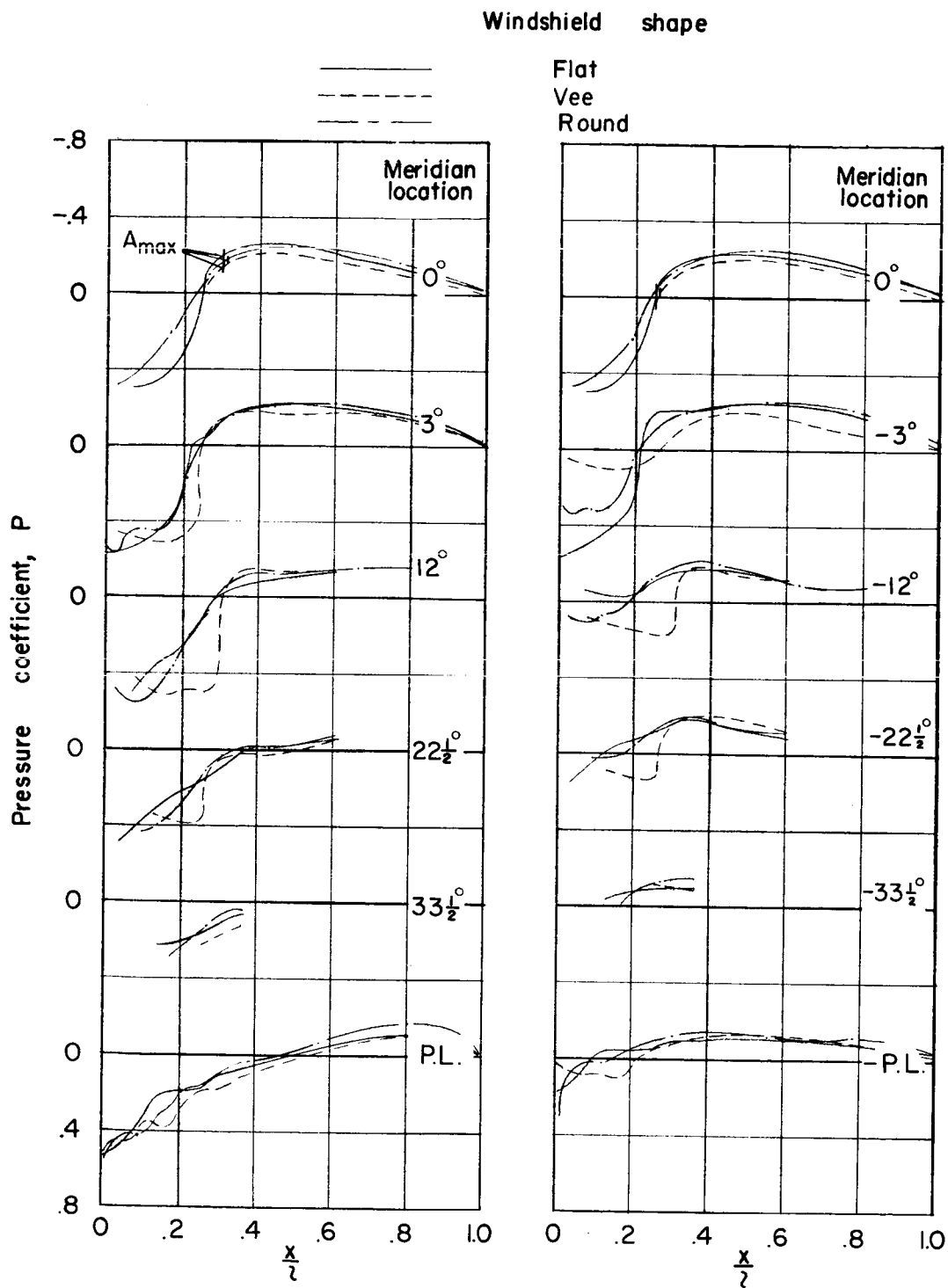
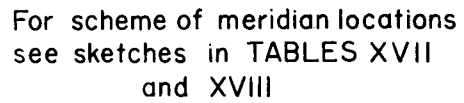


Figure 12.- Continued.



(f) $\alpha = 6.5^\circ$; $\beta = -8^\circ$.

Figure 12.- Concluded.



(a) $\alpha = 0.4^\circ$; $\beta = 0^\circ$.

Figure 13.- Pressure distributions on small canopies at $M = 1.41$ and 2.01 for various angles of attack and sideslip.

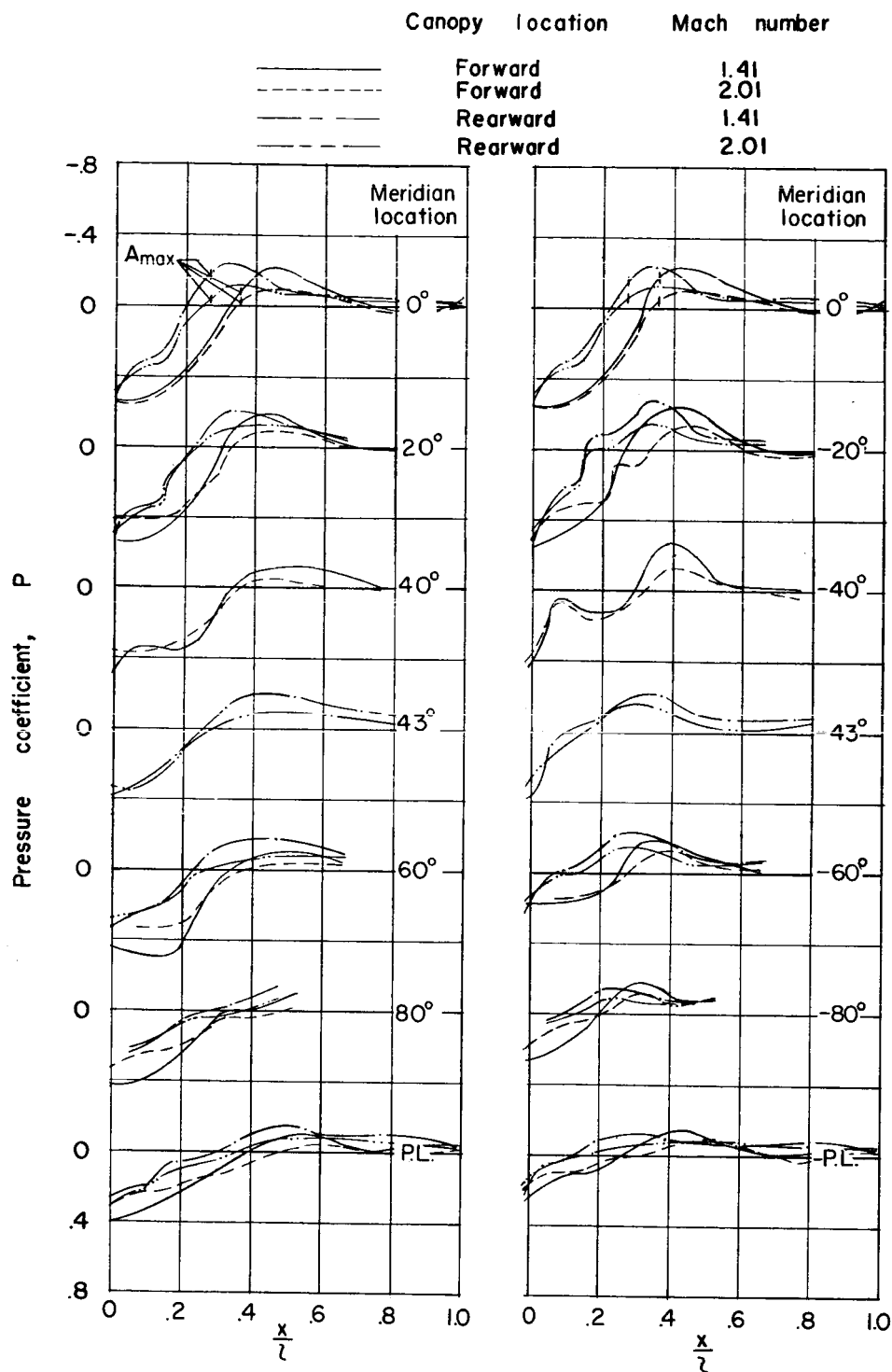


Figure 13.- Continued.

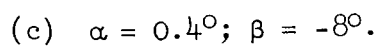


Figure 13.- Continued.

CONFIDENTIAL

97

UNCLASSIFIED

Canopy location Mach number

—	Forward	1.41
- - -	Forward	2.01
—	Rearward	1.41
- - -	Rearward	2.01

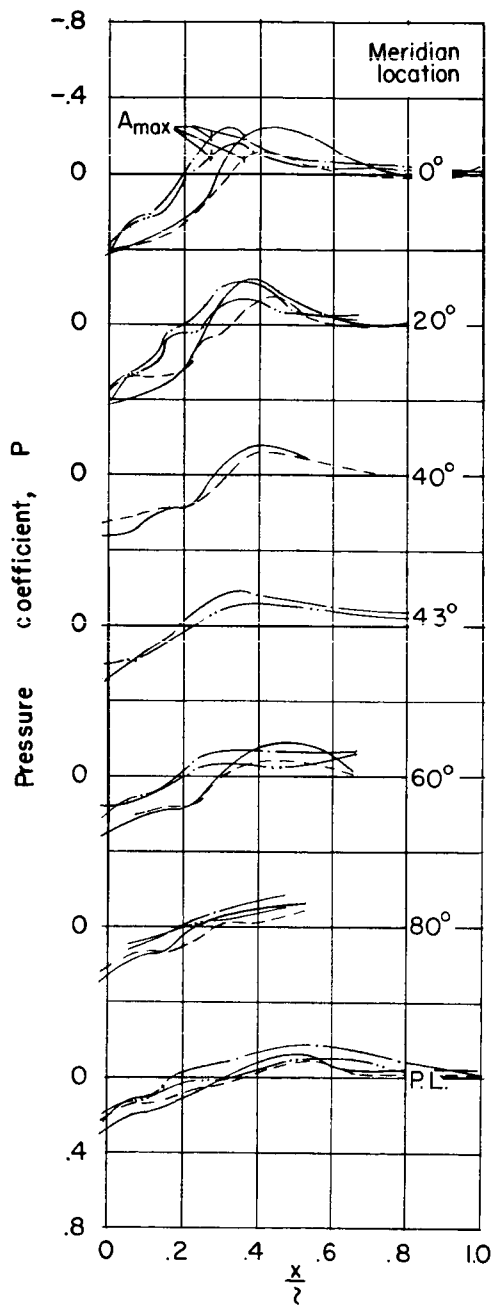
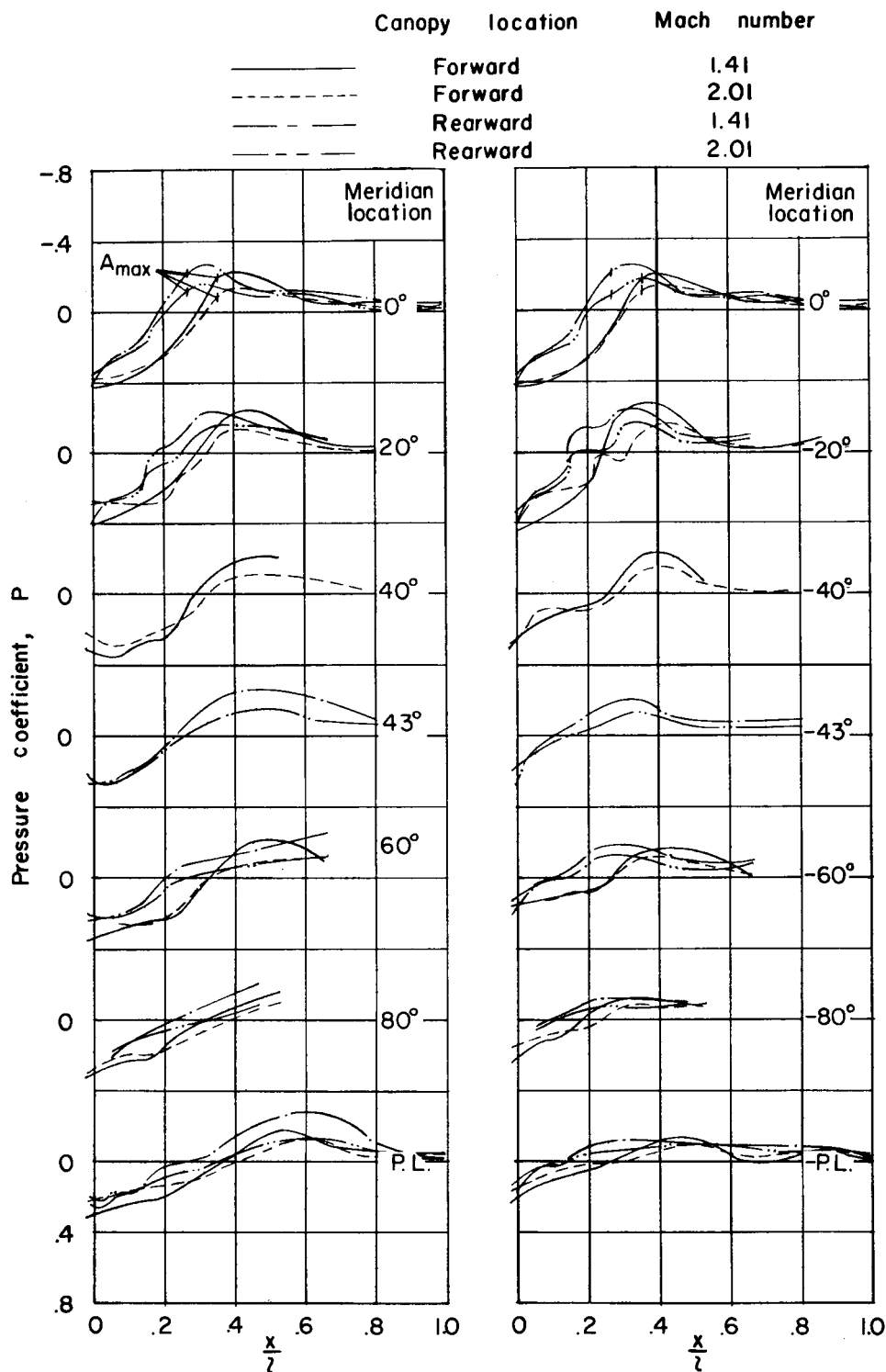
(d) $\alpha = 6.5^\circ$; $\beta = 0^\circ$.

Figure 13.- Continued.

CONFIDENTIAL



(e) $\alpha = 6.5^\circ$; $\beta = -4^\circ$.

Figure 13.- Continued.

UNCLASSIFIED

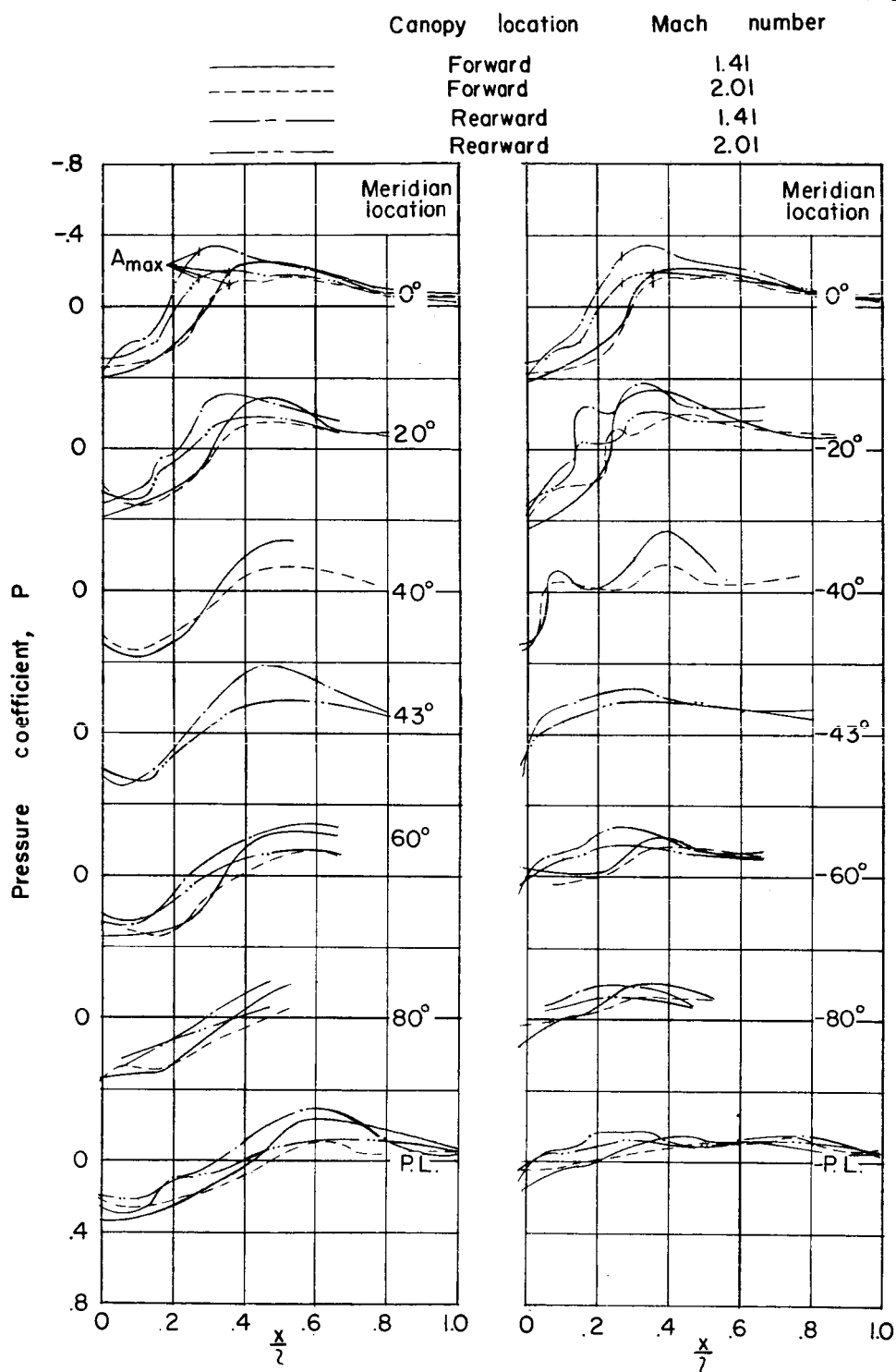
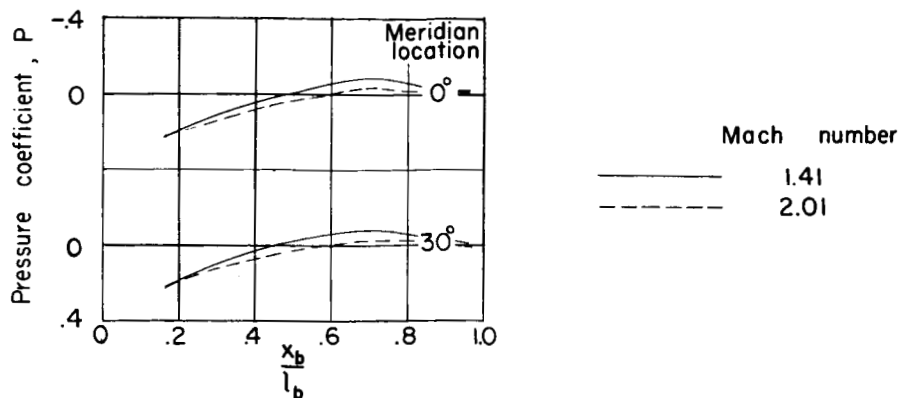
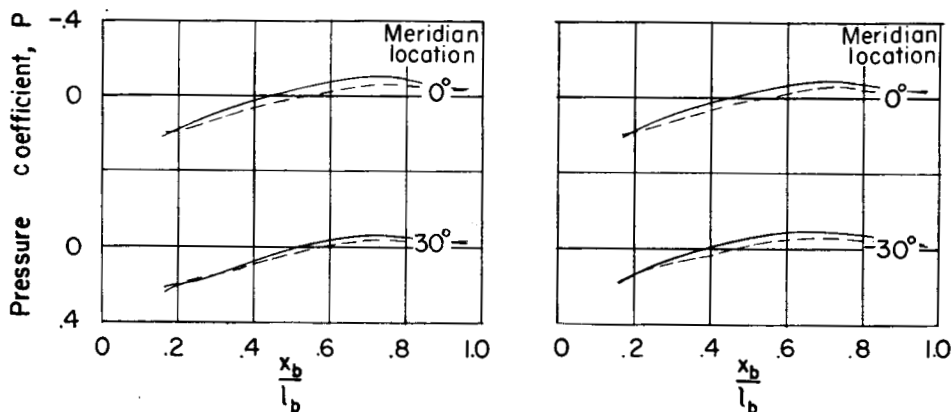
(f) $\alpha = 6.5^\circ$; $\beta = -8^\circ$.

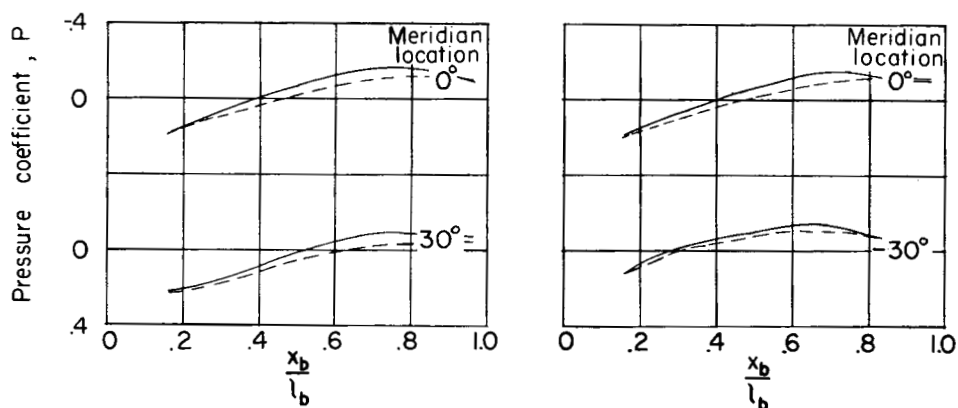
Figure 13.- Concluded.



(a) $\alpha = 0.4^\circ$; $\beta = 0^\circ$



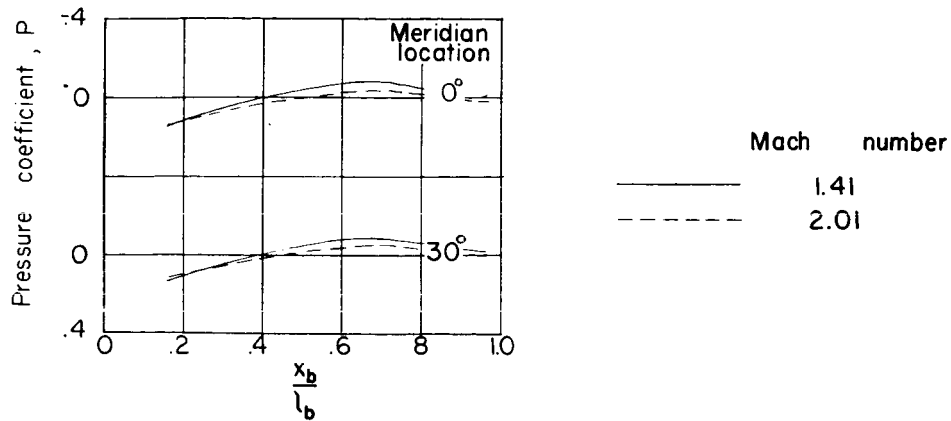
(b) $\alpha = 0.4^\circ$; $\beta = -4^\circ$



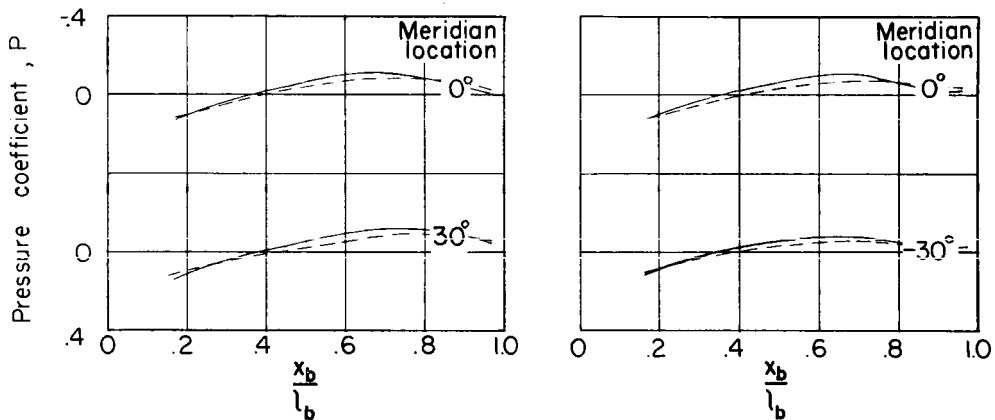
(c) $\alpha = 0.4^\circ$; $\beta = -8^\circ$

Figure 14.- Pressure distributions on body alone at $M = 1.41$ and 2.01 for various angles of attack and sideslip.

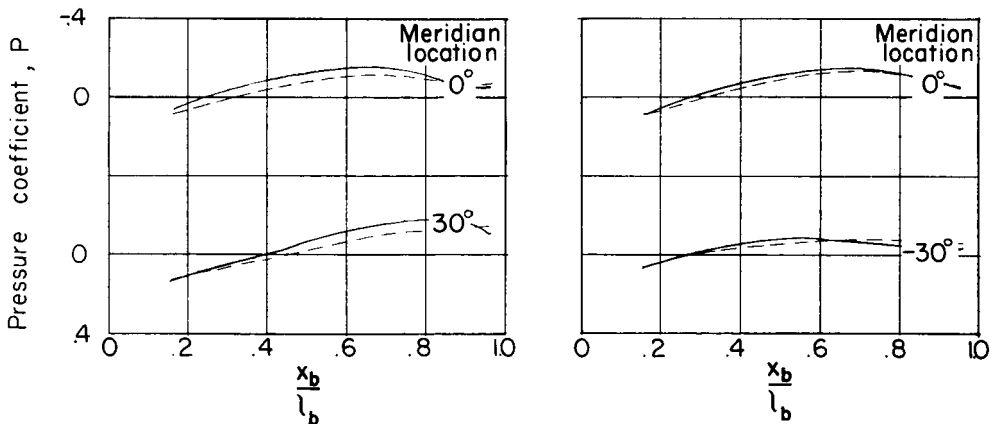
UNCLASSIFIED



(d) $\alpha = 6.5^\circ$; $\beta = 0^\circ$.



(e) $\alpha = 6.5^\circ$; $\beta = -4^\circ$.



(f) $\alpha = 6.5^\circ$; $\beta = -8^\circ$.

Figure 14.- Concluded.

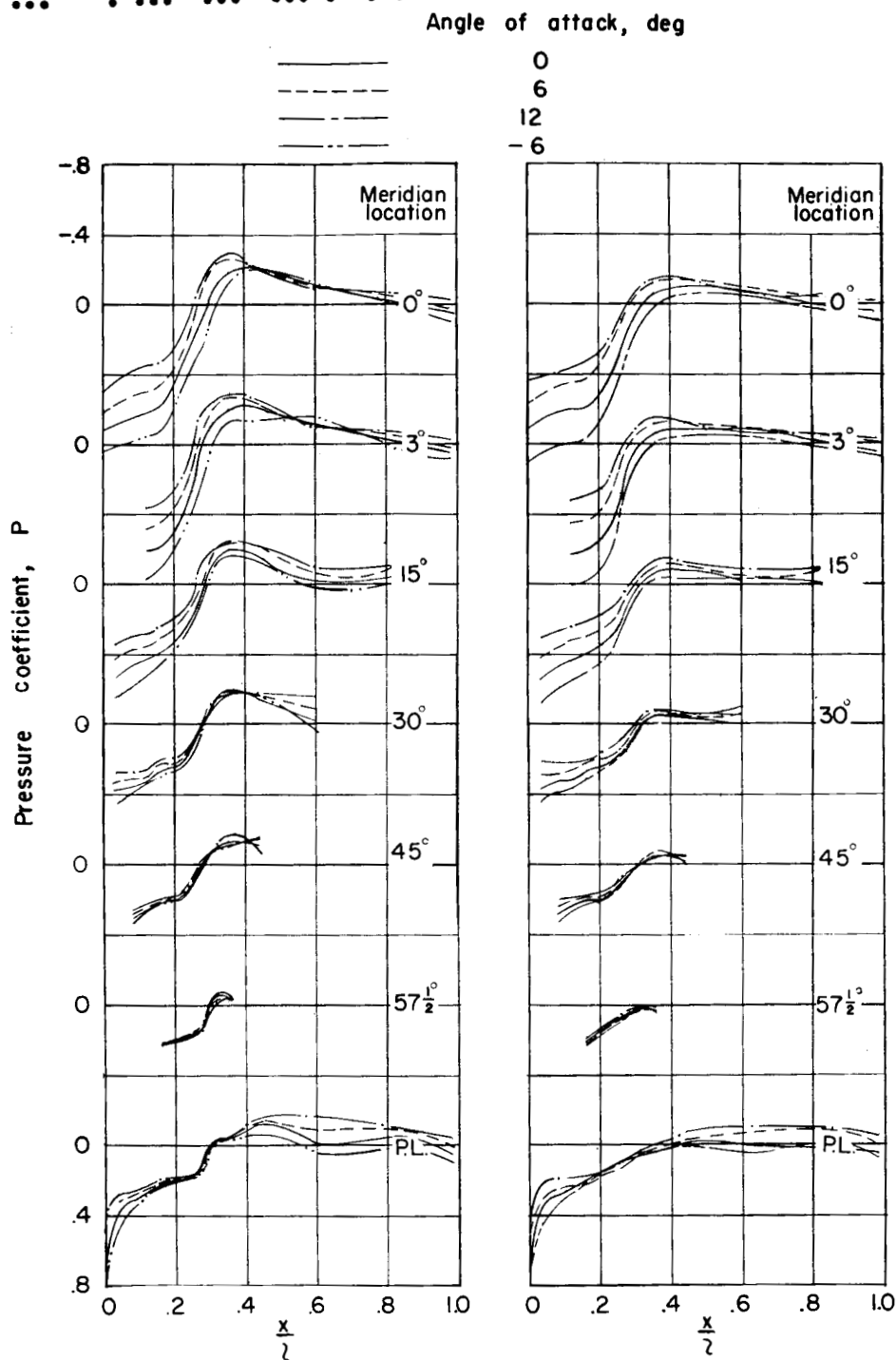
(a) $M = 1.41$.(b) $M = 2.01$.

Figure 15.- Pressure distribution on round-windshield canopy in forward location at $M = 1.41$ and 2.01 for various angles of attack and 0.3° sideslip.

CONFIDENTIAL

UNCLASSIFIED

103

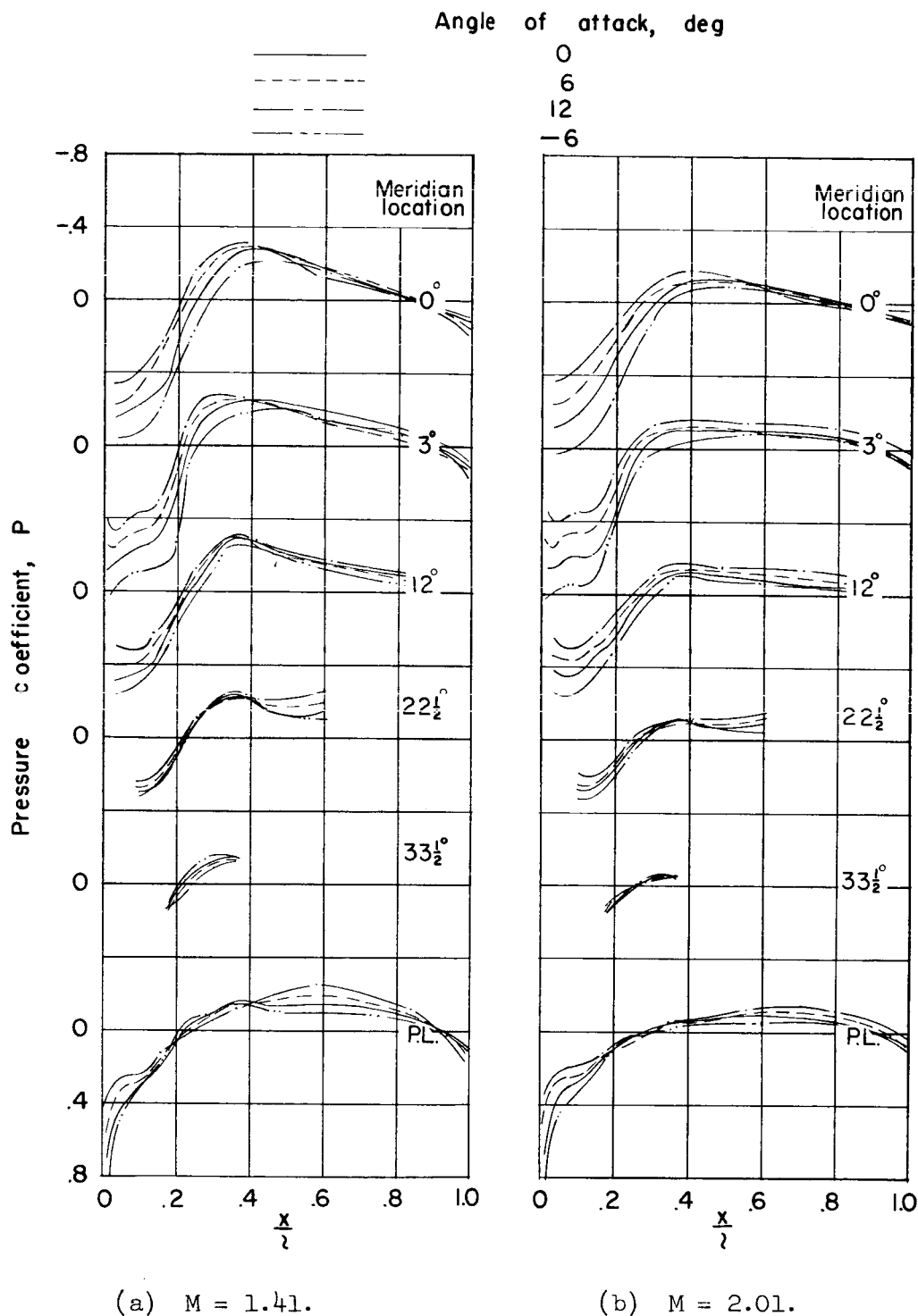
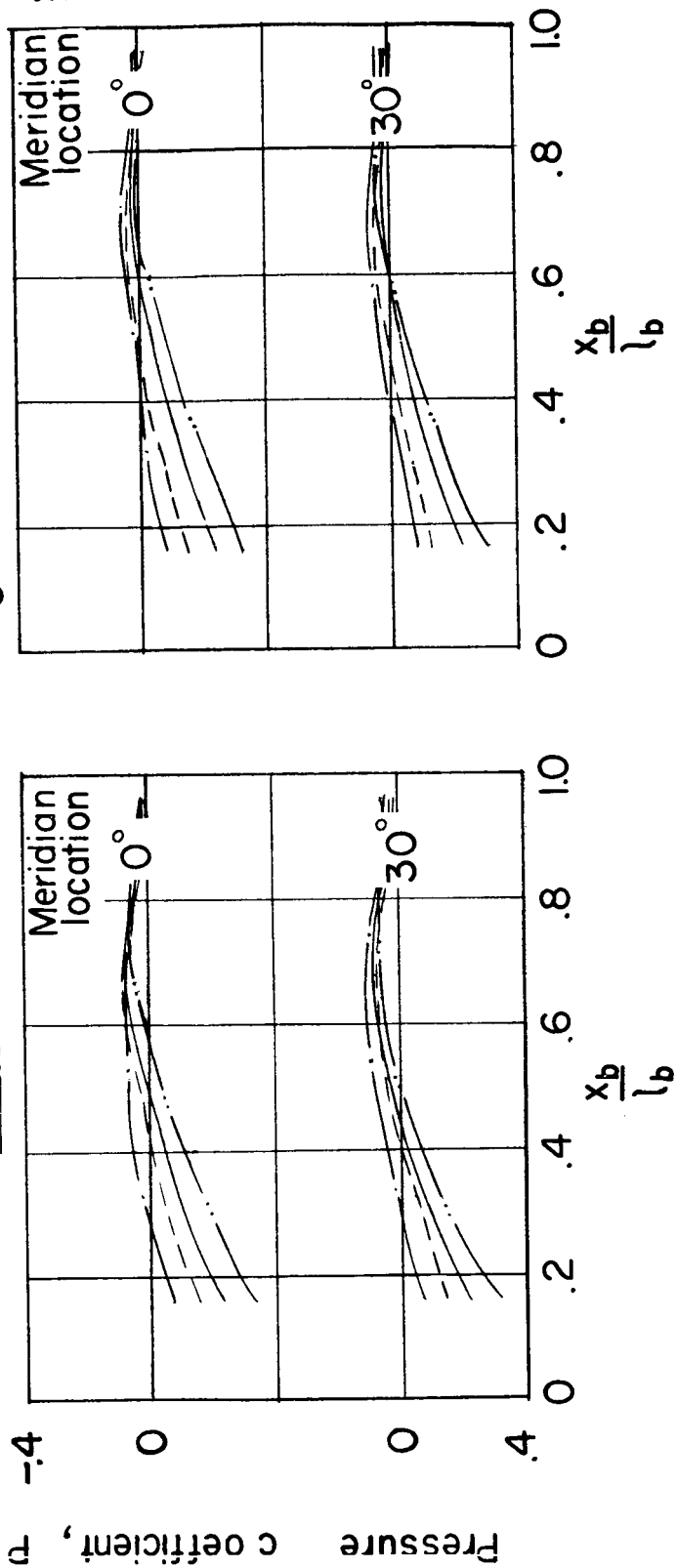


Figure 16.- Pressure distribution on round-windshield canopy in rearward location at $M = 1.41$ and 2.01 for various angles of attack and 0.3° sideslip.

CONFIDENTIAL

Angle of attack, deg

0
6
12
-6

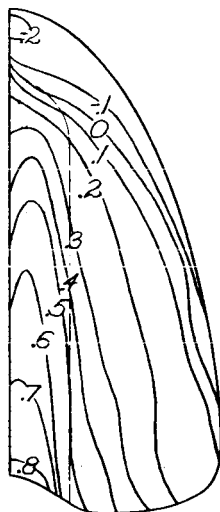


(a) $M = 1.41$.

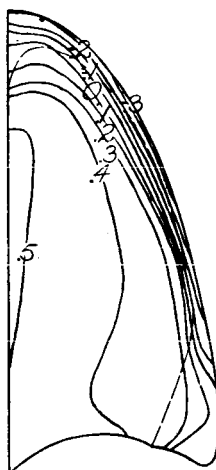
(b) $M = 2.01$.

Figure 17.- Pressure distribution on body alone at $M = 1.41$ and 2.01 for various angles of attack and 0.30 sideslip.

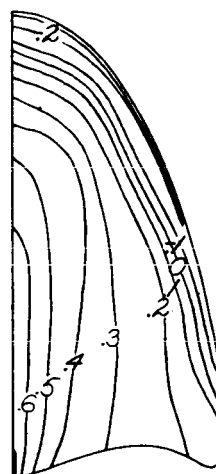
Flat Windshield


 $\alpha = 0.4^\circ$
 $\beta = 0^\circ$

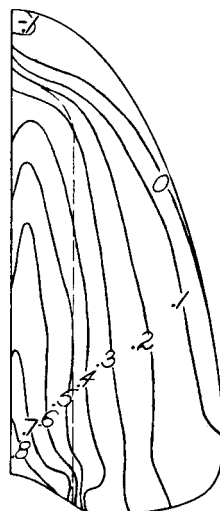
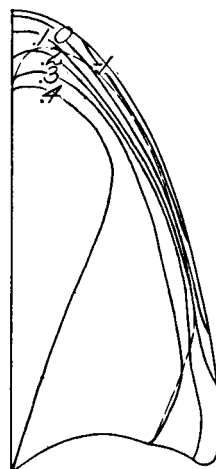
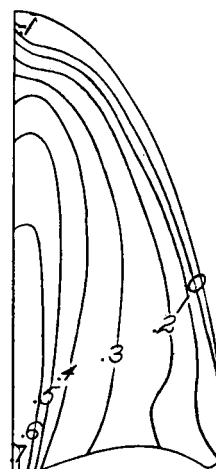
Vee Windshield


 $\alpha = 0.4^\circ$
 $\beta = 0^\circ$

Round Windshield


 $\alpha = 0^\circ$
 $\beta = 0.3^\circ$

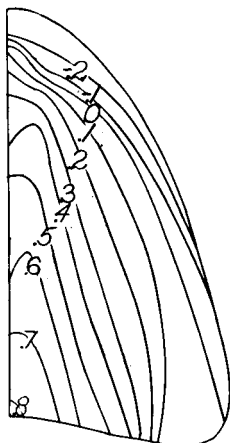
M=1.41


 $\alpha = 0.4^\circ$
 $\beta = 0^\circ$

 $\alpha = 0.4^\circ$
 $\beta = 0^\circ$

 $\alpha = 0.4^\circ$
 $\beta = 0^\circ$

M=2.01

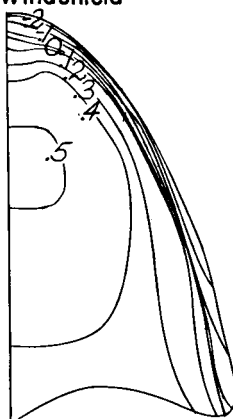
Figure 18.- Pressure coefficient contours on one-half the frontal projections of each of the large forward-located canopies for $M = 1.41$ and 2.01 .

Flat Windshield



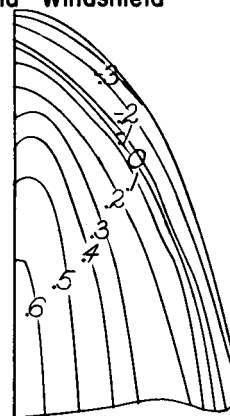
$\alpha = 0.4^\circ$
 $\beta = 0^\circ$

Vee Windshield



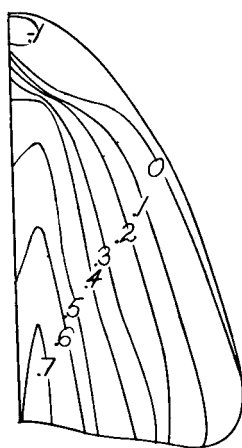
$\alpha = 0.4^\circ$
 $\beta = 0^\circ$

Round Windshield

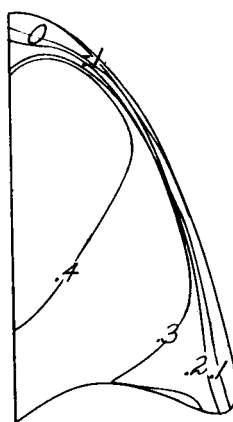


$\alpha = 0.4^\circ$
 $\beta = 0^\circ$

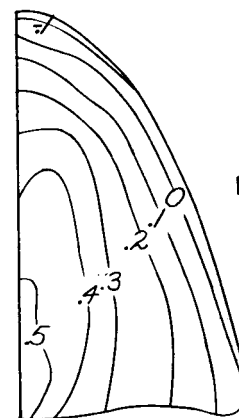
M=1.41



$\alpha = 0.4^\circ$
 $\beta = 0^\circ$



$\alpha = 0.4^\circ$
 $\beta = 0^\circ$



$\alpha = 0^\circ$
 $\beta = 0.3^\circ$

M=2.01

Figure 19.- Pressure coefficient contours on one-half the frontal projections of each of the large rearward-located canopies for $M = 1.41$ and 2.01 .

UNCLASSIFIED

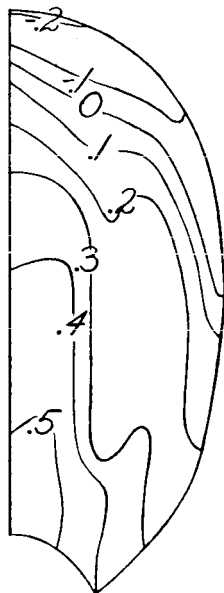
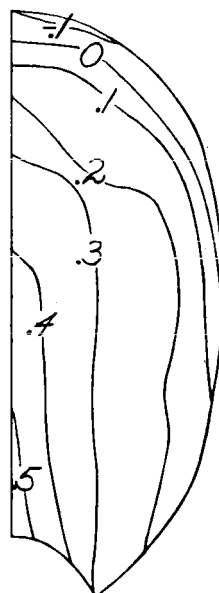
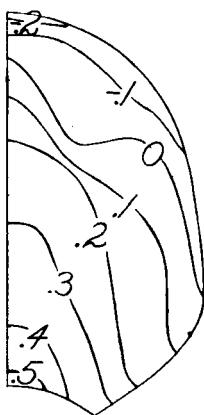
 $M=1.41$  $\alpha = 0.4^\circ$
 $\beta = 0^\circ$ $M=2.01$  $\alpha = 0.4^\circ$
 $\beta = 0^\circ$ Forward-
located $\alpha = 0.4^\circ$
 $\beta = 0^\circ$  $\alpha = 0.4^\circ$
 $\beta = 0^\circ$ Rearward-
located

Figure 20.- Pressure coefficient contours on one-half the frontal projections of each of the small canopy configurations at $M = 1.41$ and 2.01 .

UNCLASSIFIED

CONFIDENTIAL

03171224 J040

CONFIDENTIAL
NO DISSEMINATION
EXCEPT BY AUTHORITY
OF THE DIRECTOR

CONFIDENTIAL